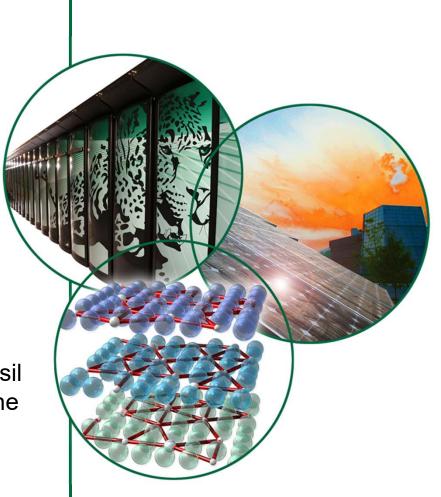
Additive Manufacturing of High Gamma Prime Alloys

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Project Objectives

- Optimize additive manufacturing (AM) fabrication processes for:
- Chromia forming 282: (57Ni-20Cr-10Co-8.5Mo-2.1Ti-1.5Al)
- Alumina forming Nimonic 105 (Ni-20Co-5Mo-15Cr-4.5Al-1Ti)
- Improving understanding of the process-microstructure-property relationships
- Generate data (Tensile, Fatigue, Creep, Oxidation) relevant for FE applications
- Compare two AM techniques, electron beam melting (EBM) and selective laser melting (SLM).
- Effect of annealing/HIP'ing on microstructure and mechanical properties
- Collaboration with other 282 related programs such as A-USC

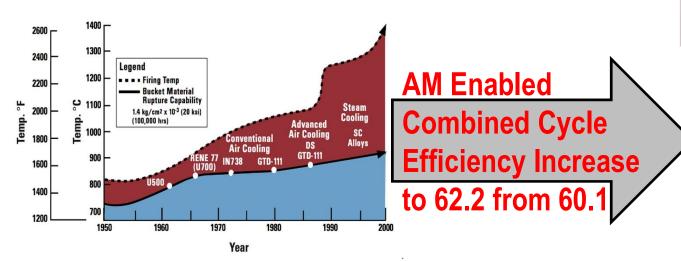


Why Additive of Ni-base Superalloys?

Component cooling designs have existed since the 70s

 Additive manufacturing provides the empowerment for the next evolutionary phase in high temperature component design

Siemens = AM for critical rotating parts



GE gas turbine breaks efficiency record through metal AM optimisation

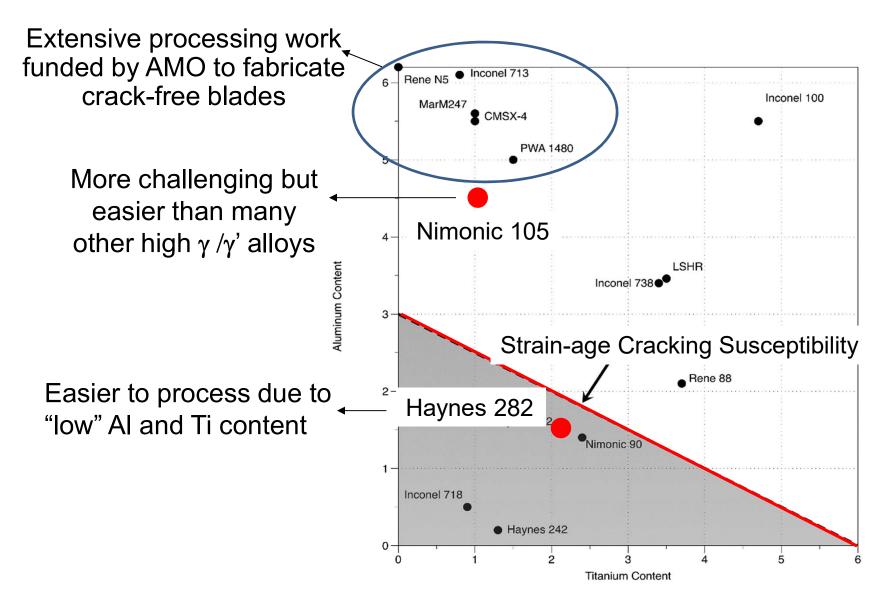
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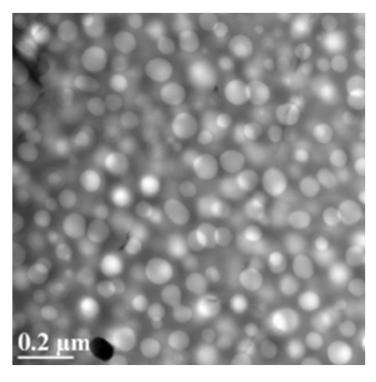
The GE Power-produced 9HA.02 gas turbine is now said to offer 64% efficiency in combinative power plants (Courtesy GE Power)

Alloy 282 Shows Unique Combination of High Strength and Fabricability



Creep Strength Depends on the γ ' Precipitates Stability

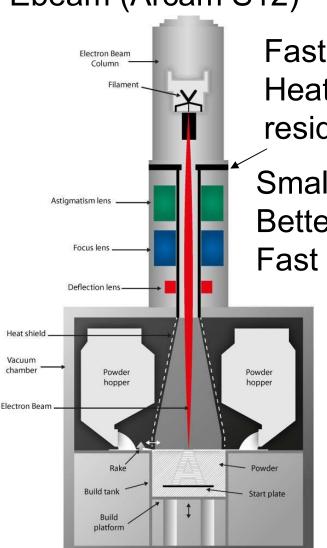
- Optimum γ' precipitate size is ~25nm in wrought 282 after solution annealing at 1121 to 1149°C and two-step aging 2h at 1010°C + 8h at 788°C (Haynes recommendation)
- ~19% γ' vol. fraction after aging
- Recent work has shown that one step 4h 800°C heat treatment led to similar properties (Bruce Pint's talk tomorrow on 282 code case)
- Best heat treatment for AM 282?



2810h, 750°C

Alloy 282 Made by EBM and SLM

Ebeam (Arcam S12)



Faster

Heated bed = lower residual stress

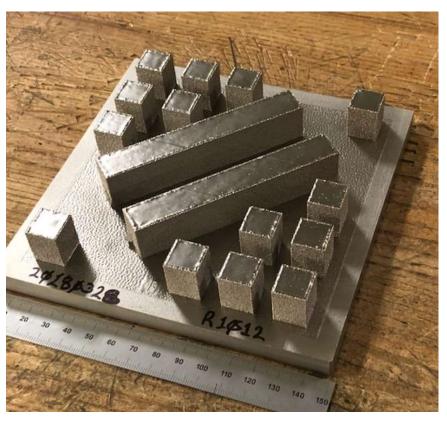
Smaller beam size = Better resolution Fast cooling rate

EOS 250 machine





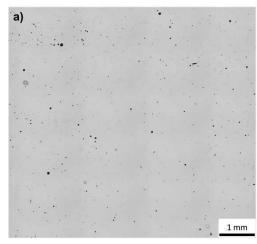
Ongoing Fabrication of Small Cubes and Bars for Microstructure Characterization

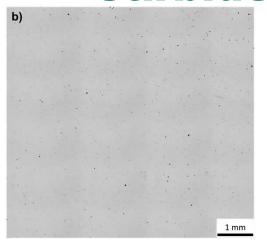


EBM build

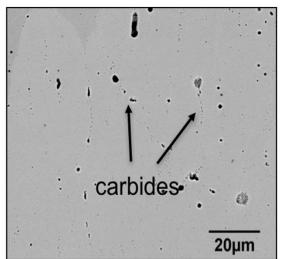
- After parameter optimization based on defect and density, rods and plates will be fabricated for extensive testing
- Testing along and perpendicular the build direction
- SLM rods were provided by Siemens for characterization

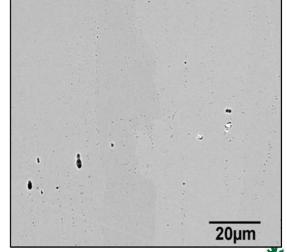
Previous work on As Built EBM 282: Limited Number of Voids + M₂₃C₆ Carbides





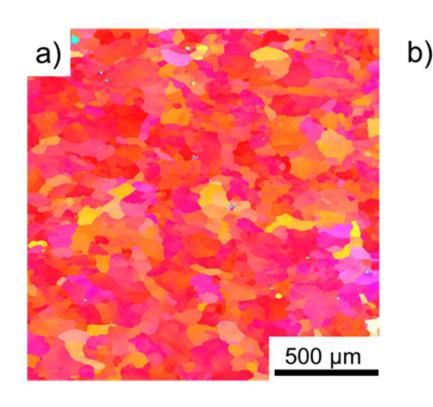
Density from 97.9% to a high of 99.5% depending on build parameters

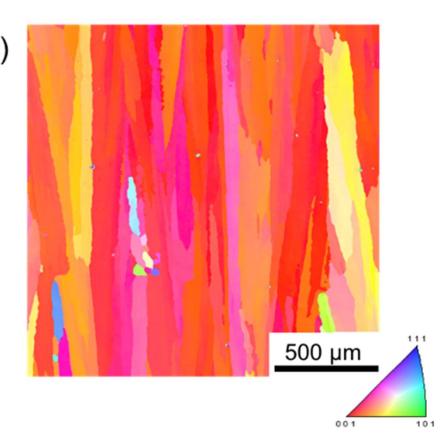




Carbides not expected in wrought 282 before aging treatment

As Built EBM 282: Elongated Grains with [001] Crystallographic Orientation Aligned with the Build Direction

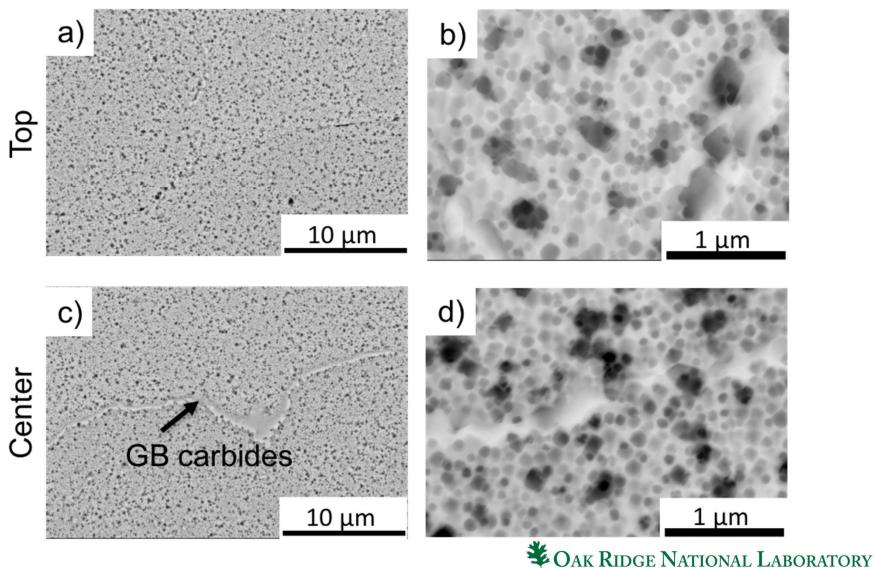




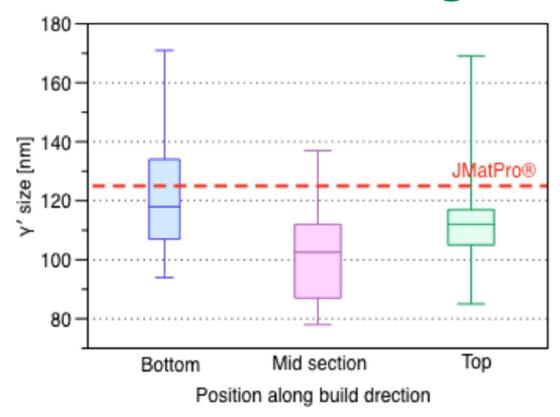


As Built EBM 282: ~110nm γ ' precipitates

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As built EBM 282: γ Larger than optimized γ size after annealing of Wrought 282

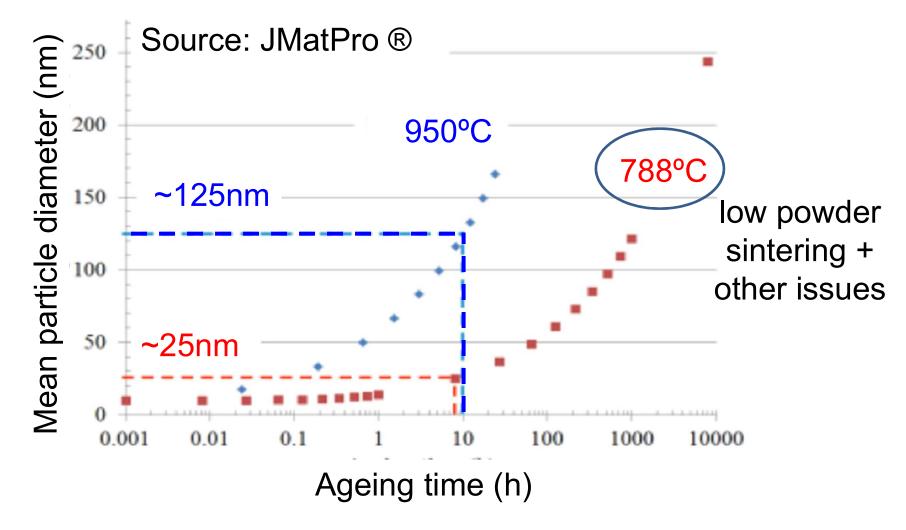


Optimum Gamma prime size is ~25nm in wrought 282 after annealing

Based on image analysis measurements

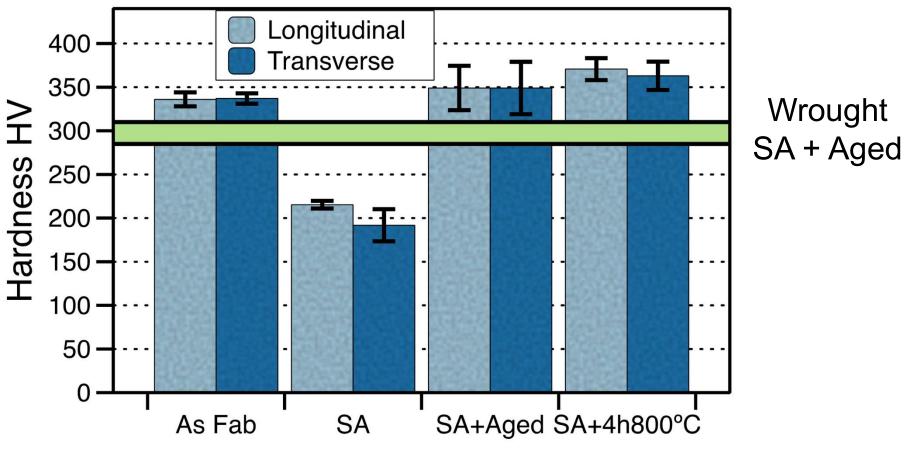


Build Temperature Estimated to be around 950°C Based on JMatPro Model





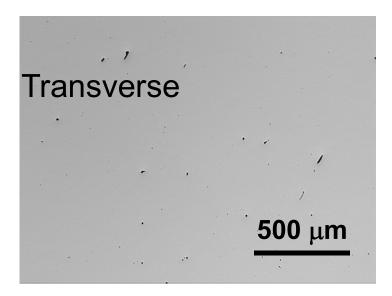
Ongoing Work to Optimize EBM 282 Heat Treatment

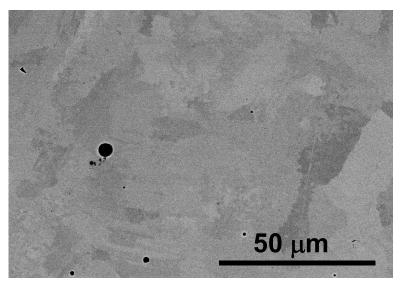


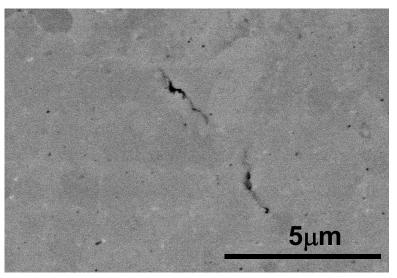
- Solution Anneal leads to γ' dissolution
- Next step γ' characterization



As built SLM 282 : Few Voids + Defects Grain size ~5-50um



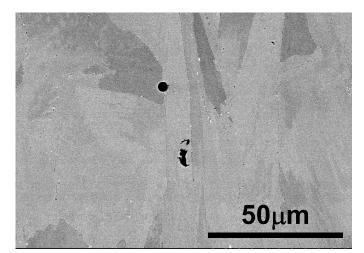




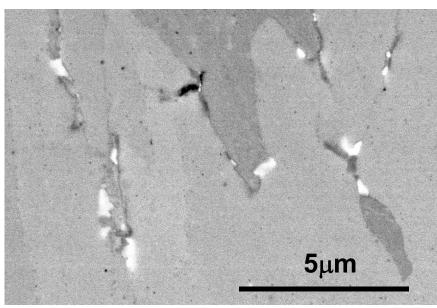
- Few small carbides +Local segregation.
- No precipitates at GB
- Need TEM for gamma prime characterization

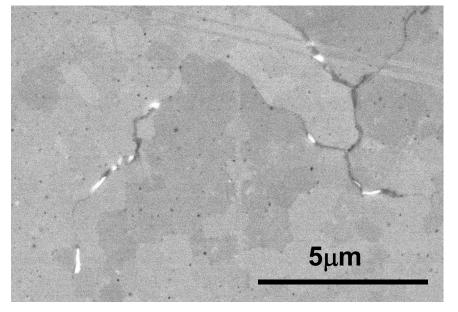


Precipitates formation at GB after 2h at 1010°C + 8h at 788°C



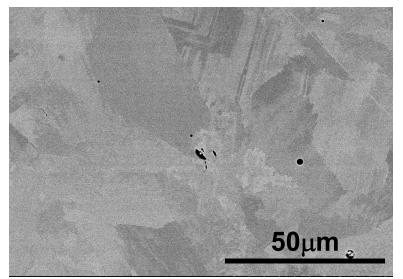
Formation of precipitates at GB Similar defect structure





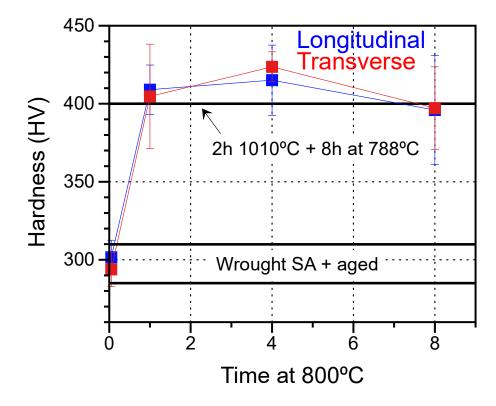


Very High Hardness After Standard Aging or Annealing at 800°C



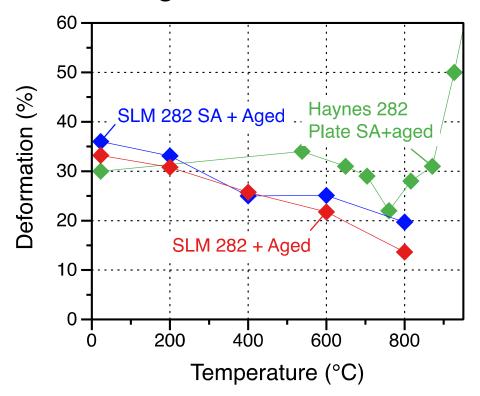
5μm

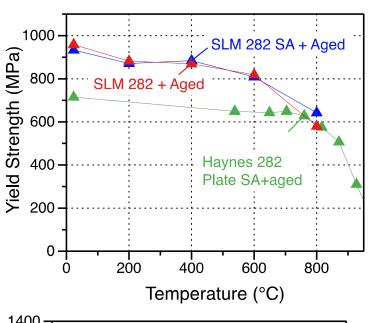
- Similar microstructure and hardness after 1h to 8h at 800°C
- Optimized HT for SLM 282?

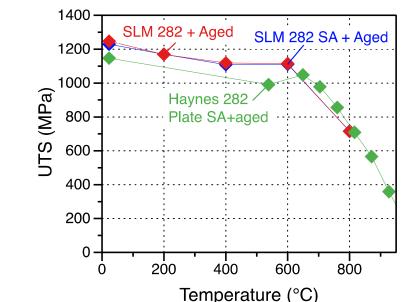


High Strength for SLM 282 after Standard HT

- No effect of solution anneal on tensile properties
- Slightly lower ductility compared to wrought 282?









Conclusion

- Initial microstructure characterization and Hardness measurement for EBM 282 shows high strength is achievable with appropriate heat treatment
- High strength for aged SLM 282. Because of fast cooling, solution anneal might not be necessary
- Next step is to optimize heat treatment (HIP'ing?) and conduct extensive characterization (tensile, fatigue, creep and oxidation)
- Fabrication of actual turbine components
- Working on modeling γ ' evolution during processing
- Aiming at better properties than properties of wrought 282?

