Microstructure and Properties of Nibased Components Fabricated by Additive Manufacturing

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Projects Goals & Objectives

- Optimize additive manufacturing (AM) fabrication processes for Nibased gas turbine components
- Three main AM techniques, electron beam melting (EBeam), laser metal deposition (LMD) and selective laser melting (SLM) will be assessed
- Generate long term data relevant for FE applications for Hastelloy (HX, Ni-22Cr-18Fe-9Mo) alloy
- Fabricate rods of HX by EBM and LMD and assess the effect of Hip'ing and/or annealing on the alloy microstructure *in progress*
- Determine the tensile properties of EBM, SLM and LMD alloys from room to 900°C *in progress*
- Perform or initiate at least three creep tests and three fatigue tests on specimens fabricated by two different AM processes in progress
- Complete comparison of EBM, LMD and SLM AM approaches *in progress Source Complete Comparison of EBM, LMD and SLM AM approaches in*

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Outline

- Fabrication of electron beam melting (EBeam) and selective laser melting (SLM) specimens
- As Fab Ebeam and SLM microstructure and tensile properties (Building direction)
- As Fab Ebeam fatigue and creep properties
- Effect of HIP'ing on microstructure and tensile properties
- Effect of different Ebeam precursor powders on mechanical and oxidation behaviors



Mechanical Properties of HX Made by Ebeam and SLM

Ebeam (Arcam S12)

Electron Beam Column

Astigmatism lens

Focus lens

Deflection lens

Powder

hopper

Rake

Build tank

Build

Heat shield

Vacuum chamber

Electron Beam

Filament

Laser (Renishaw AM250)

Faster

Powder

hopper

Powder

Start plate

Heated bed = lower residual stress

Smaller beam size Better resolution Pulsed laser beam

AM 250 RENISHAW Dak Ridge National Laboratory MANAGED BY UT-BATTELLE FOR THE DEPARTMENT OF ENERGY

ORNL SLM Builds For Microstructure and Tensile Characterization



- Build made with leftover powder from AMO project
- Characterization of the microstructure and tensile properties in 3 directions
- Presentation on vertical specimens
- Study the effect of annealing and Hip'ing



Small SLM HX Cube for Parameter Optimization



- Spot time
- Spacing
- Energy



New SLM Build For Extensive HX Characterization



- 35h, ~2000 layers 65mm tall
- Rectangular blocks to study properties anisotropy
- Thin wall effect



Fabrication of 20-30 Ebeam Specimen For Tensile, Creep and Fatigue Testing



- · 27h, 1240 layers, ~65mm
- Small builds first to optimize parameters based on 718 work
- Study the effect of post annealing and Hip'ing



Very Similar Gas Atomized Powder Composition For Ebeam & SLM

AM	Particle Size				
SLM	size < 44µm				
Ebeam	44µm < size < 125µm				

AM	Ni	Cr	Fe	Мо	Со	Mn	Si	W	С
Ebeam Powder	Bal.	21.76	18.43	8.91	1.51	0.07	0.08	0.6	0.08
Ebeam Alloy	Bal.	21.38	18.55	9.05	1.55	0.01	0.05	0.64	0.078
SLM powder	Bal.	21.47	18.83	8.96	1.51	0.01	0.16	0.63	0.07
Wrought	Bal.	22.06	17.86	9.53	1.8	0.65	0.31	0.6	0.067

Low level of Mn and Si content in comparison with wrought Specification: Mn and Si <1%

Alloy composition consistent with Ebeam powder composition



EBeam Powder Morphology Typical of Gas Atomized Powder



Most powder particles contain large numbers of satellite particles

Irregularly shaped particles

Entrapped gas lead in the powder



Ebeam: Porosity from Powder Porosity + Precipitates along some GB



Transverse direction



Ebeam: Elongated Grain Structure Along The Build Direction



Build direction







Cr or Mo-rich Carbides with Slight Mn and Si enrichment





SLM: Local Delamination/Lack of Melting



SLM: More Complex As Fab Microstructure + Elongated Grains?





SLM HX Exhibits Superior Strength but Lower Ductility



Meet cast HX requirement



Ebeam: Good Low Cycle Fatigue Properties at 800°F/425°C



Ongoing extensive LCF characterization at Siemens



Ebeam: Low Creep Properties at 816°C Better Creep Strength at Lower Temp?





Ebeam: Fully Dense Material after HIP'ing at 1177°C/2h/150MPa



HIP'ed Ebeam: Increasing Number of Small Grains + Precipitates Along Few Grain Boundaries







Haynes data

HX sheet

SLM: Fully Dense Material After HIP'ing at 1177°C/2h/150MPa+Recrystallization



SLM: Increase of Ductility & Decrease of YS after HIP'ing 2h/1177°C/150MPa

Release of Residual Stress + Microstructure evolution





Ebeam Builds Fabricated with High Si (AMO) and Low Si (FE) HX Powders

AM	Ni	Cr	Fe	Мо	Со	Mn	Si	W	С
Powder FE	Bal.	21.76	18.43	8.91	1.51	0.07	0.08	0.6	0.08
Alloy FE	Bal.	21.38	18.55	9.05	1.55	0.01	0.05	0.64	0.078
Powder AMO	Bal.	21.7	18.7	9	1.56	0.93	0.86	0.66	0.06
Alloy AMO	Bal.	21.43	18.87	9	1.56	0.67	0.71	0.65	0.048
Wrought	Bal.	22.06	17.86	9.53	1.8	0.65	0.31	0.6	0.067

HX specification: Mn and Si < 1 wt%

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Significant Impact of Powder Chemical Composition on Microstructure





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High Si HX-AMO Oxidation Coupons 100h Cycle, Air, 950°C



- High spallation rate likely related to High Si grain boundaries
- Less spallation at edges due to different microstructure?

TOYOTO

Limited Spallation for the Low Si HX Coupons at 950°C







No spallation for wrought HX 4x100h at 950°C



HX wrought



Significant Impact on Mass Changes Long Term Effect on Oxidation Behavior?





Conclusion

- Ebeam HX shows good ductility and strength superior to the cast HX requirement
- No effect of HT and Hip'ing on Ebeam HX tensile properties
- SLM HX exhibited good tensile strength and acceptable ductility
- HT or Hip'ing increased the SLM HX ductility but reduced YS
- Good fatigue properties for the Ebeam HX but limited creep strength at 816°C. Larger creep database will be generated + testing of Hip'ed HX alloy
- Significant spallation was observed for the High Si HX alloy Need to control Si content or new scanning strategies to improve oxidation at the surface

