Advanced Multi-Tube Mixer Combustion For 65% Efficiency DE-FE0024006

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Imagination at work



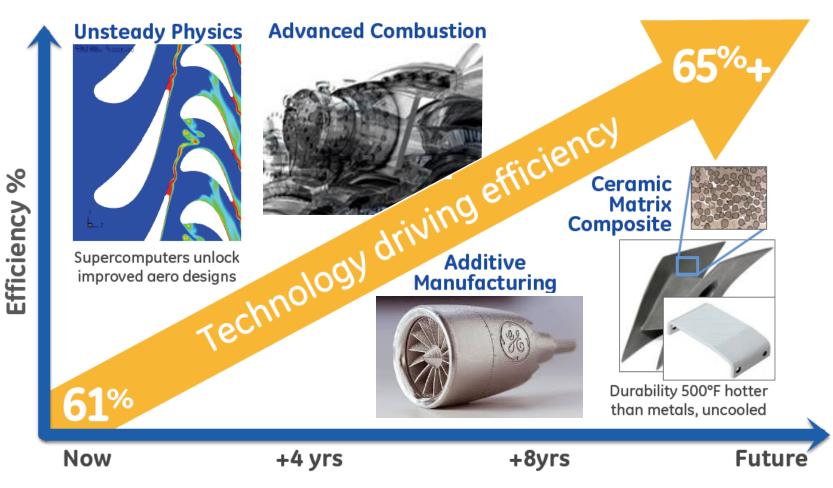
#### 65% GTCC Efficiency

WHY is reaching 65% important?

✓ Less emissions produced
→ Tons per year less CO<sub>2</sub> than 60% GTCC
✓ Lower \$/kW & heat rate

 $\rightarrow$ Faster adoption

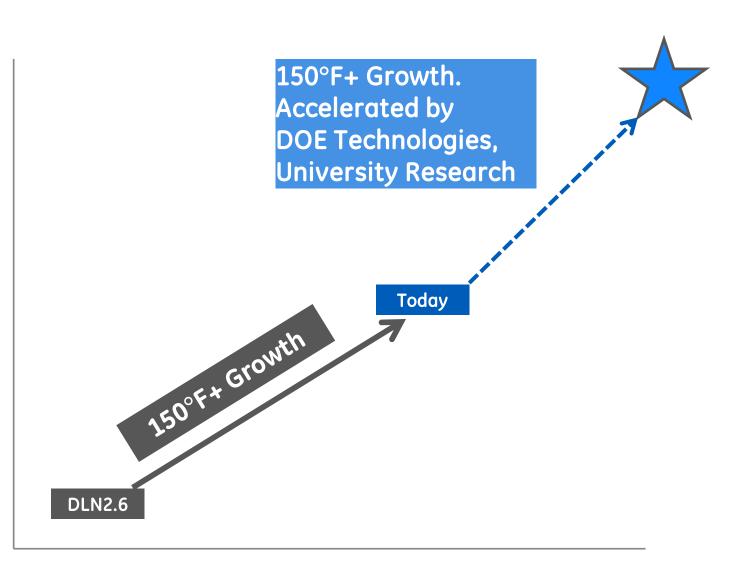
#### Technology Pipeline for 65% Efficiency





### Picking up the Pace

**Combustor Exit Temperature** 





#### **Time in Years**

# Phase 1 Program

#### Advanced Cooling Technologies

- Survey GE Aviation and GE Power for advanced cooling techniques.
- Couple available techniques with recent/future materials and coatings to select a cooling approach that minimizes W<sub>nch</sub> and dP/P.

#### Limited Test Campaign

- Select the system parameters required to reach the target temperature
- Implement the system parameters in GTTL hardware.
- Verify the 'recipe' at full scale conditions.

Configuration Proposal for Phase 2 Development

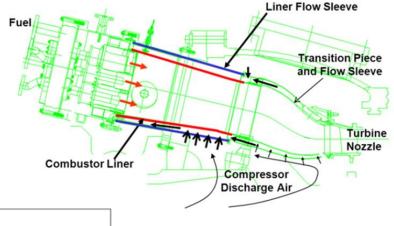
• Understand key trades and select key parameters.



# Materials and Cooling Technologies

Challenge: Increase heat load while holding durability and dP/P

- 1. Research cooling technologies best suited to handle higher heat loads. Review both internal and external material
- 2. Pair with materials and coatings technology and combustor configuration to optimize cooling dP required for 65%



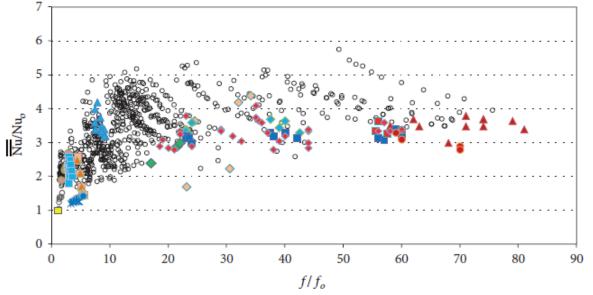


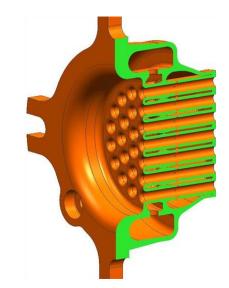
Figure: Ligrani, Phil. 2013. "Heat transfer augmentation technologies for internal cooling of turbine components of gas turbine engines." International Journal of Rotating Machinery 2013.



#### Multi-Tube Mixer

The Multi-Tube Mixer is a distributed, premixed approach to combustion of high-hydrogen fuels.

- Utilizes jet-in-cross flow mixing of fuel and air
- Design air velocity above flame speed with reasonable pressure drop.
- Mixing length ( $L_{mix}/D$ ) may vary based on fuel and conditions.



Multi-Tube mixer is easily scalable without changing fundamental geometry and adaptable to range of fuels.

#### GT2012-69913

DEVELOPMENT AND TESTING OF A LOW NO<sub>X</sub> HYDROGEN COMBUSTION SYSTEM FOR HEAVY DUTY GAS TURBINES



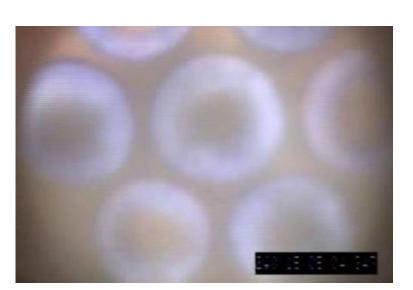
Larger scale MT Mixer Nozzle

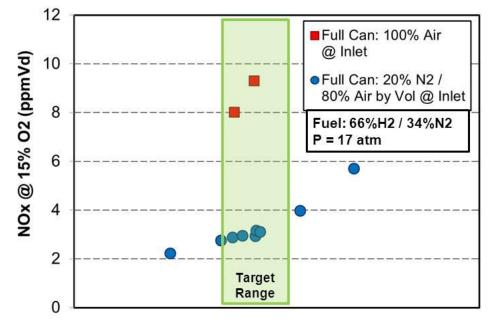


Small prototype MT Mixer



#### Multi-Tube Mixer Full Can Rig Results





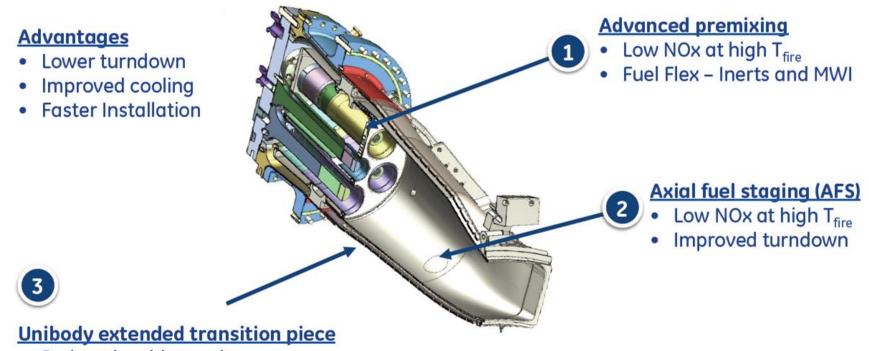
**Combustor Exit Temperature (K)** 

Fuel: 63/4/33 % H2/CH4/N2

- Single digit NOx (corrected to 15% O2) over target temperature range with hydrogen-nitrogen fuel
- Below 3 ppm NOx (corrected) with 20% of inlet air replaced with pure nitrogen.



## DLN2.6+ with Axial Fuel Staging



- Reduced residence time
- Tailored TP cooling

#### Going to hotter temperatures with constant emissions

Going to hotter temperatures with constant emissions

# Final Summary

- 65% is an aggressive but necessary goal.
- Will require advancements in every aspect of combustion design.
- Multi-Tube Mixer and Axial Fuel Staging are excellent building blocks.
- DOE and University projects have already accelerated the pace of technology development, and will continue to be critical to the success of the project.

# Questions?

Imagination at work