

# New Capabilities for MHD Power Generation Enabled by Nanosecond High-Voltage Pulses and Electron Beam Methods

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## Past Experience in Aerospace Applications of MHD

- Electric power generation using MHD
- Use of MHD power to control aerodynamics and propulsion:

- Surface plasmas for control of the boundary layer and laminar-turbulent transition.
- Plasma Virtual shapes created by off-body energy addition for drag reduction, steering, shock control, flow turning
- Plasma-assisted combustion

- Power extraction from one region and its use in another region (MHD bypass)
- Dual-use MHD devices (both power generation and flow control)

## Methods of Ionization of Cold Hypersonic Air Flows

Parameter range:  
T=200-1000 K, P=0.01-1 atm, U=1000-4500 m/s  
Required electron density:  
 $10^{12} - 10^{14} \text{ cm}^{-3}$

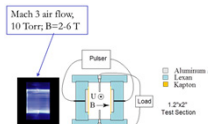
### Thermal ionization:

- Works well only at T>2000 K and with alkali seed
- Problems with mixing the seed into the supersonic flow
- Natural ionization for reentry



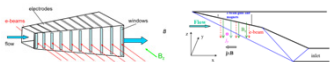
### Nonequilibrium field-induced ionization:

- High-voltage ns pulses, recently ps pulses
- Issues with arcing instability (caused by coupling between T and ionization through E/N)

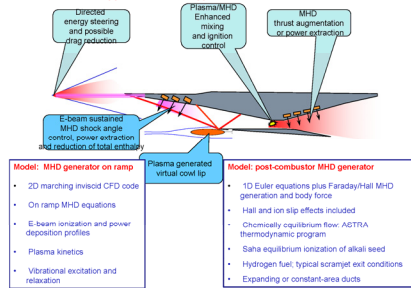


### Ionization by electron beams

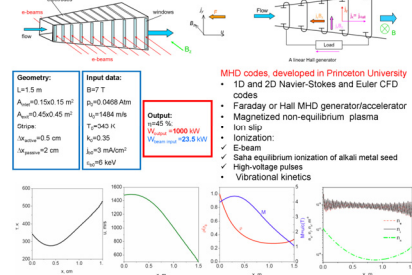
- The most efficient (40-50%) ionization source
- Straightforward and flexible control
- Stability: short-circuiting through boundary layers and arcing instability can be avoided
- Principal issue: transmission windows (mechanical and thermal strength)



### Combined Approach to MHD Bypass and Plasma Flow/Inlet Control



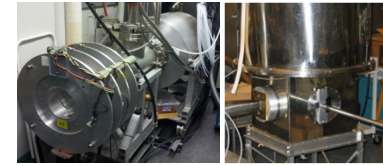
### An example of Hall type non-equilibrium air MHD generator with e-beam ionization



## Unique Experimental Facilities

- 60 kV high aperture E-beam gun
- Superconducting magnet B=6 T
- High-voltage picosecond pulse generator

Output voltage – 200 kV  
Load – 50 Ohm  
Peak current – 4 kA  
Rise time – 100 ps  
Pulse width – 250 ps  
Maximum PRF – 1 kHz



E-beam gun

Magnet

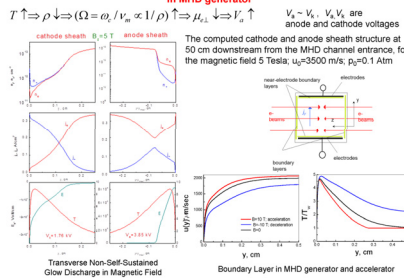


HV pulse generator

## New Opportunities for Enhanced Power Generation

- Extension of MHD "topping cycle" power extraction to higher efficiency through higher velocity expansion (to Mach 3 or greater)
  - Low static temperature associated with high Mach number normally reduces the conductivity below operable limits.
  - We propose overcoming the loss of conductivity with high repetition rate, high voltage nanosecond pulses or electron beams that repeatedly ionize the seed material.
  - The low density extends the recombination time, reducing the enthalpy/entropy penalty
  - The high Hall parameter at low density allows for "flying electrodes".
- For solving these problems it is necessary to carry out experiments and theoretical modeling
  - Princeton University and PPPL have experimental facilities available, modeling capabilities, and experience with supersonic MHD power extraction that permit us to address this concept.

### Strong coupling between boundary layers and near electrode sheaths in MHD generator

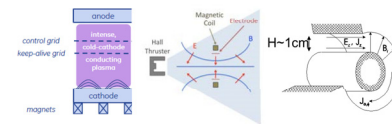


### PPPL Particle-in-cell codes used as a tool of study:

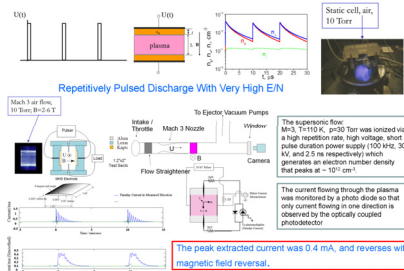
- An electrostatic parallel, implicit, 1D PIC code EDIPIC. Scattering, ionization, and excitation was added into the code, as well as electron-ion and electron-electron collisions.
- 3D LSP and VARP codes include electromagnetic and electrostatic modules, collision modules were implemented into LSP and validated.

### Numerical PIC models were applied for:

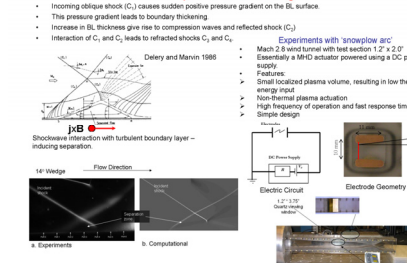
- Controlling plasma properties with applied magnetic field
- Electric Propulsion, Plasma Sources for Ion Drives, Magnetron Discharges, High Power Plasma Switch
- Controlling plasma properties in low-collisional plasmas with active boundaries
- Electric Propulsion, Accelerators, Divertors for Magnetic Fusion Devices



### First Experimental Demonstration of MHD Effect in Cold Supersonic Air Flow With External Ionization



### MHD Shock-Wave Boundary Layer Control, "Snowplow Arc": Experiment and Modeling



## Relevant Publications

- M.N.Shneider, S.O.Macheret, R.B.Miles, AIAA paper 99-3532.
- S.O. Macheret, M.N. Shneider, R.B. Miles, R.J. Lipinski, AIAA J. **39**, 1127 (2001).
- M.N. Shneider, S.O. Macheret, R.B. Miles, AIAA J. **42**, 2303 (2004).
- R.C. Murray, S.H. Zaidi, N.R. Carraro, L.M. Vasilyak, S.O. Macheret, M.N. Shneider, R.B. Miles, AIAA J **44**, 119 (2006).
- S.O. Macheret, M.N. Shneider, R.B. Miles, AIAA J. **45**, 2157 (2007).
- T. Wan, G.V. Candler, S.O. Macheret, M.N. Shneider, AIAA J. **47**, 1327 (2009).
- C.S. Kalra, M.N. Shneider, R.B. Miles, Phys. of Fluids **21**, 106101 (2009).
- M. Campanell, A.V. Khrabrov, I.D. Kaganovich, Phys Rev Lett. **108**, 255001 (2012).
- A. Starikovskiy, N. Aleksandrov. Aeronautics and Astronautics, ed by: Max Mulder. ISBN 978-953-307-473-3. 2011.
- A. Starikovskiy, N. Aleksandrov. Progress in Energy and Combustion Science **39**, 61 (2013).
- A. Starikovskiy, N. Aleksandrov, A. Rakitin. Phil. Trans. R. Soc. A **370**, 740 (2012).