

# Development and Evaluation of a Novel Fuel Injector Design Method using Hybrid-Additive Manufacturing

Project DE-FE12806463, Oct. 2019 – Sept. 2022

Program Monitor: Mark Freeman

Project Review – November 7, 2019

**PI:** Jacqueline O'Connor

**Co-PIs:** Guha Manogharan, Yuan Xuan

Mechanical Engineering

Pennsylvania State University

**Industry Partner:** Solar Turbines Incorporated

Hanjie Lee, Dave Voss



# Overview of presentation

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- Background
- Technical approach
- Project objectives
- Project structure
- Next steps

# Overview of presentation

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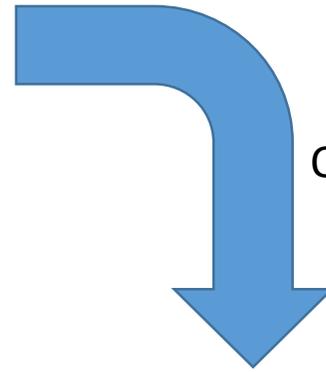
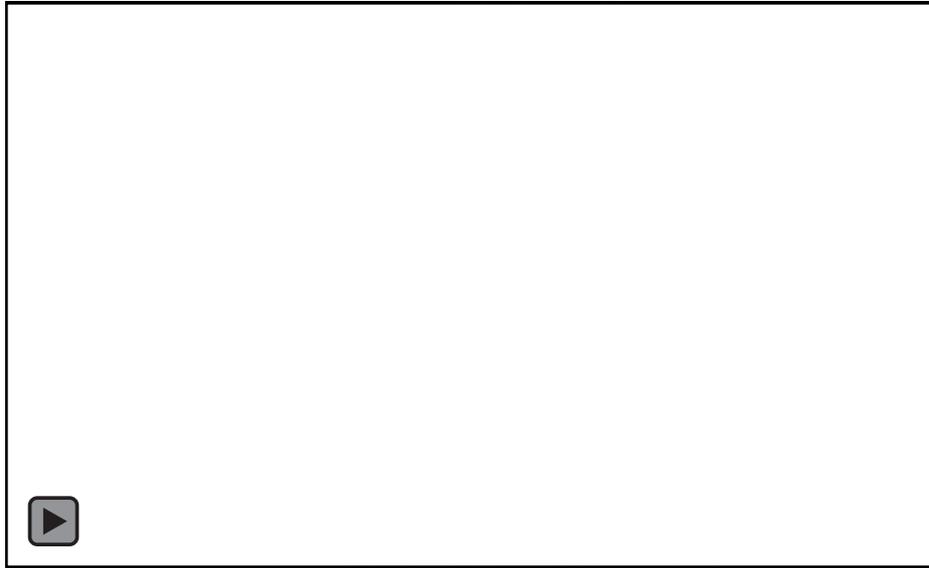
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Project origin: Discussions with industry about issues related to combustion operability and fuel injector manufacturing

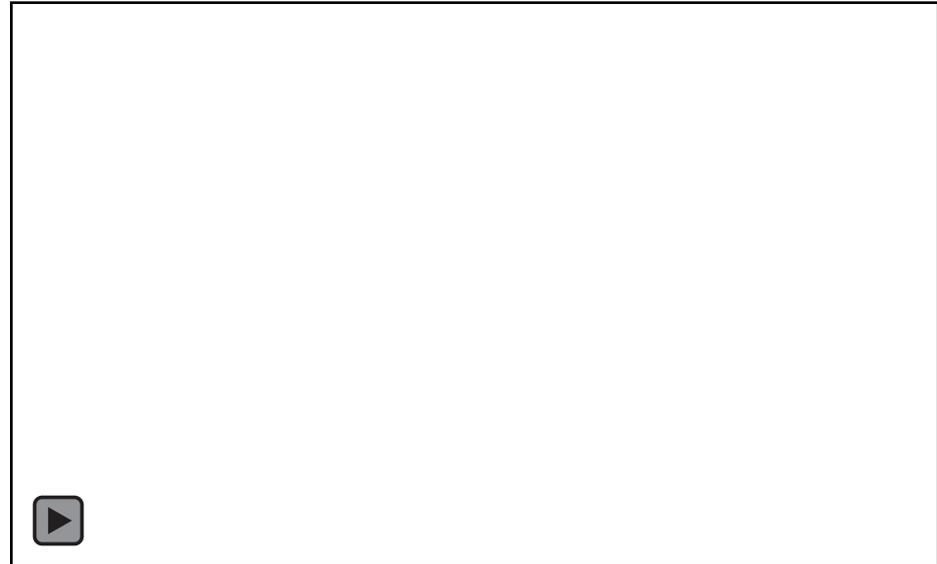
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“Why do fuel injectors have to look like fuel injectors?”

Current fuel injector designs do well at flame stabilization for a moderate range of fuel compositions, operating conditions

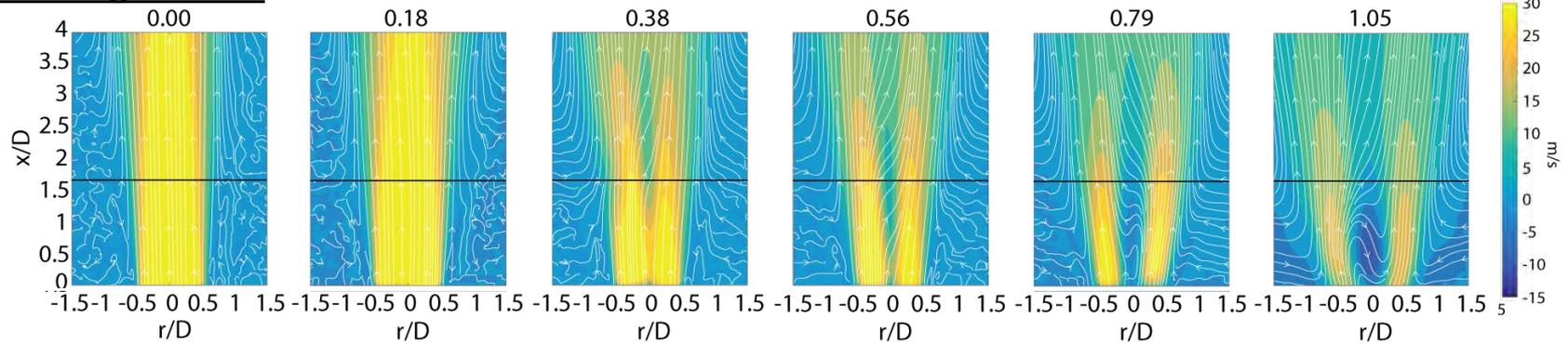


Off-design operation

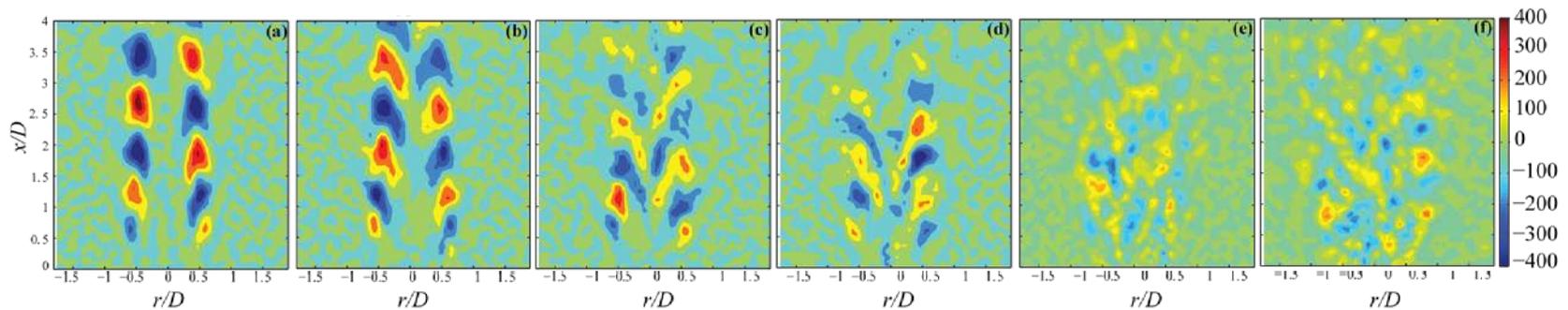


Recent work by PI and collaborators has showed that a stable flow can be “designed” using hydrodynamic stability analysis

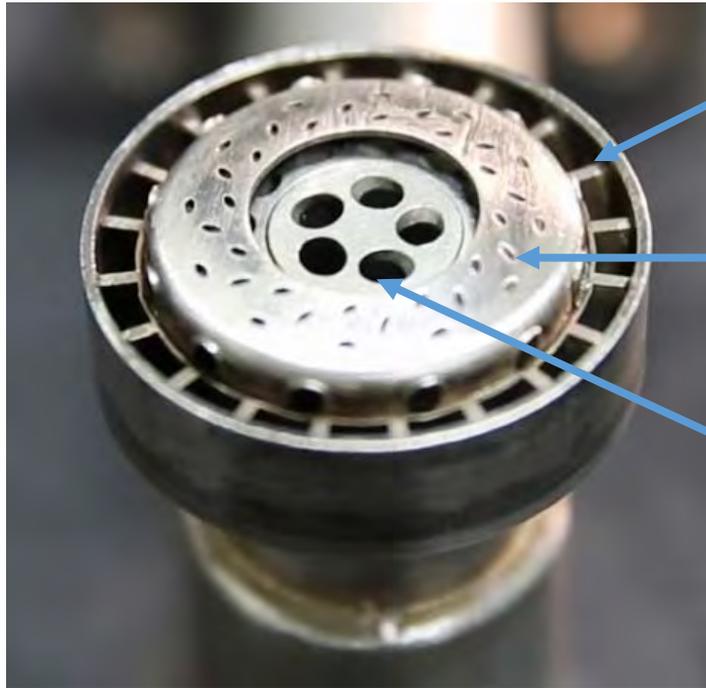
### Time-averaged flow



### Coherent response



Fuel injectors are notoriously difficult to manufacture and can be comprised of dozens of components, assembled by hand

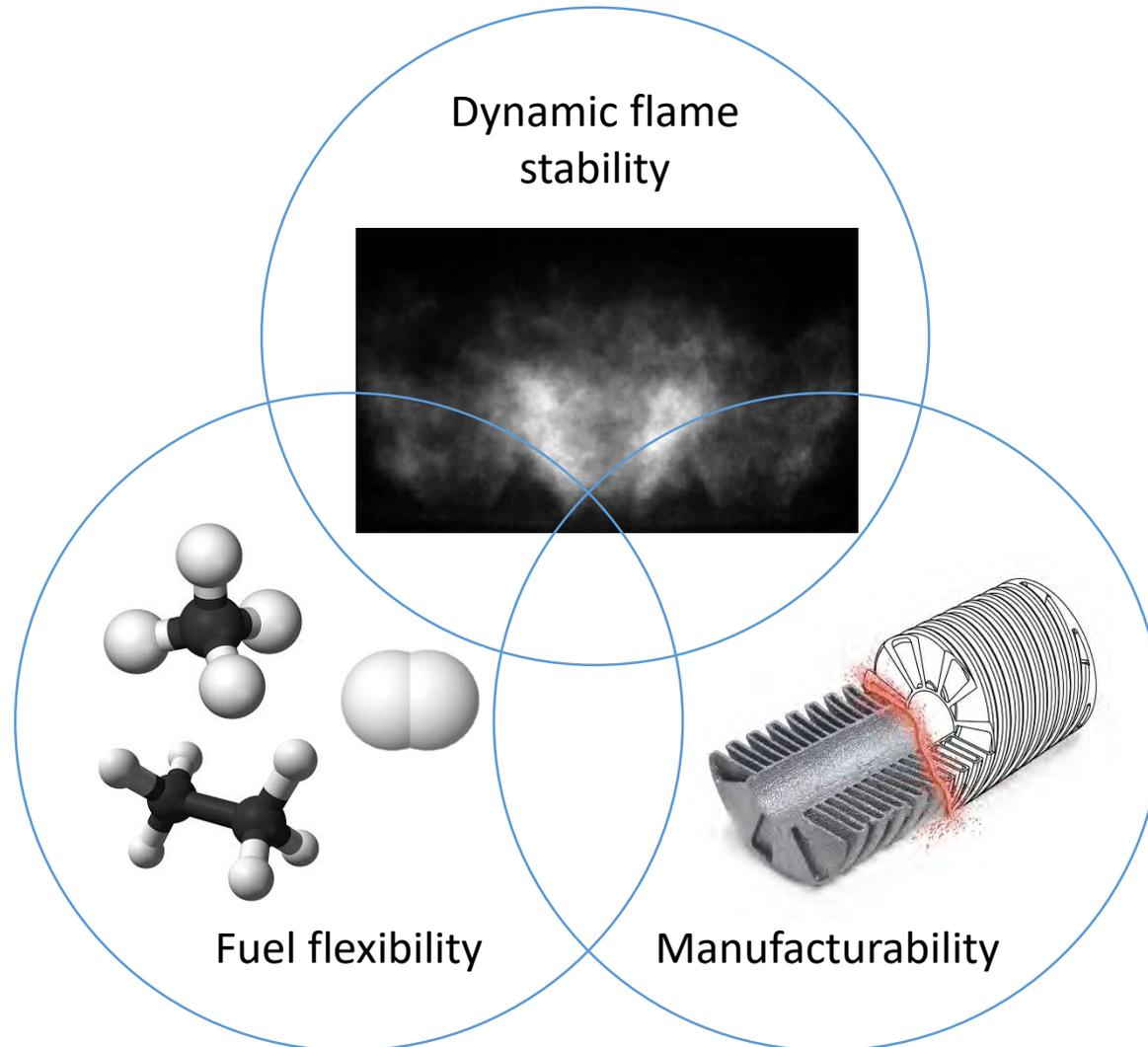


Complex aerodynamic surfaces

Small orifices with specified surface finish

Internal flow passages

# Goal of this project is to create a design optimization paradigm that marries combustion physics and manufacturing



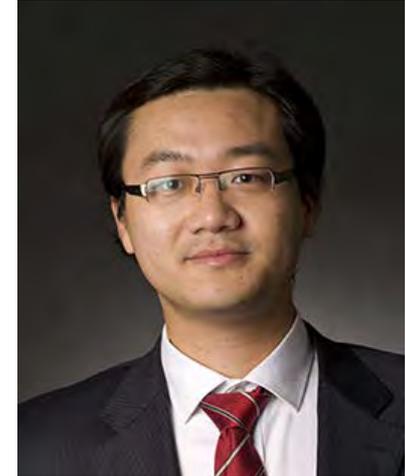
The team is comprised of three PIs and two grad students from Penn State and industrial partners Solar Turbines



PI: Jacqueline O'Connor  
Associate Professor of ME  
Combustion/Gas Turbines



Co-PI: Guha Manogharan  
Assistant Professor of ME  
Design +Hybrid-Additive Manufacturing



Co-PI: Yuan Xuan  
Assistant Professor of ME  
Combustion simulation

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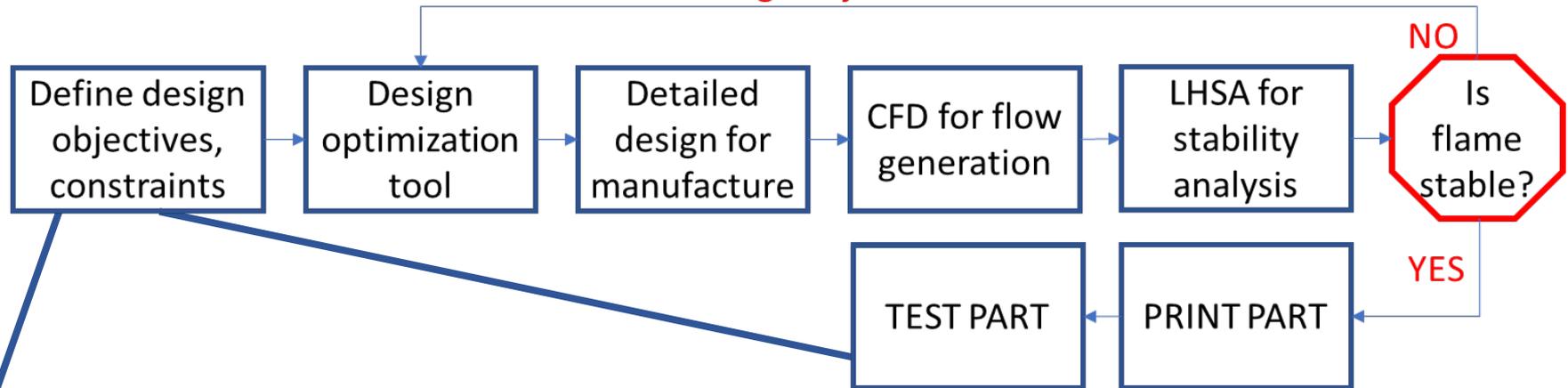
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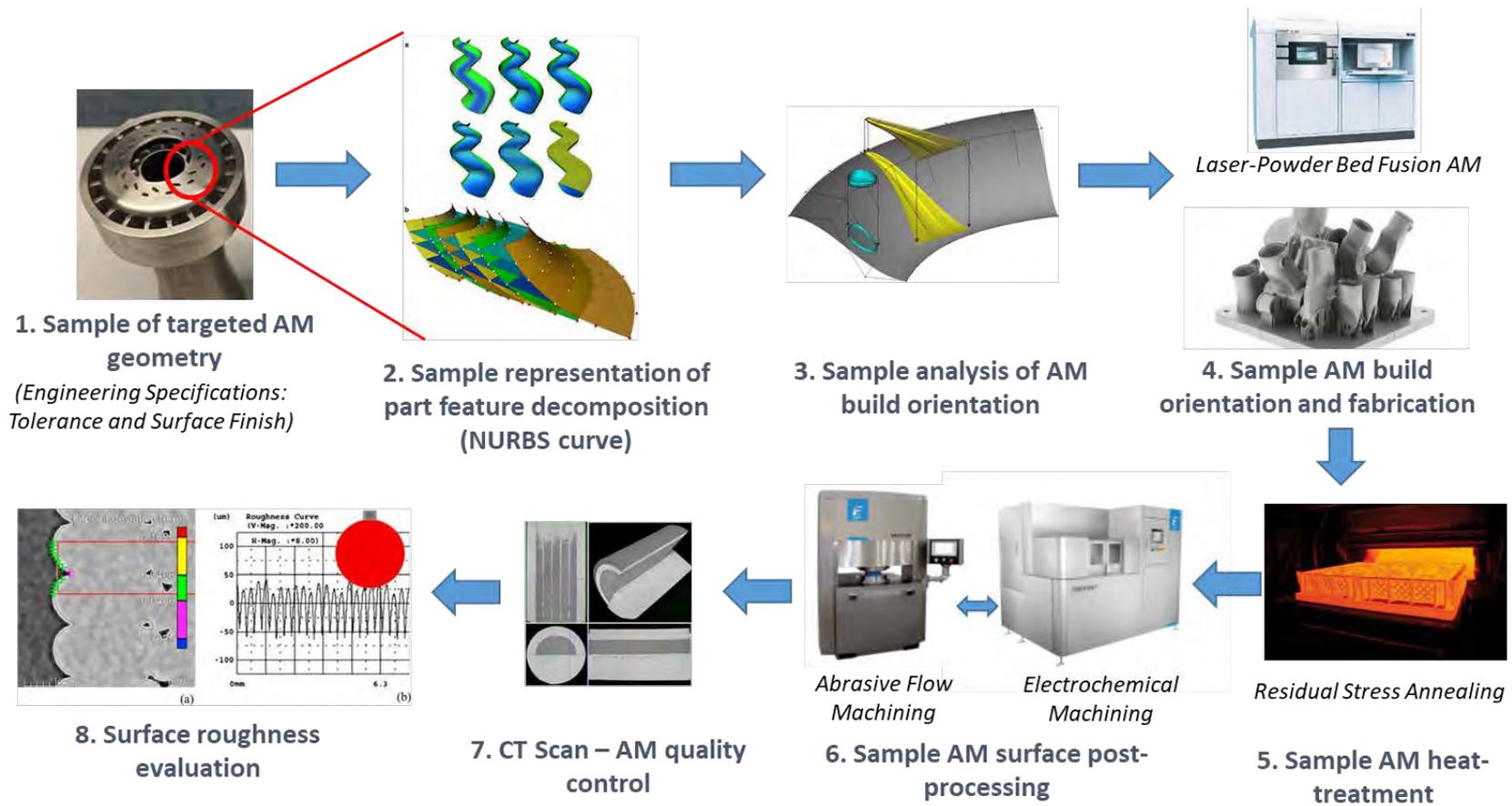
# Technical approach uses an optimization framework for incorporating combustion and manufacturing constraints

Re-design objectives and constraints



1. Flame dynamic stability through flow response modeling
2. Flame static stability through computational fluid dynamics
3. Fuel flexibility through computational fluid dynamics
4. Additive manufacturing considerations for laser powder bed fusion
5. Surface finishing considerations for abrasive flow machining

# Parametric design-process planning advanced manufacturing approach is proposed for fuel-injector applications



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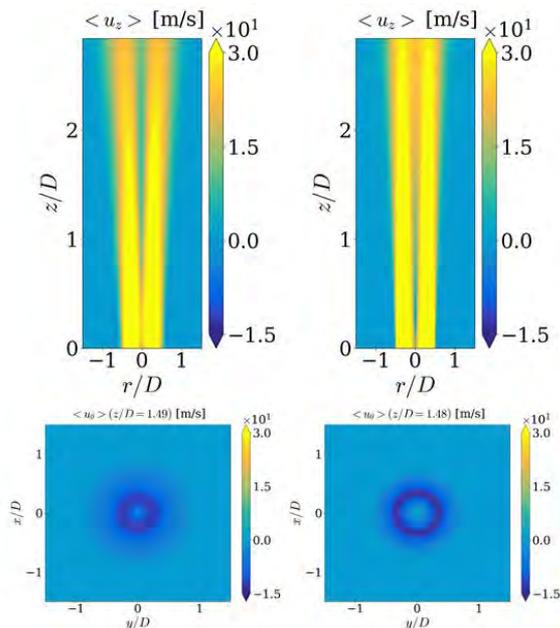
# Project objectives center around four gaps in the fuel injector design process to help industry

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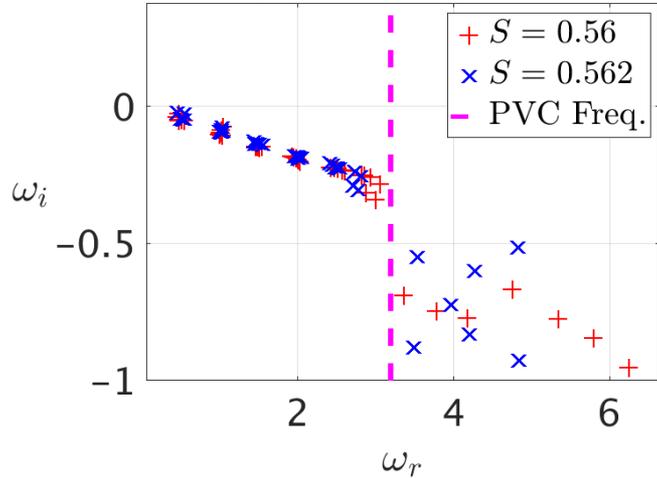
- Integrate issues related to flame static and dynamic stability more seamlessly into the design process
- Incorporate the use of hydrodynamic stability analysis for prediction of dynamic stability issues for efficient computational prediction
- Incorporate high-fidelity, multi-physics modeling into optimization processes
- Link post-processing steps of the AM component into the design optimization process

# An initial proof of concept exercise showed that hydrodynamic stability + CFD could be used for optimization

Step 1: CFD captures experimental trends



Step 2: LHSA results from CFD and experiment match



Step 3: Optimization of flow stability ( $\omega_i$ ) on parameter S

$$L(S, \lambda) = \omega_i(S) - \lambda(\omega_i(S) - \epsilon)$$

$$\frac{\partial L}{\partial S} = 0 \rightarrow \frac{\partial \omega_i}{\partial S} (1 - \lambda) = 0 \rightarrow \lambda = 1$$

$$\lambda \omega_i(S) = \epsilon$$

Result: Optimizer stepped towards stability condition

Iteration	S	$\omega_r$	$\omega_i$
1	0.56	3.40	-0.68
2	0.57	3.38	-0.69
3	0.6	3.03	-0.28
4	0.65	3.03	-0.27

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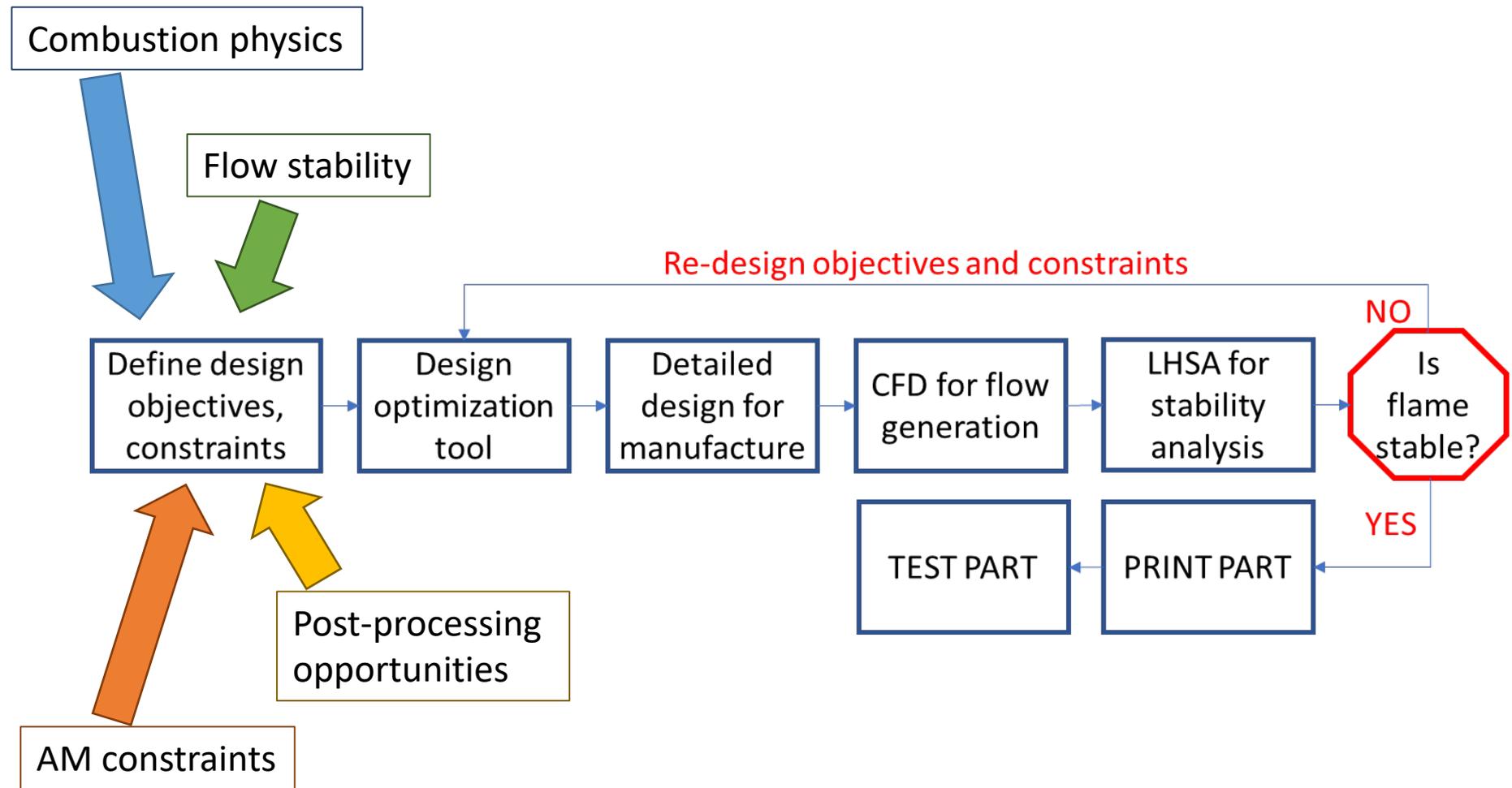
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# Project objectives center around four gaps in the fuel injector design process to help industry

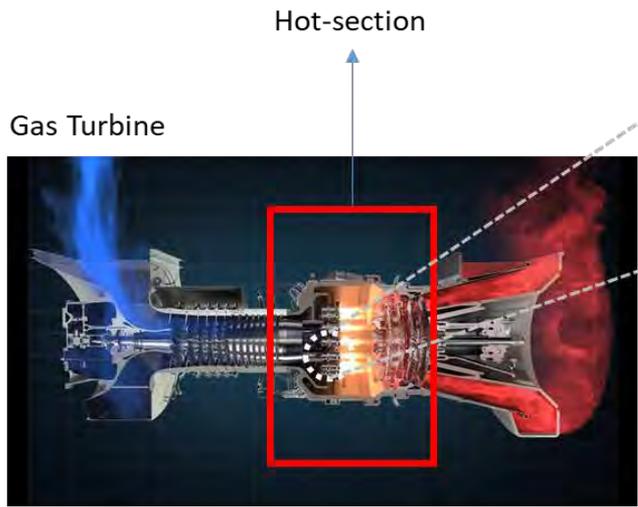
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- Task 1: Project management and planning
- Task 2: Establish baseline
- Task 3: Develop design optimization tool
- Task 4: Implement optimization process on baseline configuration
- Task 5: Design process improvement
- Task 6: Integration of improved design process
- Task 7: Final process testing and technology transfer

# Novel contribution: Design optimization tool that will incorporate physics-based constraints



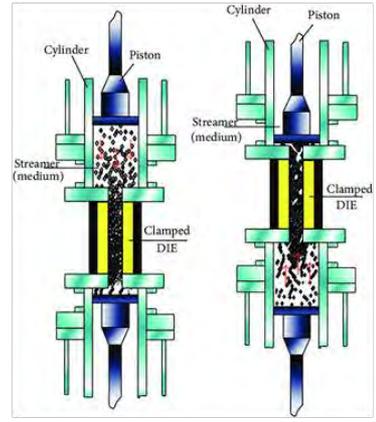
# Novel contribution: Integrated Design-AM-Post Processing Framework



(Source: www.solarturbines.com)



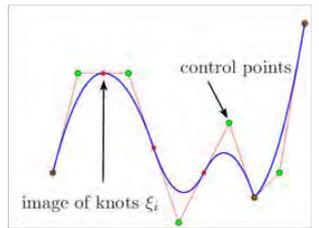
(Source: Solar Turbines (2011))



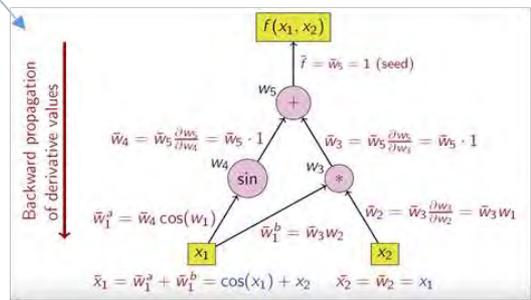
(Source: Li Junye, et. al. (2015))



NURBS curves

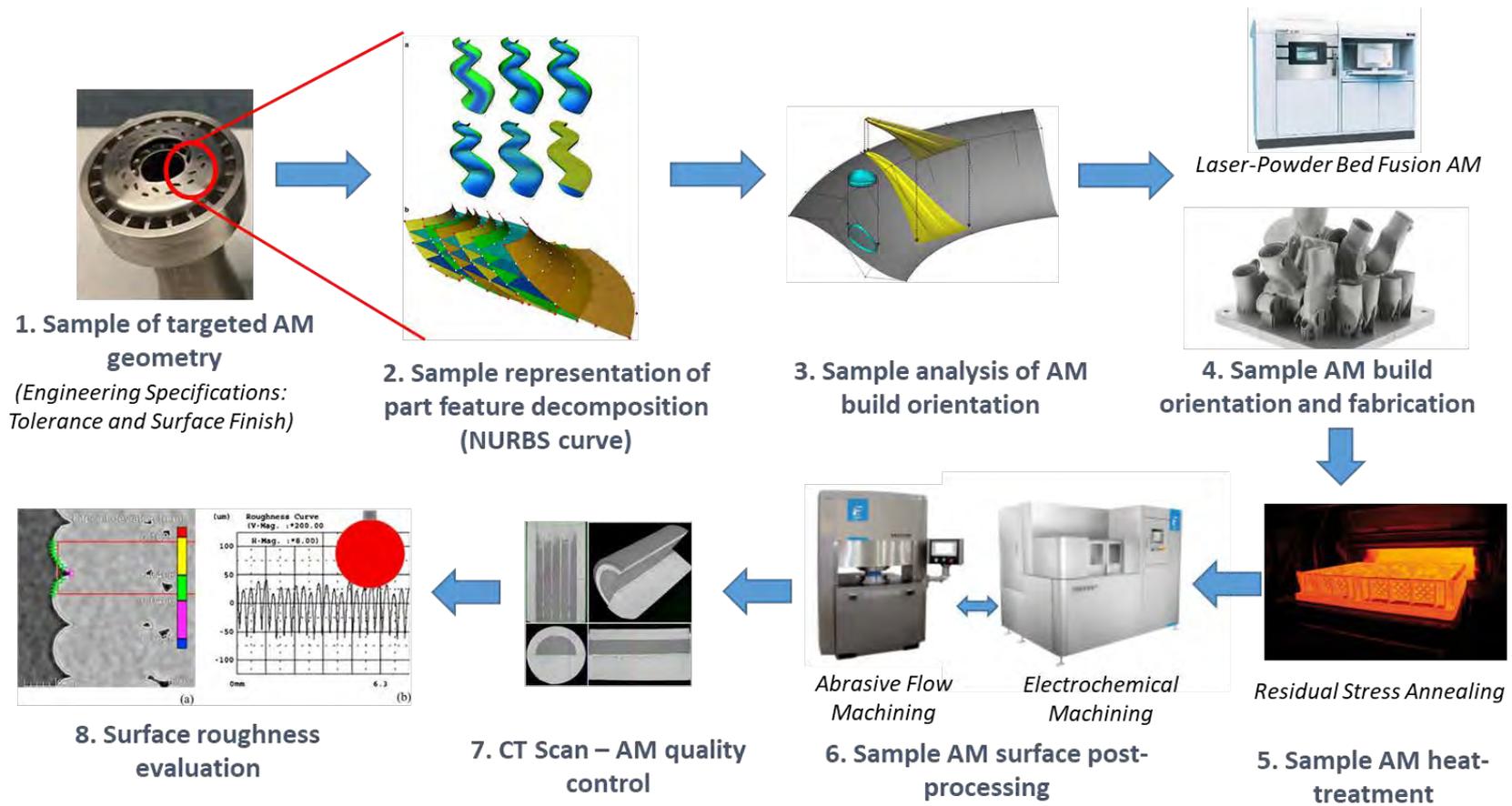


(Source: Vin Phu Nyugen, et. al. (2013))



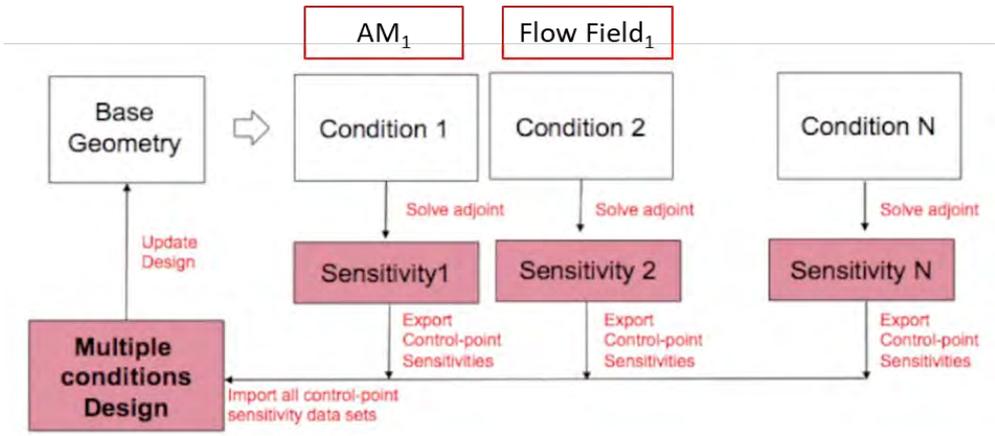
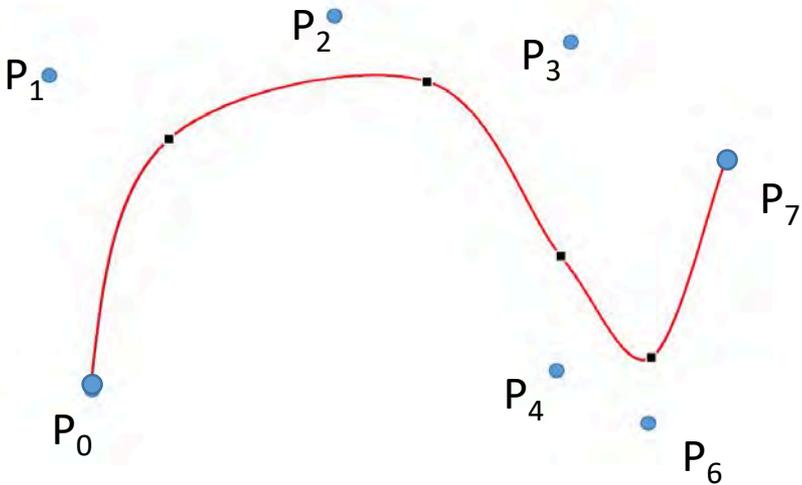
Adjoint based design

# Parametric design-process planning advanced manufacturing approach is proposed for fuel-injector applications



# Novel contribution: NURBS based design approach

- Non-Uniform Rational Basis Splines
- Super set of all curves
- Advantages
  - Can approximate most curves and surfaces mathematically
  - Flexibility
  - Computational efficiency
  - Local propagation (segments)
  - Maintains continuity requirements (slope and curvature)



Project impact stems from multi-disciplinary team and technology transfer to industry partner for future designs



- Gas turbine operability
- Combustion simulation
- Hybrid manufacturing
- Material post-processing
- Design optimization

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- Combustor design
- Manufacturing at scale
- Process integration



Questions?

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# Backup slides

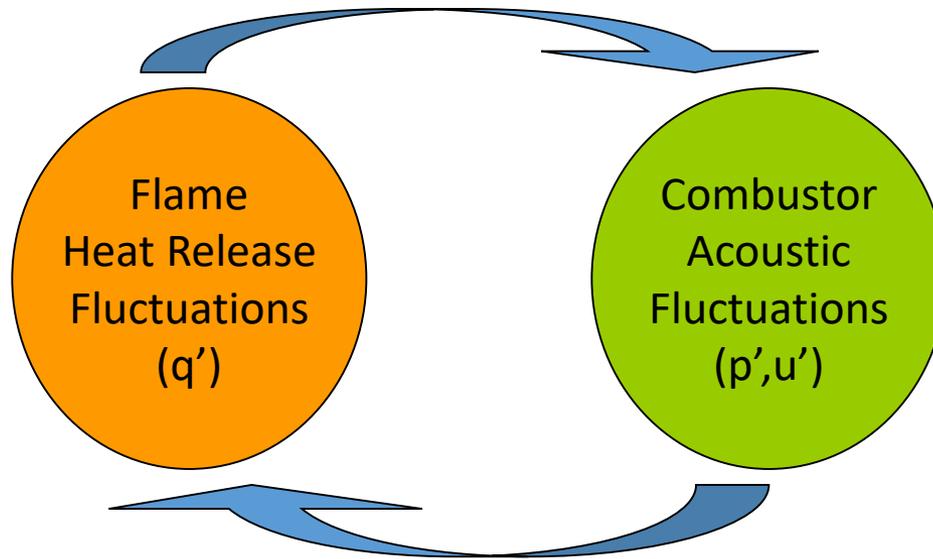
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# Combustion dynamics is one of the most expensive problems in the gas turbine industry

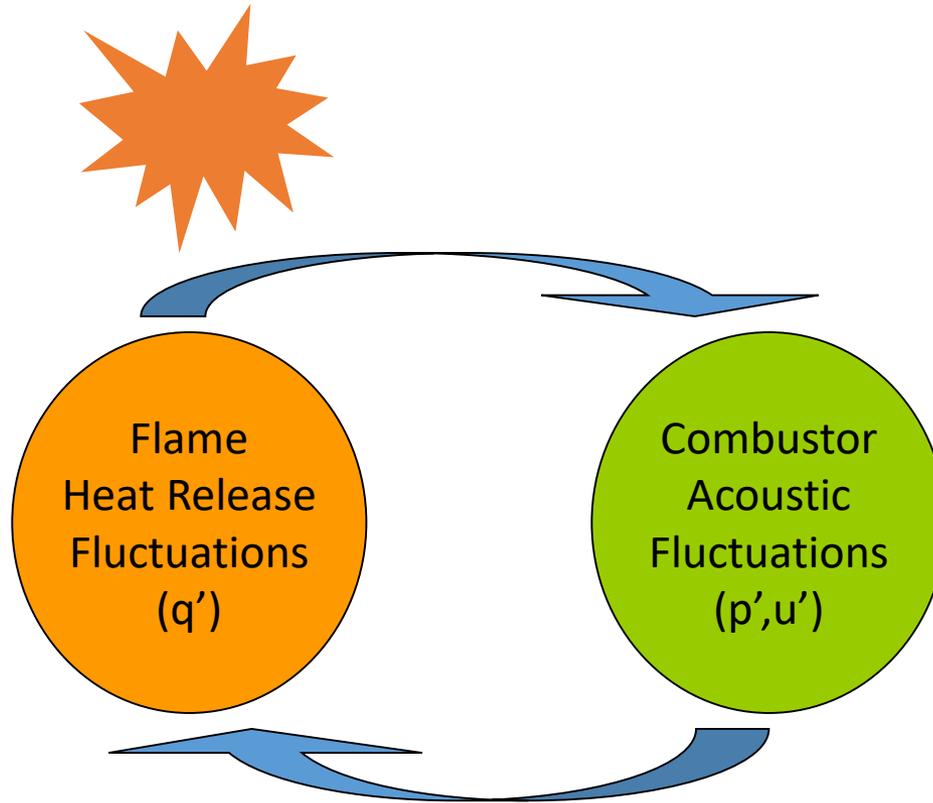
- Damages expensive hardware
- Reduces operability
- Increases emissions ( $\text{NO}_x$ , CO)



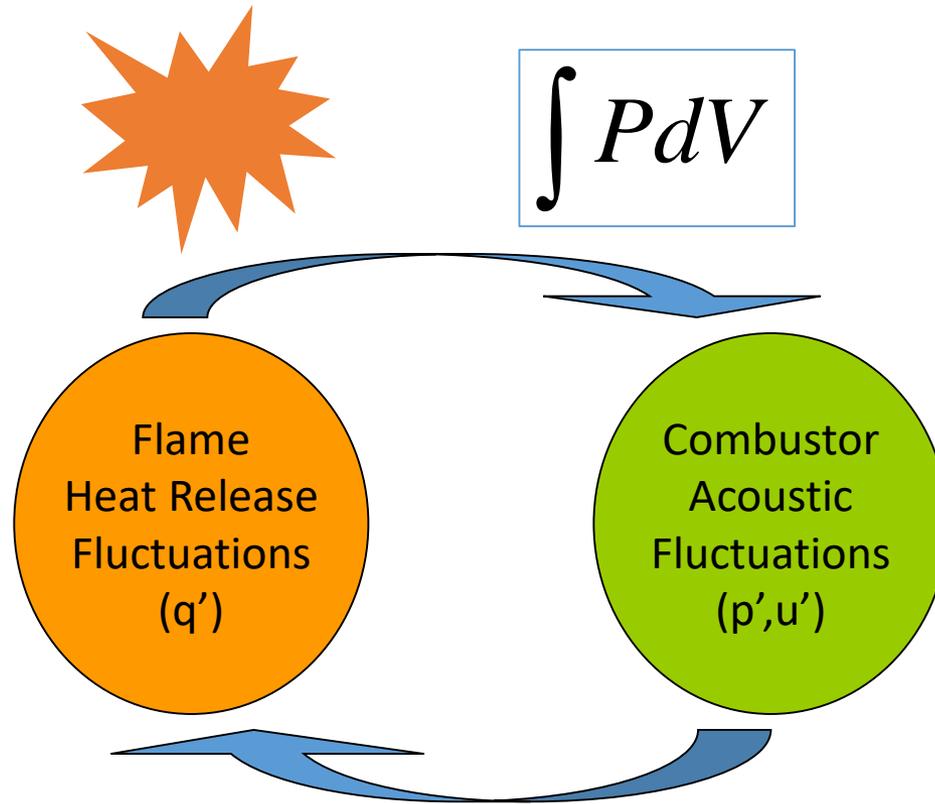
# Combustion dynamics is a coupling between combustor acoustics and flame heat release rate fluctuations



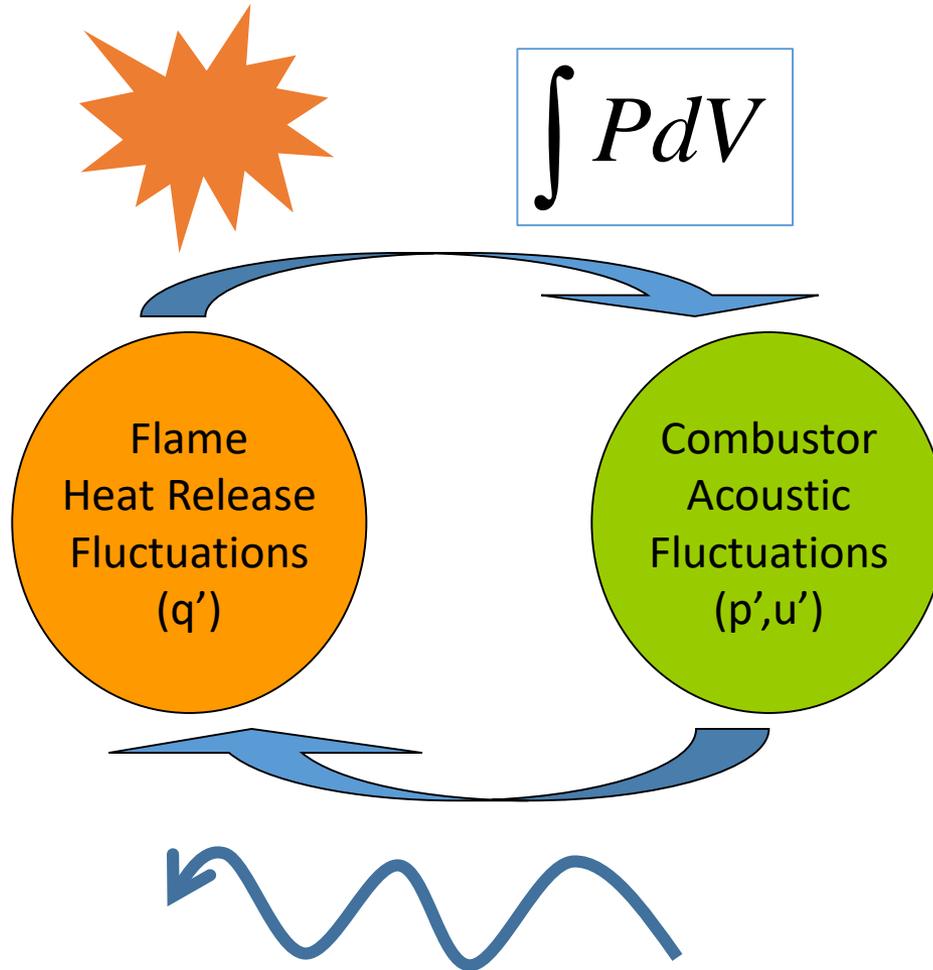
Initially, some perturbation in heat release rate causes *gas expansion* from the flame



The gas expansion does compression work, locally adding energy to the acoustic field (a pressure anti-node)



The energized acoustic field then produces a disturbance, which can excite further heat release rate oscillations

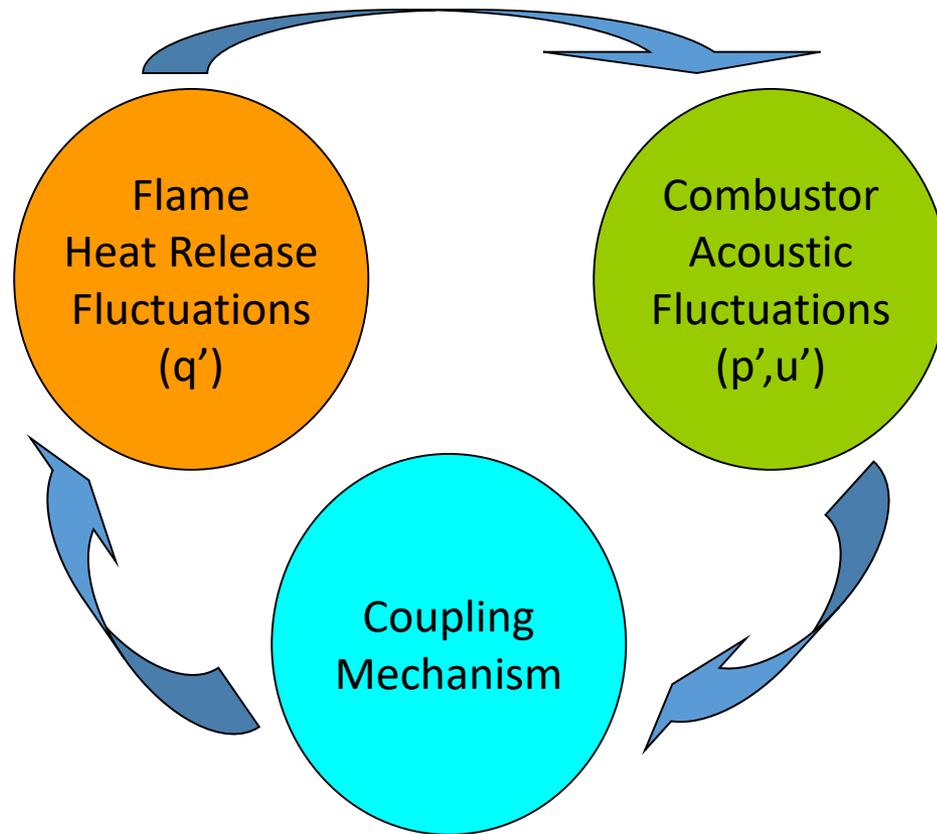


 Combustion instability is a phenomena better heard or “felt” than analyzed for better understanding

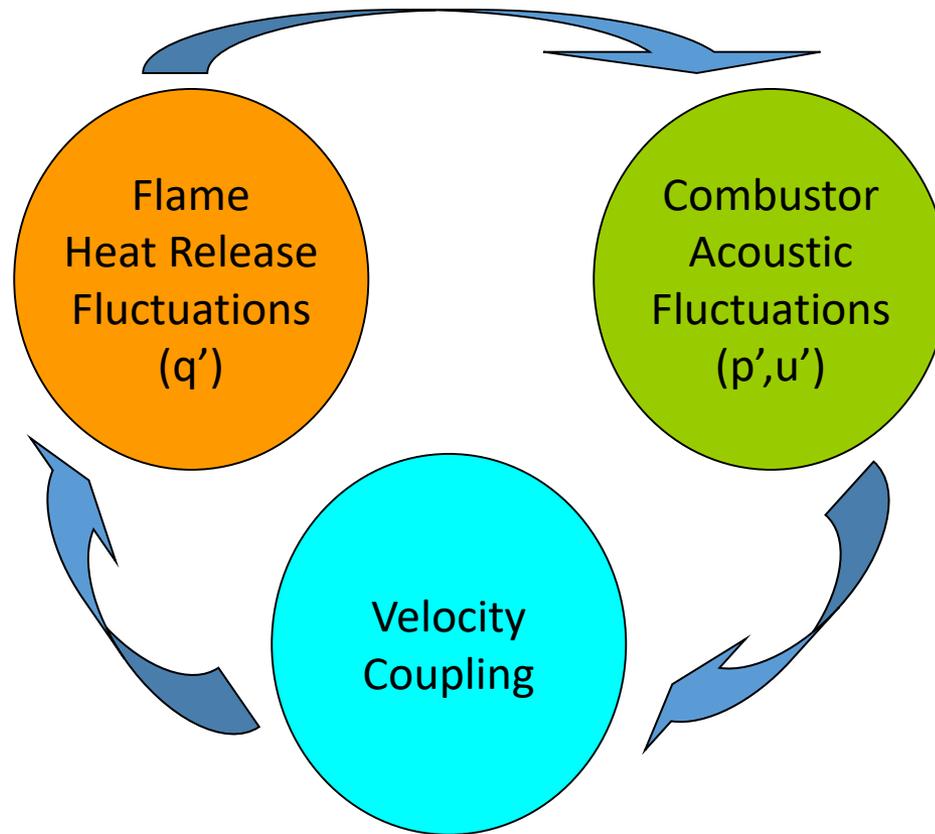
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In real devices the feedback from acoustics to the flame typically includes a “coupling mechanism”

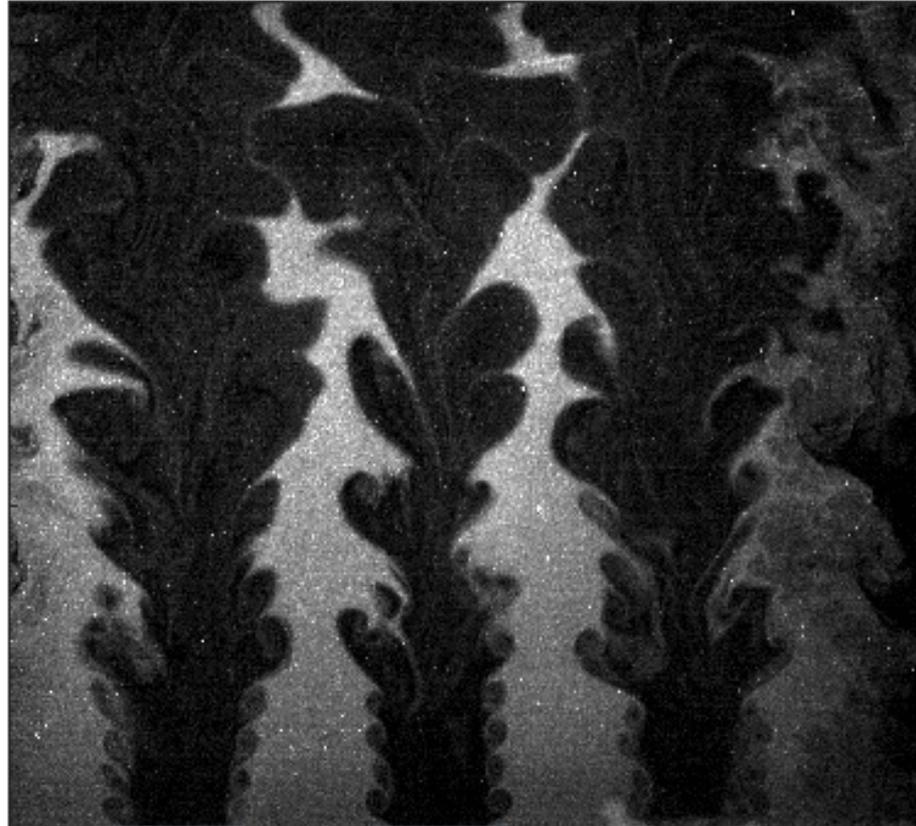


# Velocity coupling: acoustic fluctuations produce vortex roll-up in the flowfield, resulting in flame disturbances



Velocity coupling: acoustic fluctuations produce vortex roll-up in the flowfield, resulting in flame disturbances

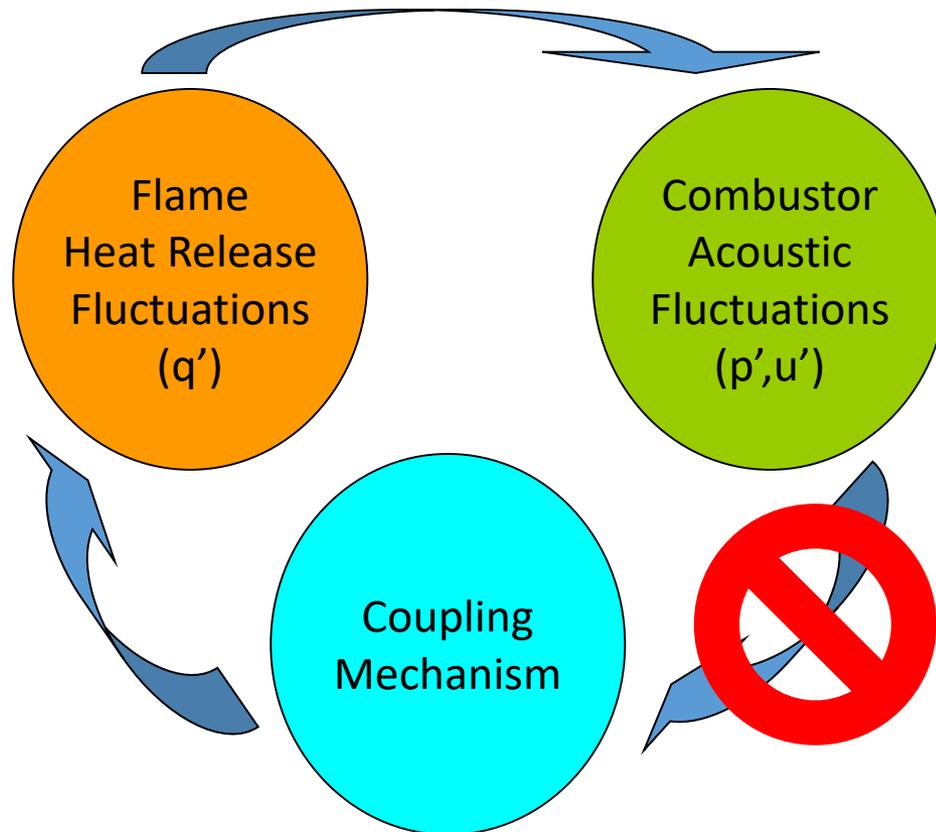
**Frame:1**



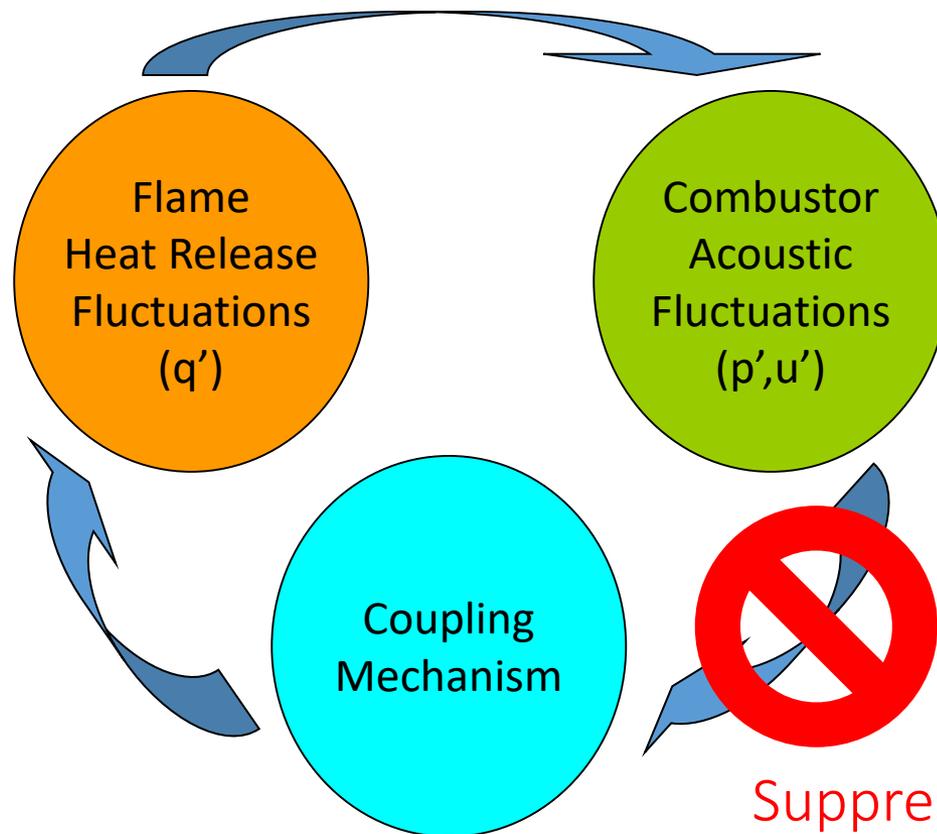
# Velocity coupling: acoustic fluctuations produce vortex roll-up in the flowfield, resulting in flame disturbances



What if we could stop the feedback loop not by changing the flame or acoustics, but changing the flow?

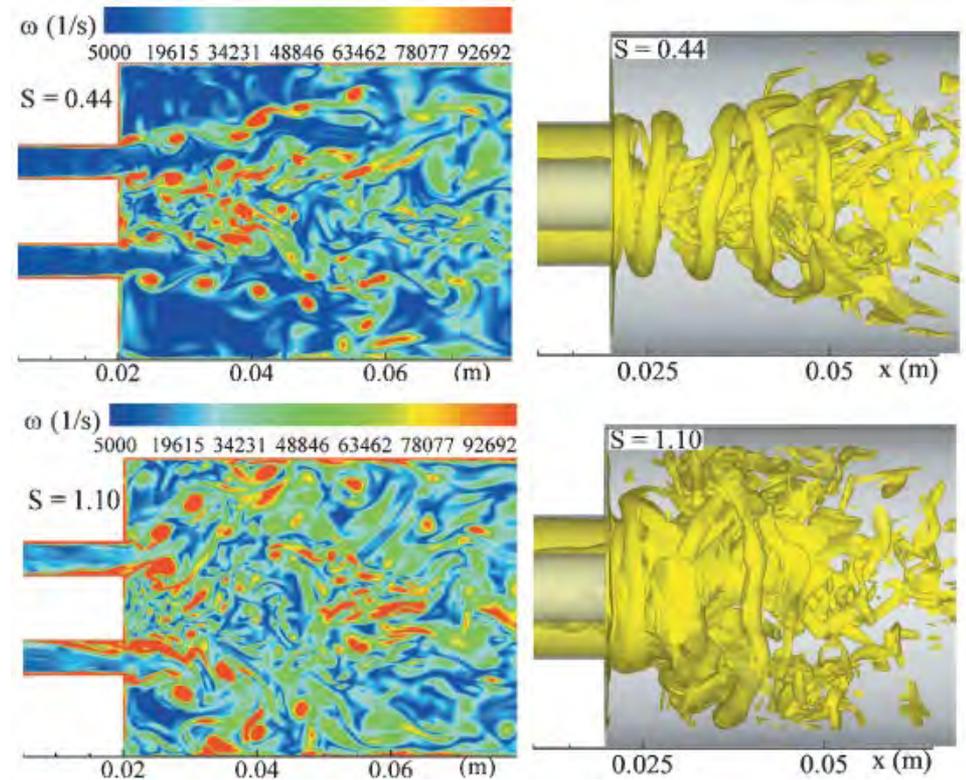
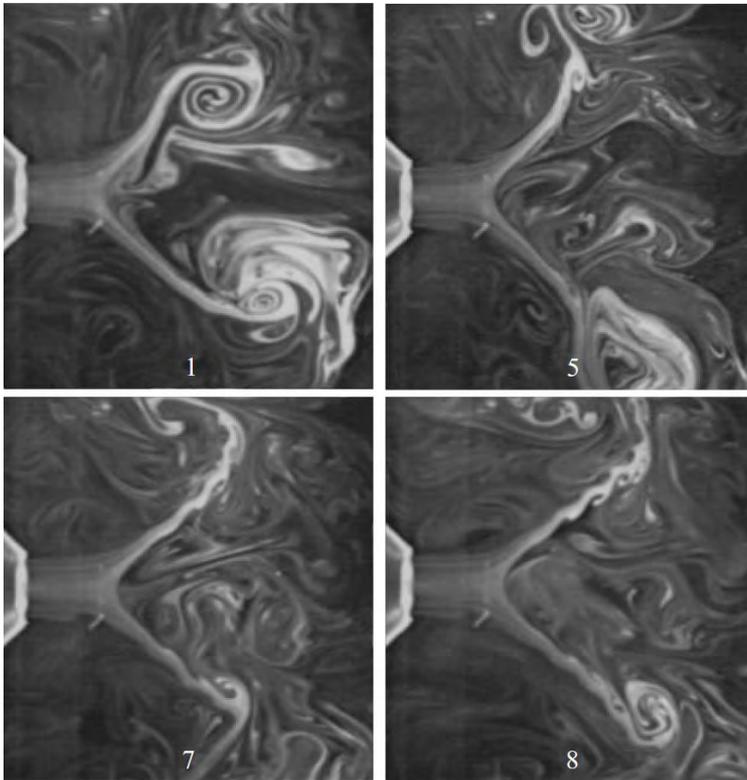


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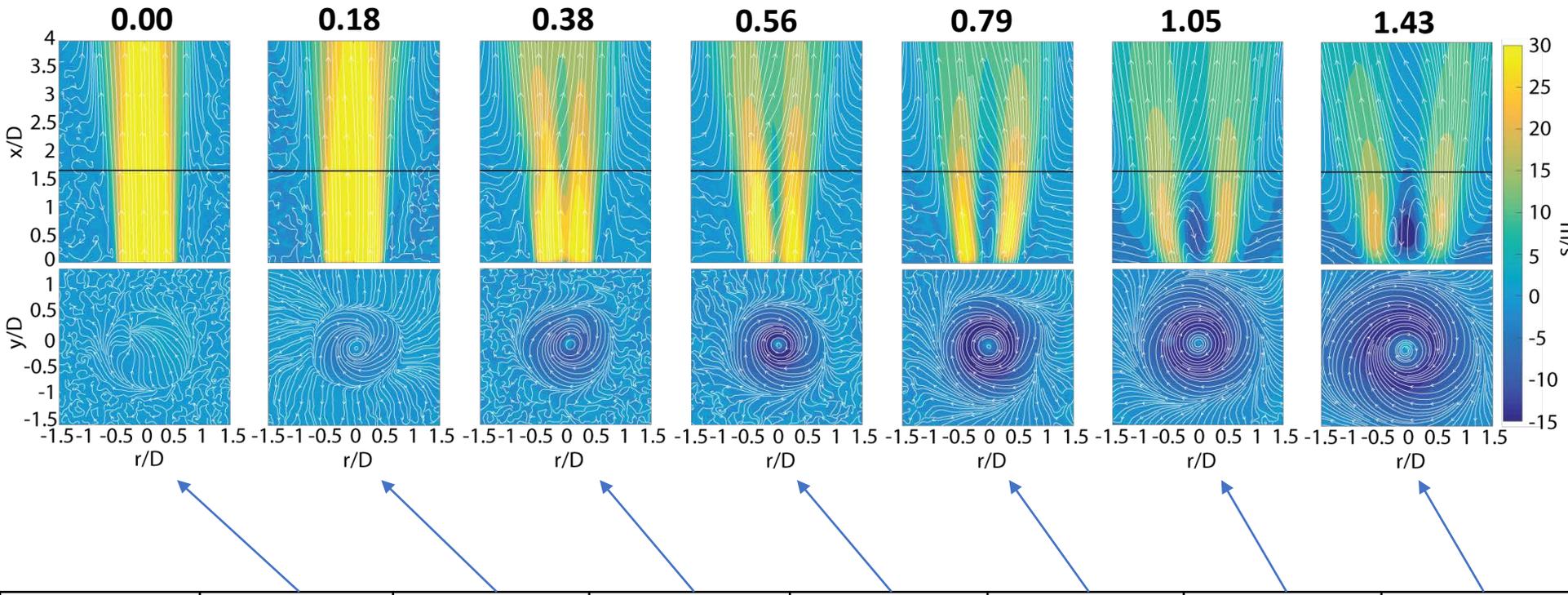
Suppression through  
changes in *shear layer*  
*receptivity*

# Swirling flows are hydrodynamically unstable and susceptible to acoustic forcing in different ways throughout the flow



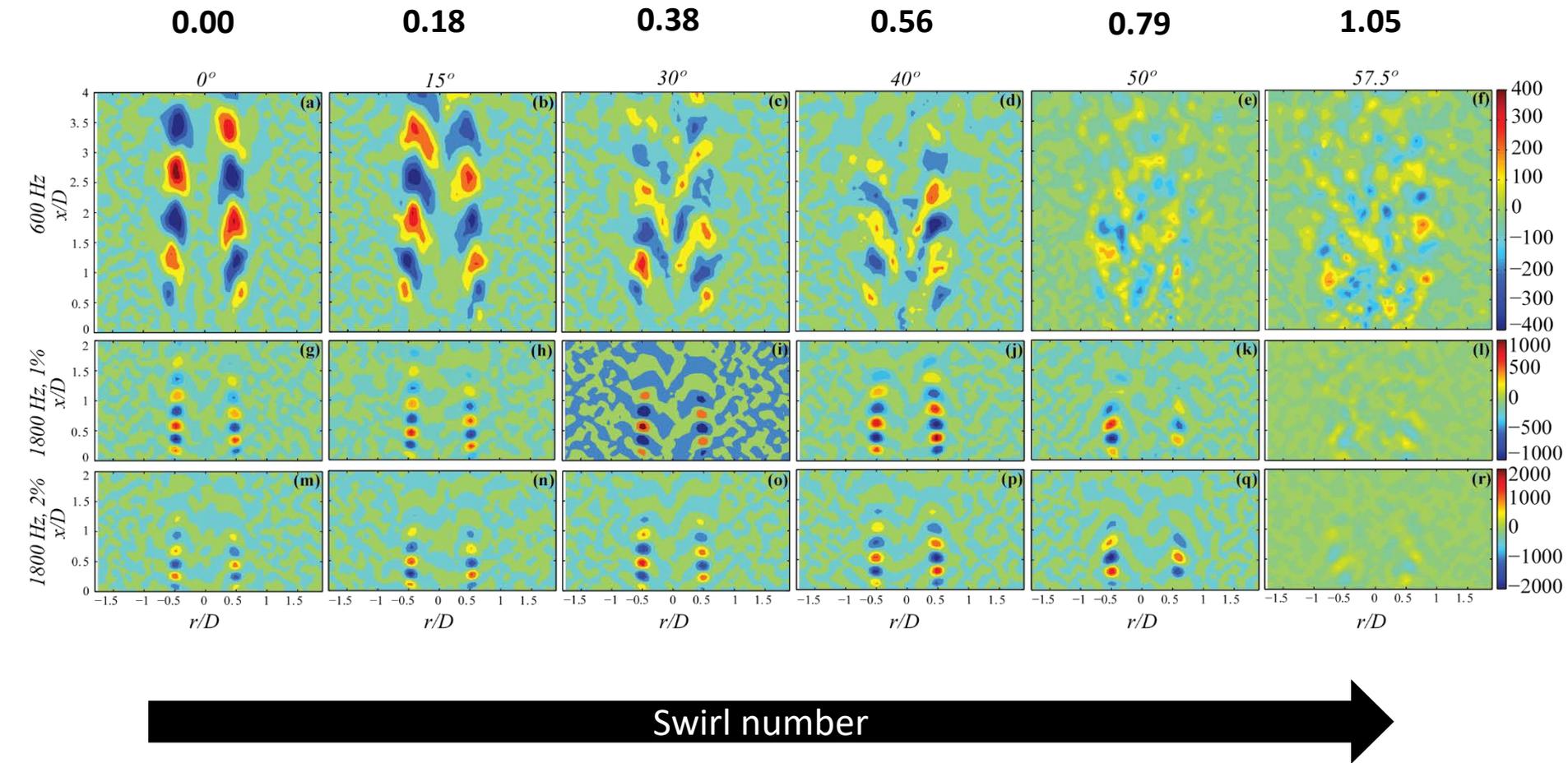


# Structure of swirling flows change significantly with swirl number, inducing vortex breakdown and PVC



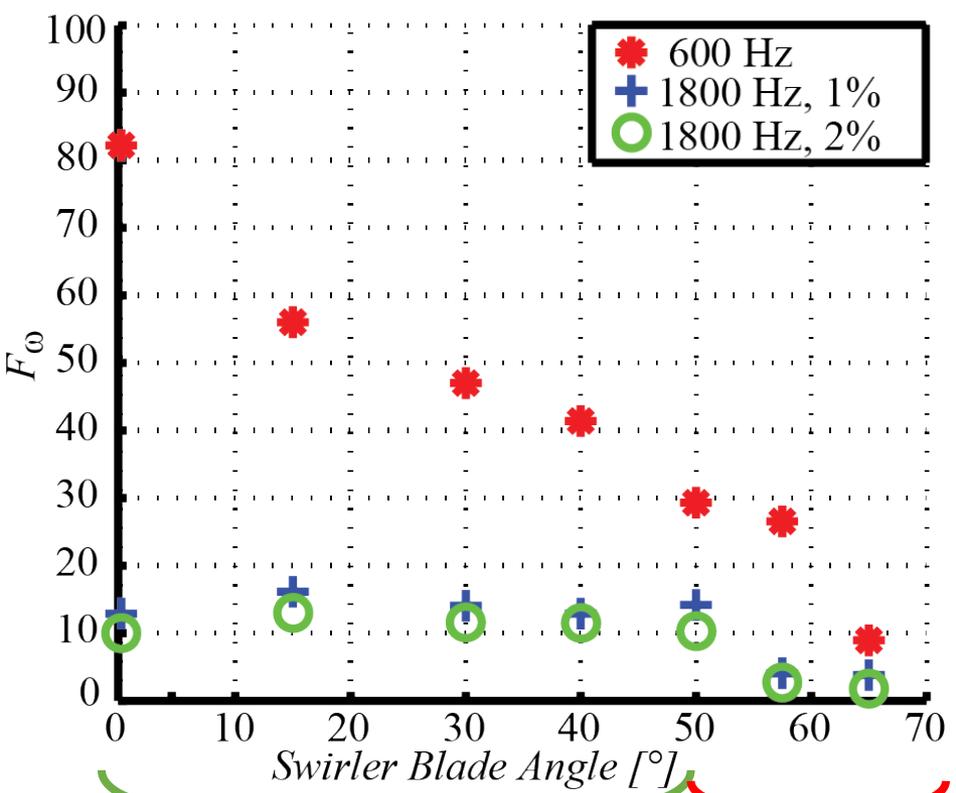
Swirl Number	0.00	0.18	0.38	0.56	0.79	1.05	1.43
Flow State	No VB	No VB	No VB	Intermittent VB	Weak PVC	PVC	Strong PVC
PVC Frequency	-	-	-	-	770–815 Hz	840 Hz	1060 Hz

# Harmonic reconstruction of forced response indicates variation in shear layer receptivity with swirl number, frequency



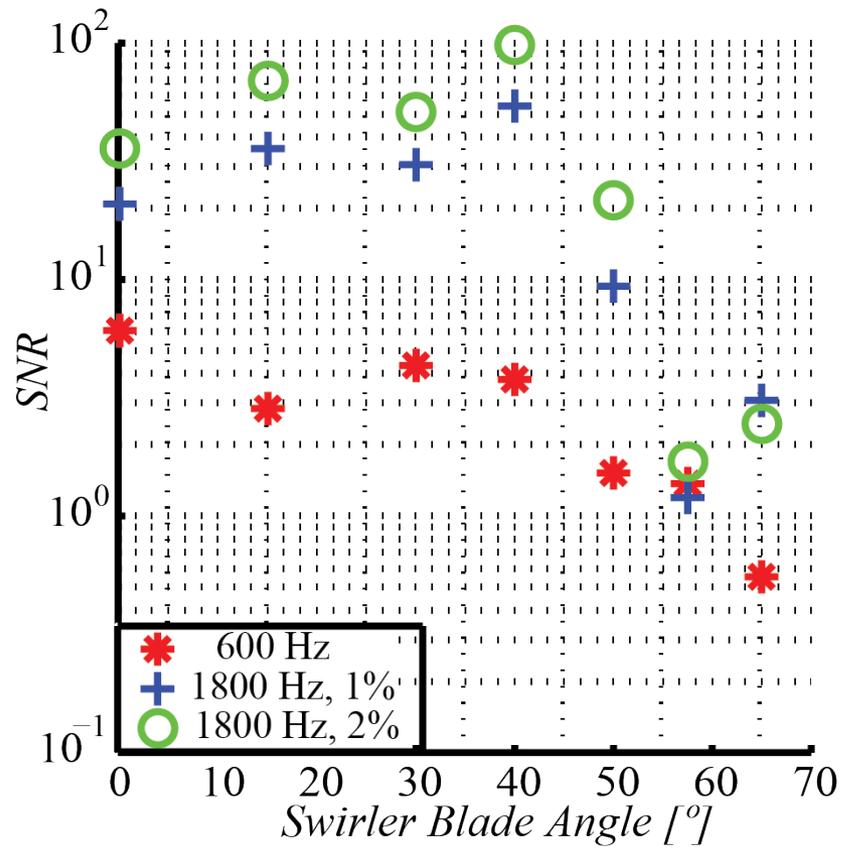
# Acoustic to vortical transfer function shows significant amplification of disturbances except at high swirl number

## Acoustic to Vortical TF

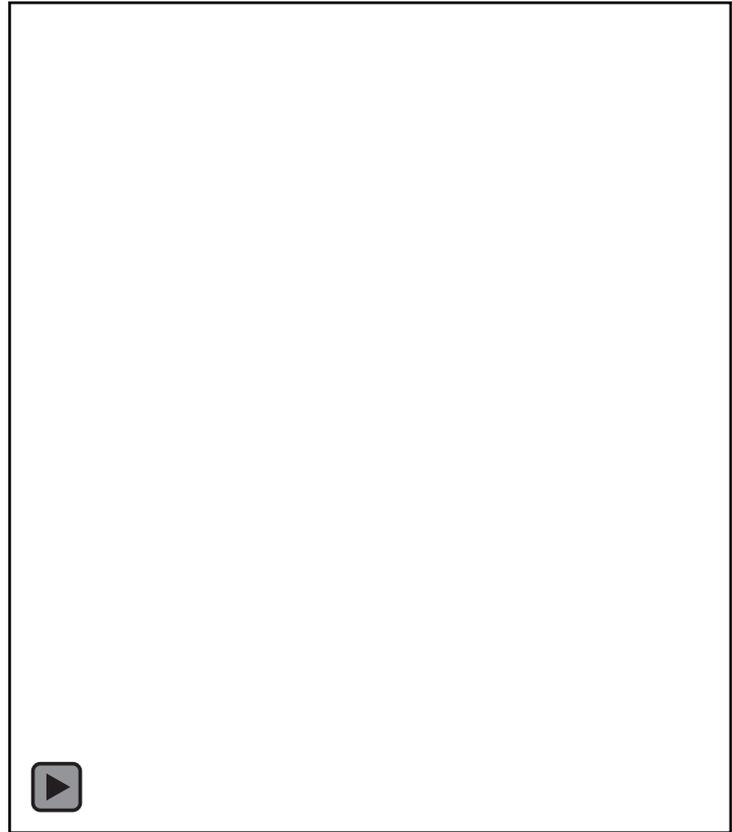
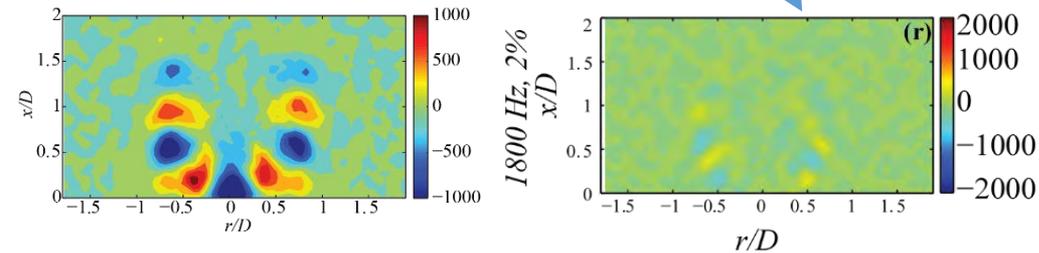
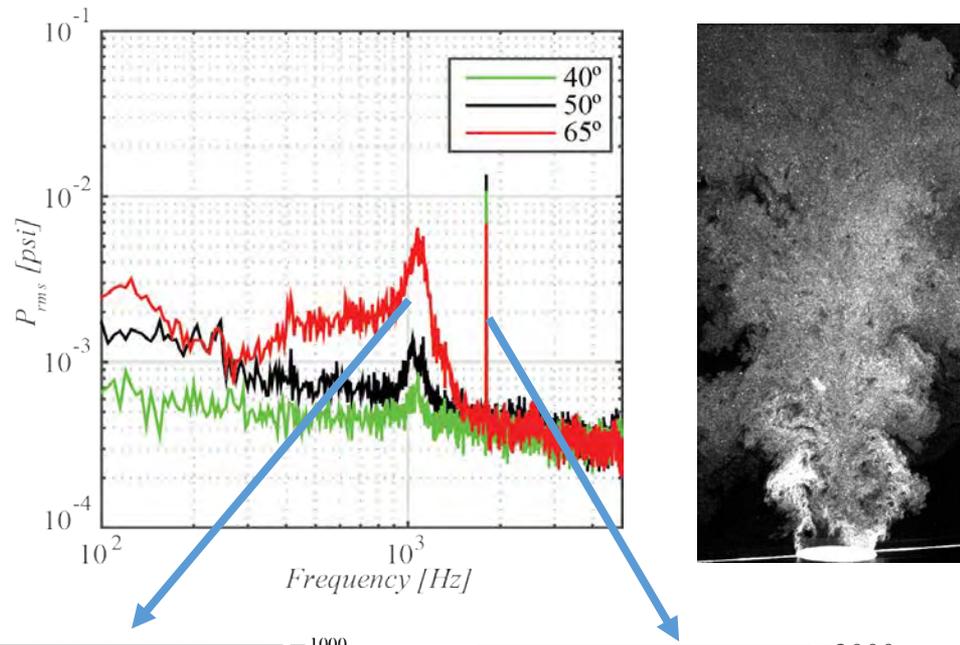


Velocity coupling possible      No velocity coupling

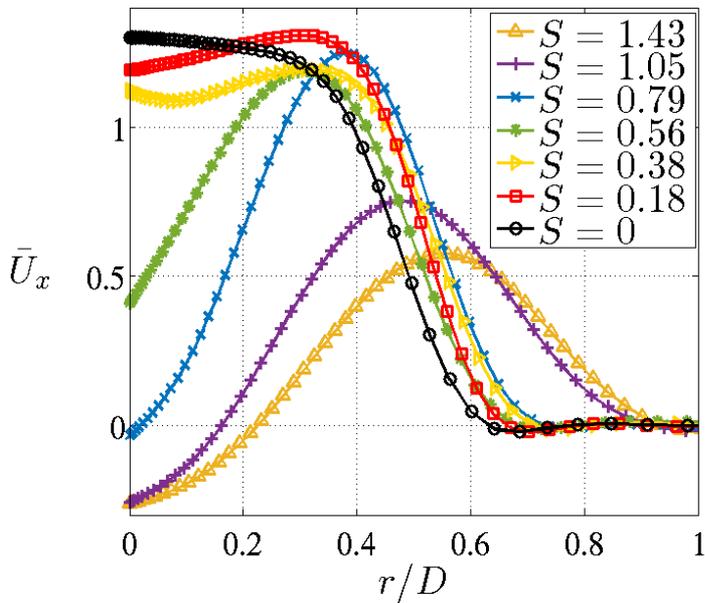
## Vorticity SNR



PVC is present in the flowfield at the three highest swirl numbers; PVC is has significant turbulent kinetic energy



# Stable design through flow design – use of hydrodynamic instability tools to tune flowfield receptivity



Base flow

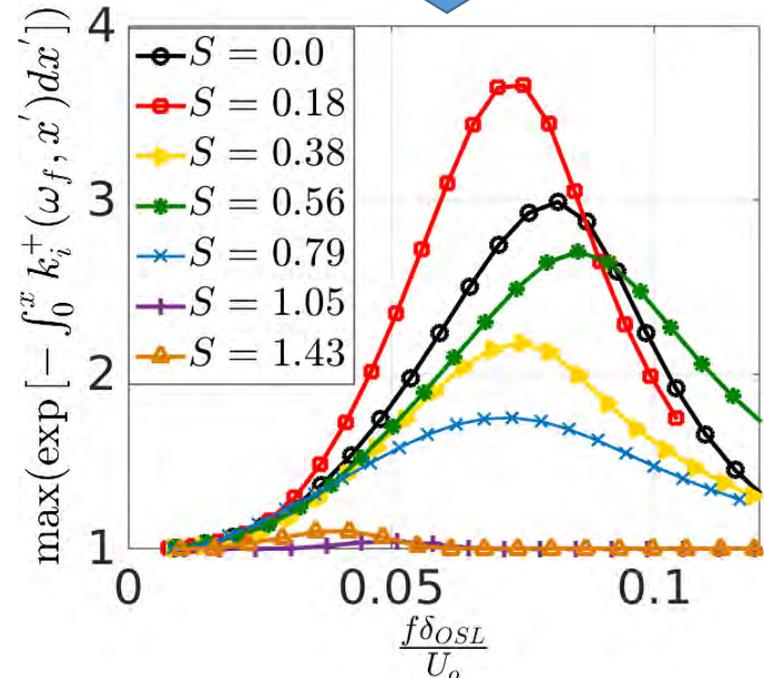
## Weakly non-linear stability calculation

$$\mathbf{q}(r, z, \theta, t) = \bar{\mathbf{q}}(r, z) + \mathbf{q}'_c(r, z, \theta, t) + \mathbf{q}''(r, z, \theta, t)$$

$$\mathcal{B} \frac{\partial \mathbf{q}}{\partial t} + \mathcal{N}\{\mathbf{q}\}\mathbf{q} + S\mathcal{N}^s\{\mathbf{q}\}\mathbf{q} + S^2\mathcal{N}^{ss}\{\mathbf{q}\}\mathbf{q} = \mathcal{L}_v\mathbf{q} + S\mathcal{L}_v^s\mathbf{q}$$

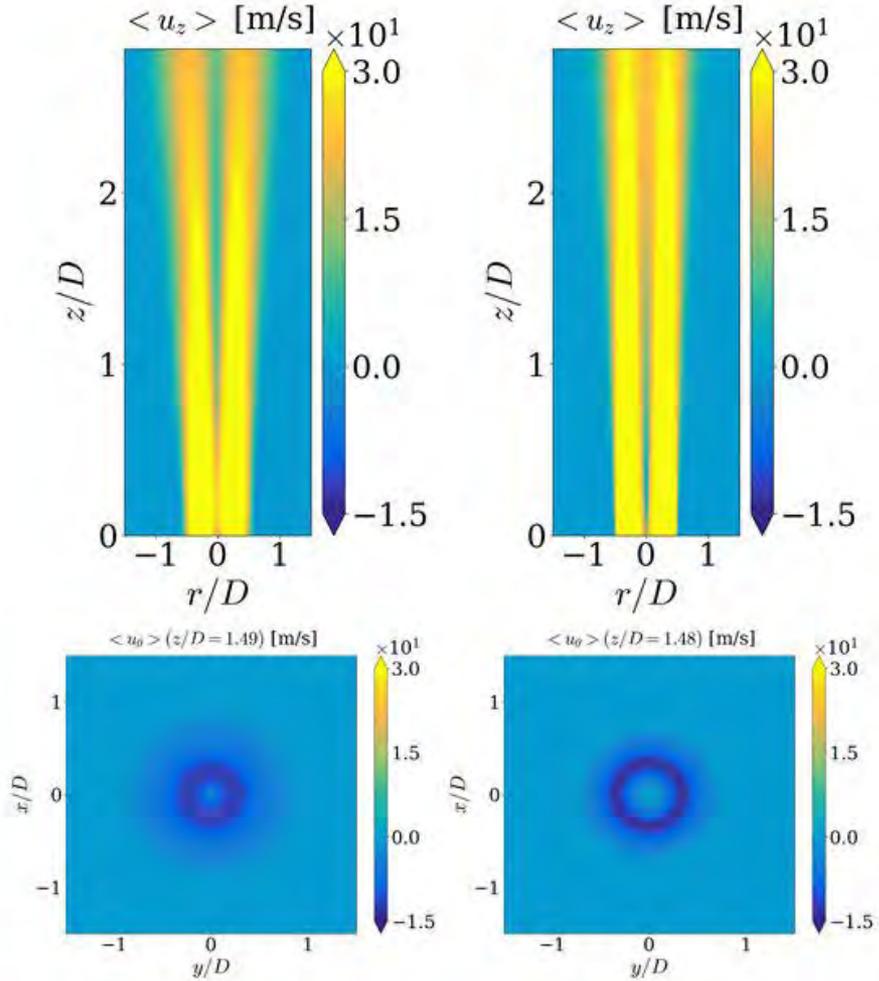


Collaborators: Santosh Hemchandra and Kiran Manoharan, IISc-Bangalore

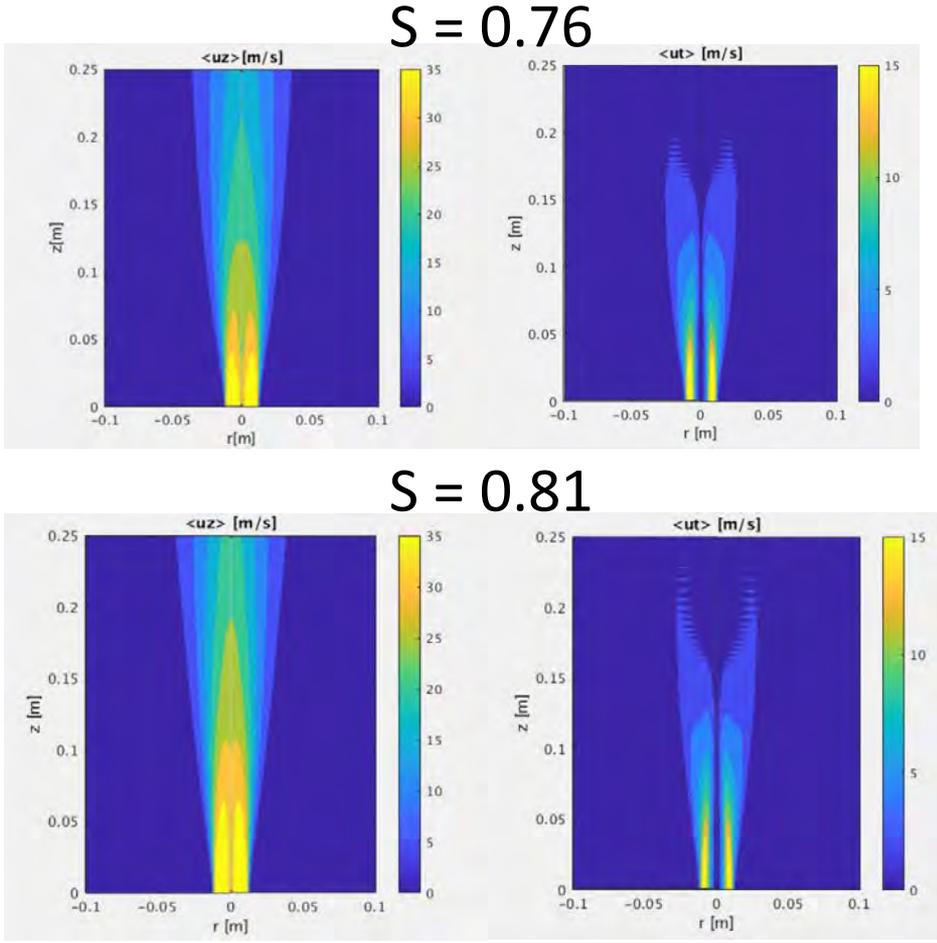


# LES results of turbulent swirling flows

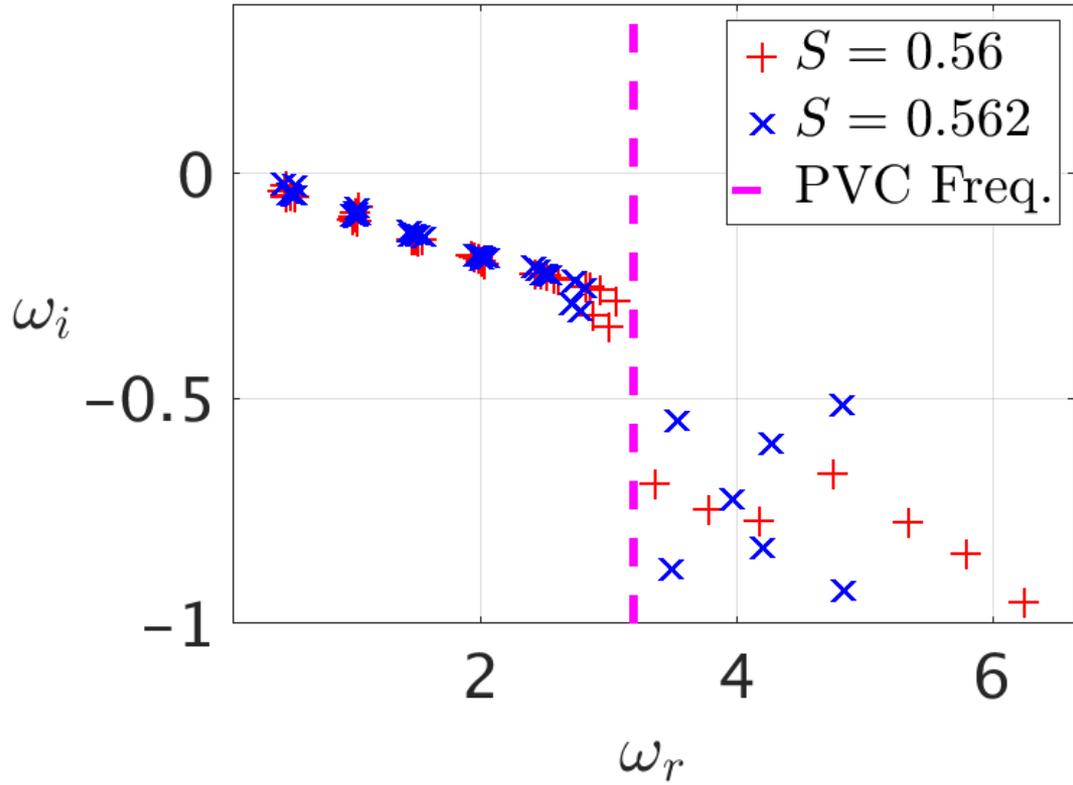
Comparison with measurements for  $S = 0.56$



Other simulation results with higher Swirl numbers



# Hydrodynamic stability analysis – proof of concept



*Table 2. Results of proof-of-concept optimization study.*

<b>S</b>	<b><math>\omega_r</math></b>	<b><math>\omega_i</math></b>
0.56	3.40	-0.68
0.57	3.38	-0.69
0.6	3.03	-0.28
0.65	3.03	-0.27