



# **Fuel Injection Dynamics and Composition Effects on RDE Performance**

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UTSR/NETL

# Summary

- **Title:**
  - Fuel Injection Dynamics and Composition Effects on RDE Performance
- **Funding agency:**
  - University Turbine Systems Research/NETL
  - Funding Opportunity Number: DE-FOA-000117
  - Topic Area 6: [Pressure Gain Combustion R&D](#)
  - [Project manager](#): Robin Ames
  - Award: DE-FE0031228
- **Personnel:**
  - [PI](#): Mirko Gamba, University of Michigan
  - [Co-I](#): Venkat Raman, University of Michigan
  - [Students](#) currently involved:
    - Fabian Chacon
    - James Duvall
    - Takuma Sato
    - Supraj Prakash
  - [Key external collaborators](#):
    - Dr. John Hoke, Innovative Scientific Solution, Inc. (ISSI)
    - Dr. Fred Schauer, AFRL WP
    - Dr. Venkat Tangirala , GE
    - Dr. William Hargus, AFRL/Edwards
    - Drs. Adam Holley and Peter Cocks, United Technology Research Center (UTRC)
    - Kyle McDevitt, Williams International

# Outline

- **Programmatic overview**
- **Introduction to the problem and general approach**
- **Experimental activities**
- **Computational activities**
- **Interactions and collaborations**

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- **Programmatic overview**
- **Introduction to the problem and general approach**
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# Overarching objectives

- **Objective 1:**

Develop a comprehensive understanding of injector dynamics, coupling with diffuser back-reflections, and their impact on RDE mixing, operation and performance.

- **Objective 2:**

Develop a comprehensive understanding of multi-component fuels (syngas and hydrocarbon blends) on RDE detonation structure and propagation, operation and performance.

- **Objective 3:**

Develop advanced diagnostics and predictive computational models for studying detonation propagation in RDEs, with arbitrary fuel composition and flow configuration.

# Expected outcomes: RDE physics advancements

- **Outcome 1:**

**Fundamental physical understanding of detonation-induced dynamics of practical injection systems:**

- Coupling with diffuser geometry;
- Their effect on RDE mixing, detonation structure, operability and performance.

- **Outcome 2:**

**Fundamental physical understanding of multicomponent fuel (MCF) chemistry effects:**

- Detonation wave structure and propagation
- RDE operability and performance;

# Expected outcomes: RDE methods advancements

- **Outcome 3:**

- Detailed experimental measurements under various injection schemes and MCFs**

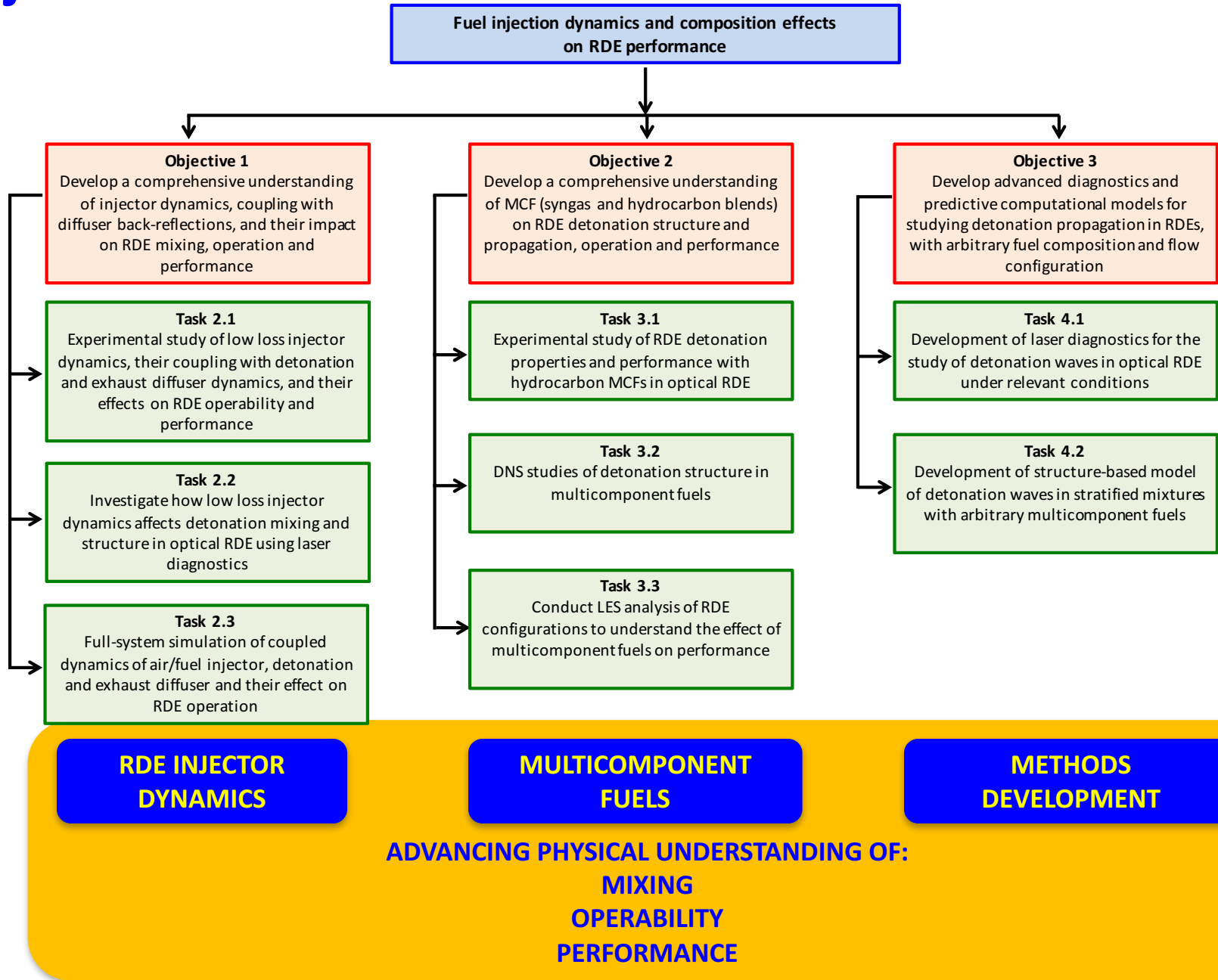
- RDE performance and detonation structure
      - Creation of experimental databases
    - MCFs representative of syngas and natural gas characterized by a range of detonability properties, ignition delays and Wobbe index
      - Relevance to DOE and industry programs

- **Outcome 4:**

- Computationally efficient detonation models for MCFs for use with DNS/LES modeling of detonations in relevant full-system RDEs under practical MCFs**

- Implementation into open-source platforms; e.g., openFoam
    - Transfer of detonation computational models to industry

# Objectives and tasks





# Timeline of the project

## Task 1

## Task 2

## Task 3

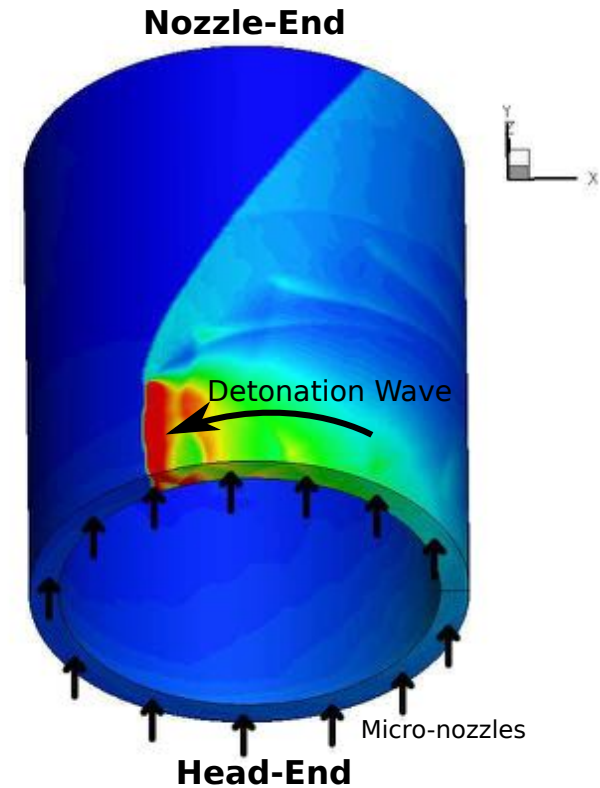
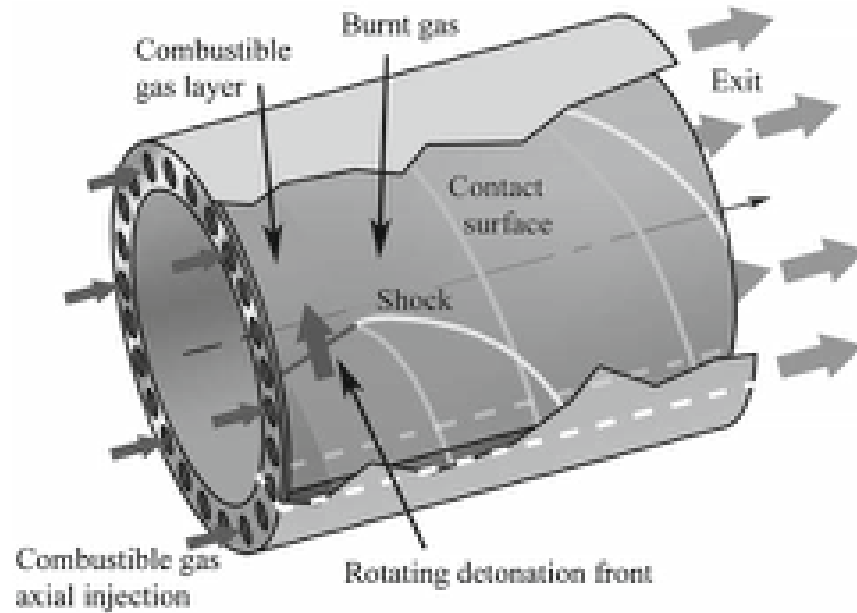
## Task 4

Task	Description	Start	Finish	2017	2018				2019				2020			
				Q1	Q2	Q3	Q4	Q1	Q2	Q3	Q4	Q1	Q2	Q3	Q4	
1.0	Project meeting and planning	10/17	9/20													
1.1	Project meetings and progress reports	10/17	9/20													
2.0	Study of the effects of low loss injector dynamics on detonation mixing and structure and RDE operability and performance	10/17	9/20													
2.1	Experimental study of low loss injector dynamics, their coupling with detonation and exhaust diffuser dynamics, and their effects on RDE operability and performance	10/17	3/19													
2.2	Investigate how low loss injector dynamics affects detonation mixing and structure in optical RDE using laser diagnostics	4/18	3/20													
2.3	Full-system simulation of coupled dynamics of air/fuel injector, detonation and exhaust diffuser and their effect on RDE operation	10/18	9/20													
3.0	Study of the effects of relevant multicomponent hydrocarbon fuels on detonation structure, RDE operability and performance	4/18	9/20													
3.1	Experimental study of RDE detonation properties and performance with hydrocarbon MCFs in optical RDE	10/18	9/20													
3.2	DNS studies of detonation structure in multicomponent fuels	4/18	9/19													
3.3	Conduct LES analysis of RDE configurations to understand the effect of multicomponent fuels on performance	10/18	9/20													
4.0	Develop advanced diagnostics and structure-based detonation models for multicomponent fuels	10/17	9/20													
4.1	Development of advanced laser diagnostics for the study of detonation waves in optical RDE under relevant conditions	10/17	9/19													
4.3	Development of structure-based model of detonation waves in stratified mixtures with arbitrary multicomponent fuels	10/17	3/19													

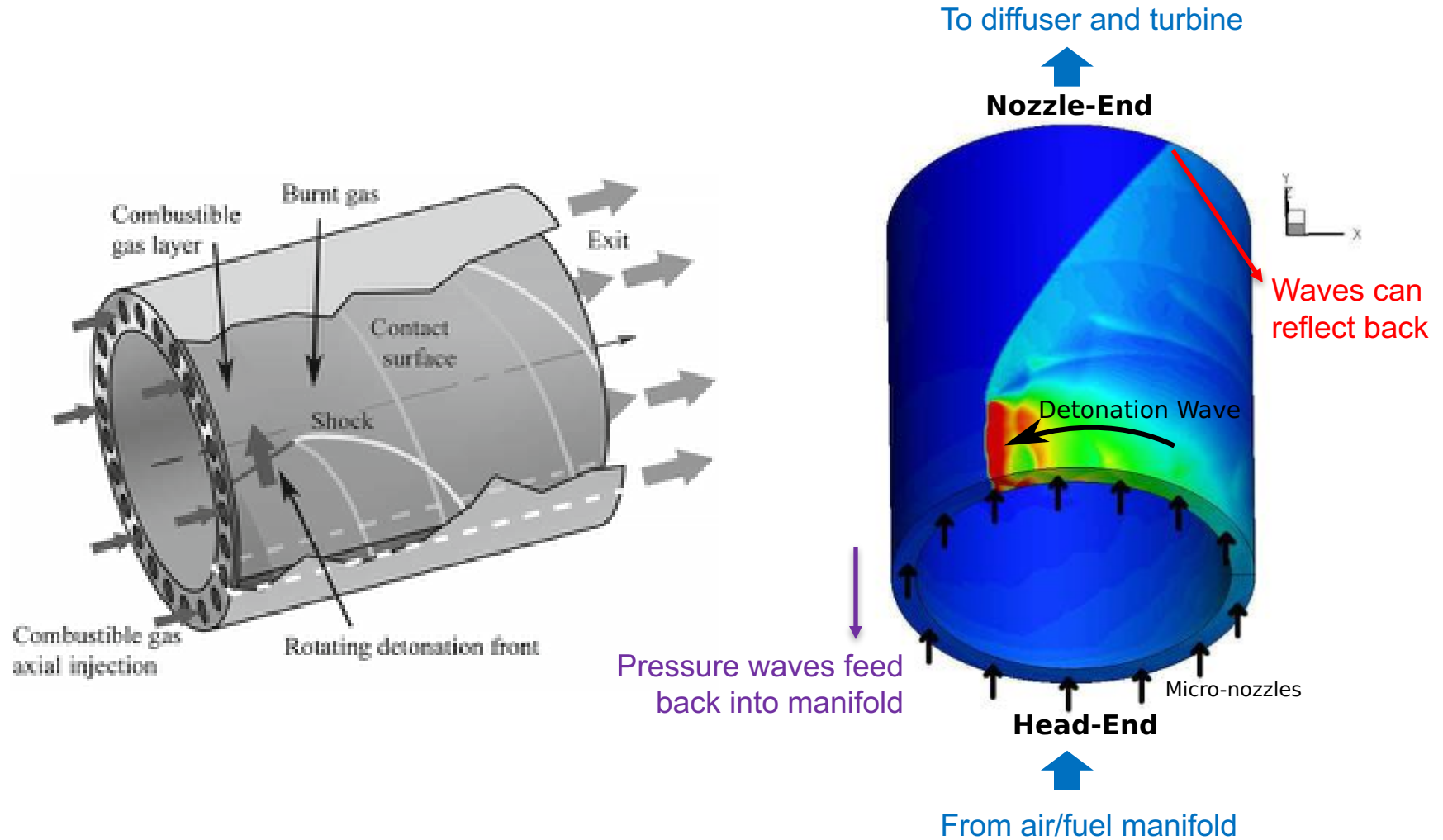
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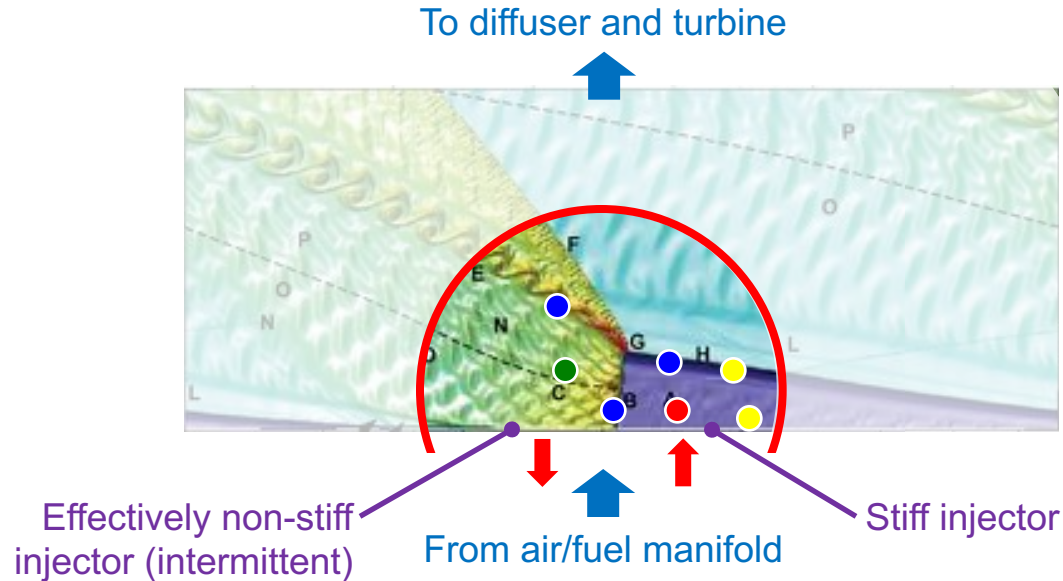
# Overview of RDE operation and Pressure Gain (PG)



# Overview of RDE operation and Pressure Gain (PG)



# Coupling, dynamics and loss of pressure gain



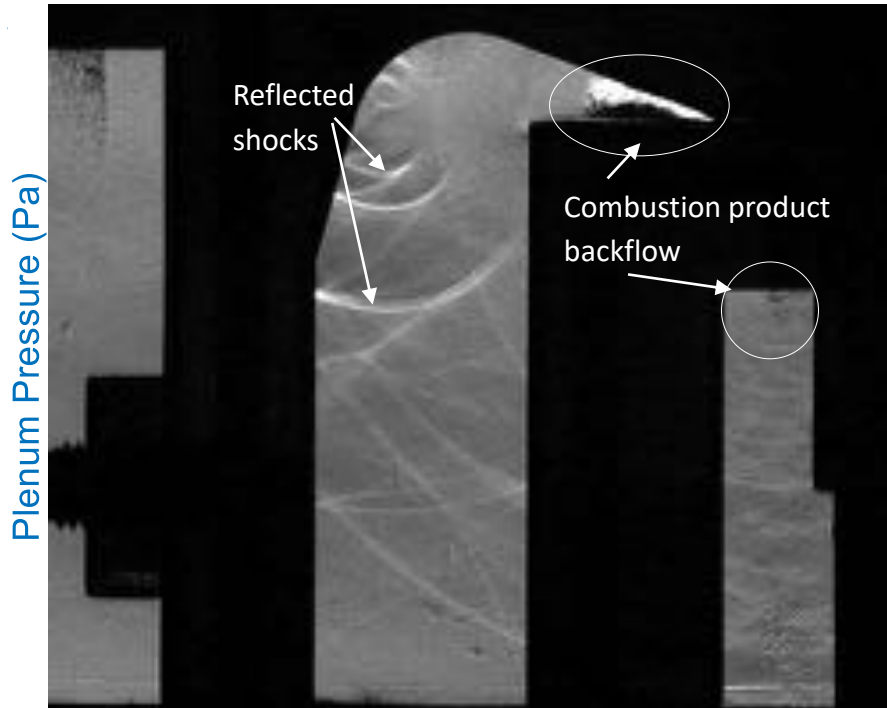
- **Unsteady operation of injection system**

- Injector effectively transition from a stiff to a non-stiff injector
- Post-detonation products backflow into plenums
- Excite plenum dynamics

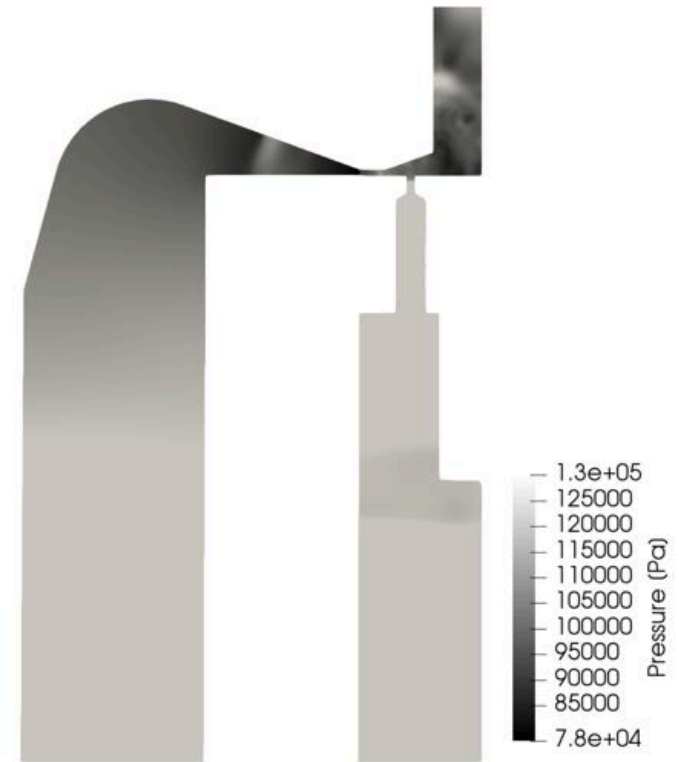
- **Strongly coupled system**

- Response of injection system to varying
- Back-reflections from diffuser (impedance mismatch and wave reflections)
- Mixing dynamics and effectiveness
  - Incomplete fuel/air mixing
  - Fuel/air charge stratification
- Detonation wave dynamics and structure
  - Mixture leakage (incomplete heat release)
  - Parasitic combustion

# Example of upstream propagation of blast wave

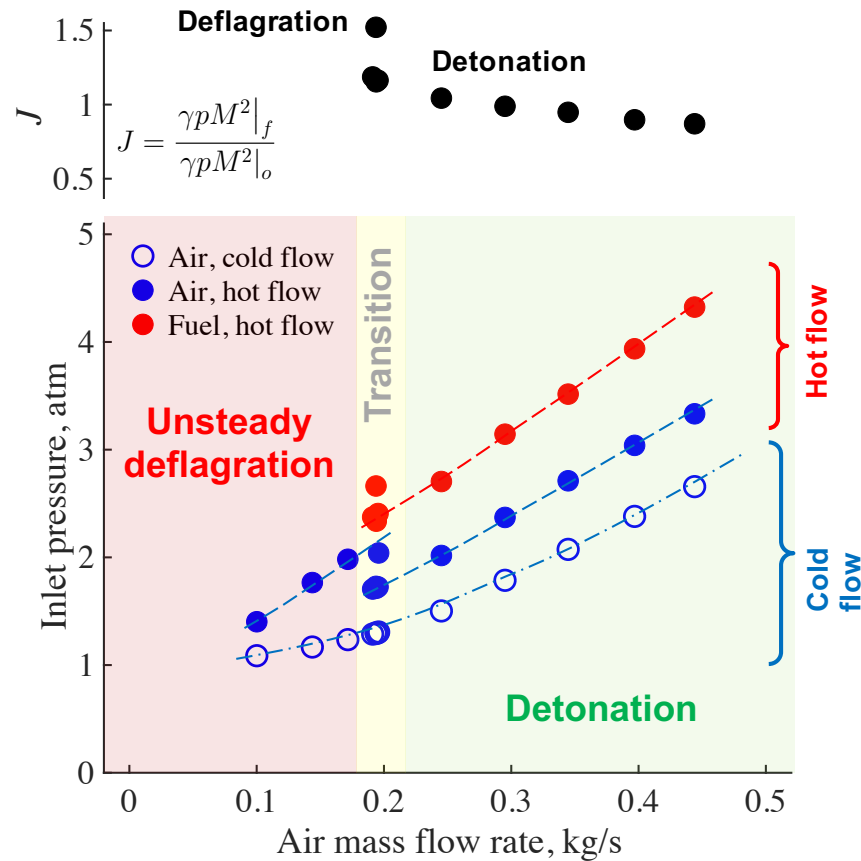
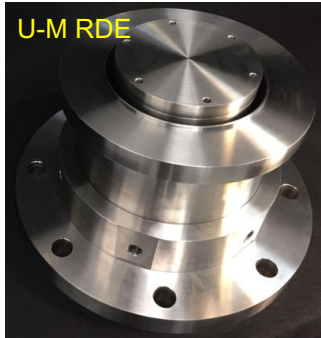


From NETL experiments  
(Ferguson) on AFRL injector



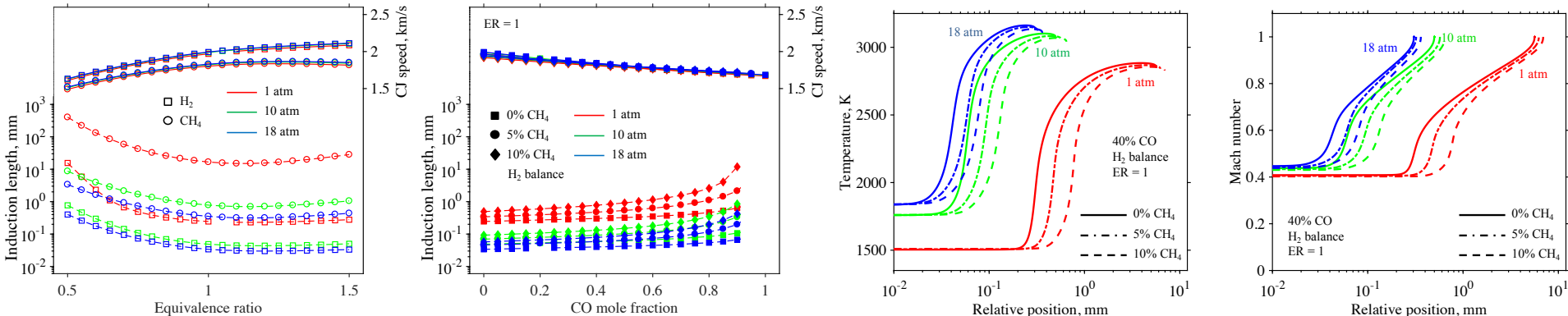
CFD computation at U-M

# Example of detonation chamber / plenum coupling



# Fuel relevance: Toward operation with NG and syngas

- **Most of work conducted so far:**
  - Hydrogen/air operation
  - Stable detonation in lab-scale RDE enabled by favorable detonation properties
    - Detonability
    - Cell size
- **Application (energy conversion systems) require:**
  - Natural gas and/or syngas operation
  - Fuel flexibility and broad operation (e.g., from part to full load)
- **Anticipated challenges**
  - Stabilization of detonation wave
    - Limits imposed by detonation cell size
  - Fuel blend of relevant species ( $\text{H}_2/\text{CH}_4/\text{CO}$ ) impacts detonation properties
    - E.g., induction length (cell size and stability) strongly reduced with  $\text{H}_2$  addition (contrary to propane/ethylene)
    - E.g., presence of  $\text{CO}_2$  shifts CO to  $\text{CO}_2$  conversion equilibrium, impacting heat release
    - E.g., difference in oxidation time scale of different components can affect overall structure





# Overarching objectives

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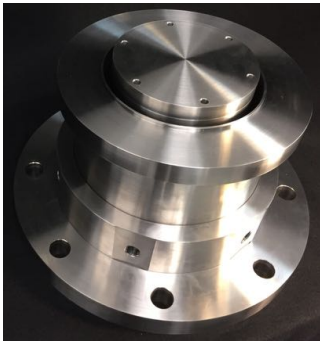
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# RDE Experimental Infrastructure at U-M

- **Injector sector subassembly**

- Sector of RDE injector for mixing effectiveness measurements

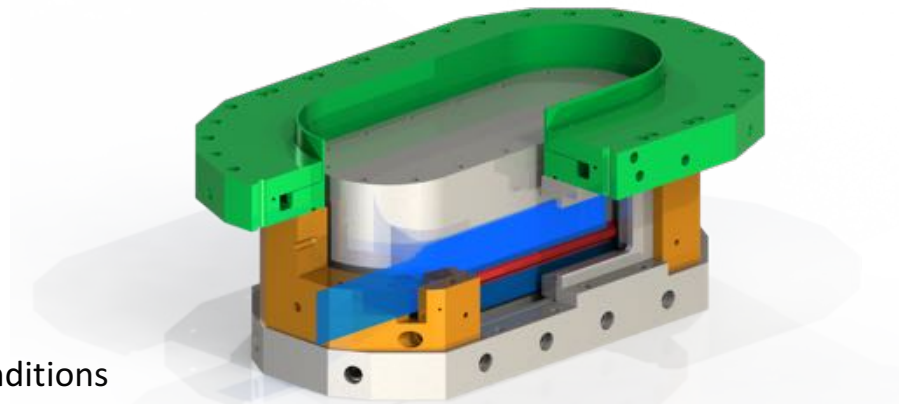


- **Reduced-scale RDE (6" RDE platform)**

- Developed under previous projects
- Operational with  $H_2$ /Air, various flow rates and equivalence ratios
- Will be expanded to include:
  - MCFs capability
  - Additional instrumentation to investigate RDE dynamics

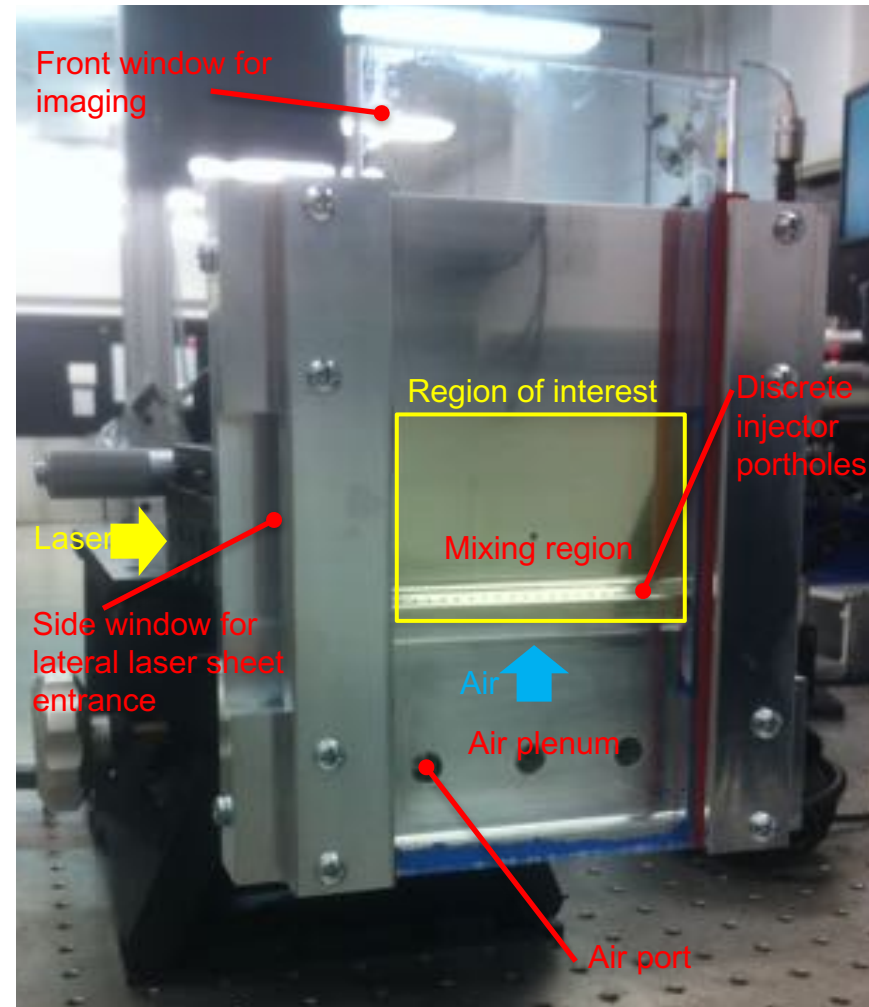
- **Optical RDE (Race-Track RDE)**

- Fabrication being completed under previous projects
- Equivalent to 12" round RDE
- Used for flowfield measurements under RDE relevant conditions

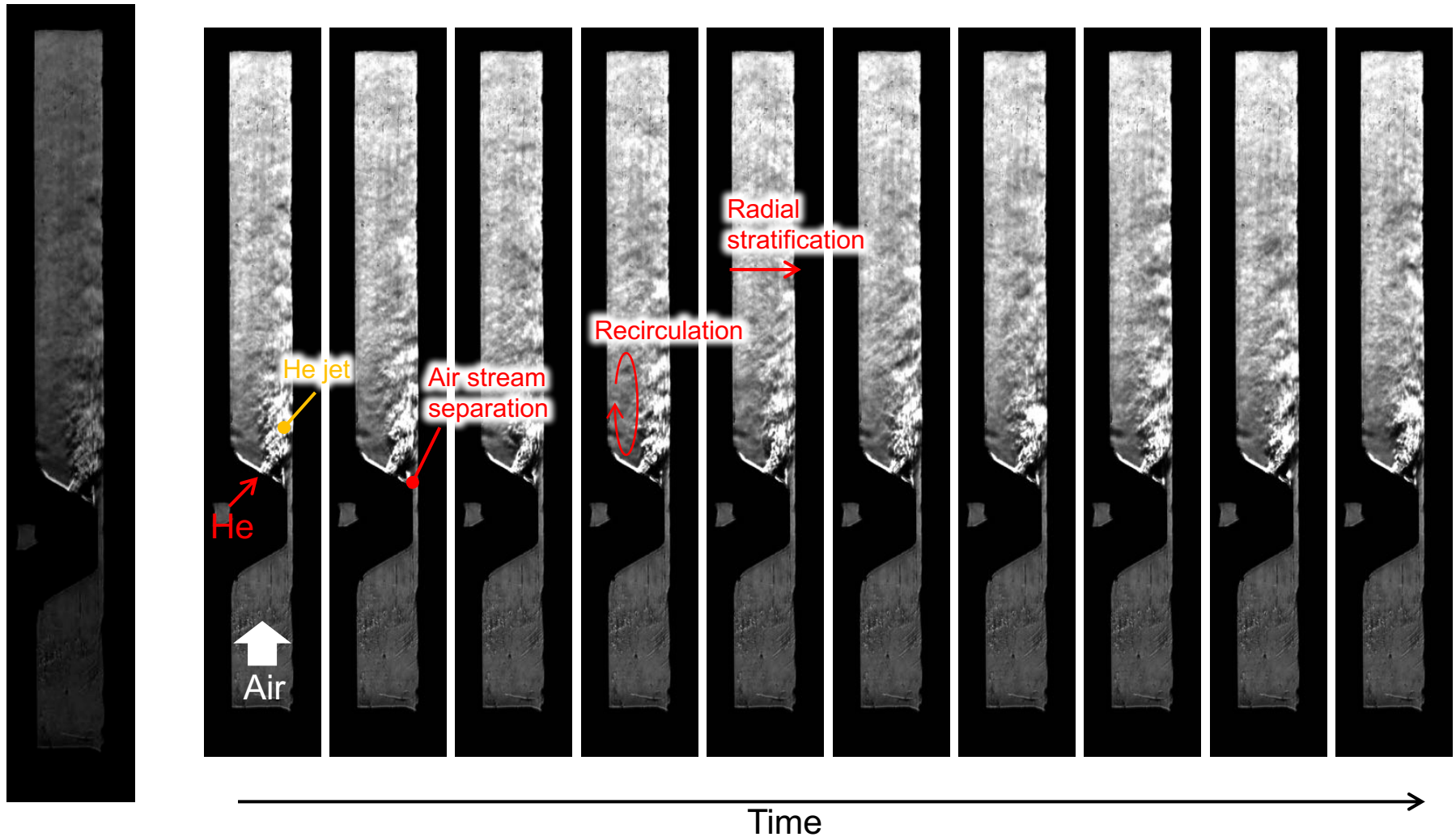


# Injector sector example (photograph)

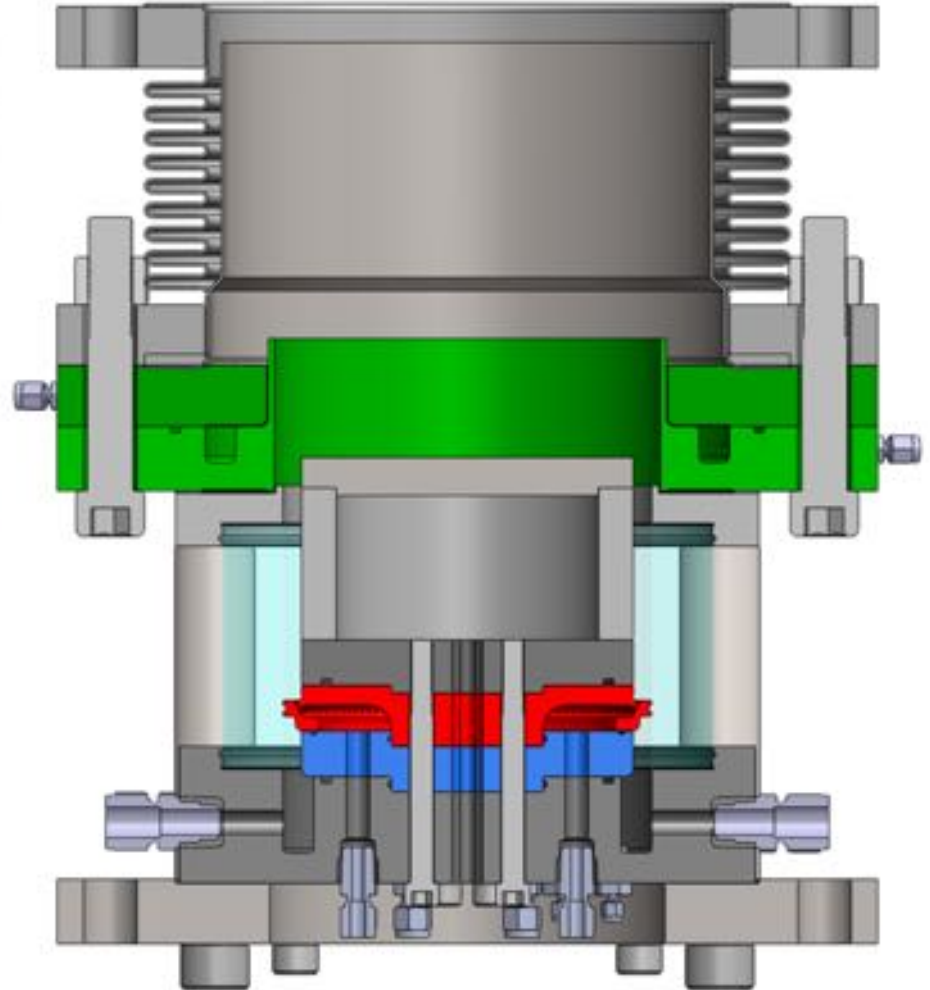
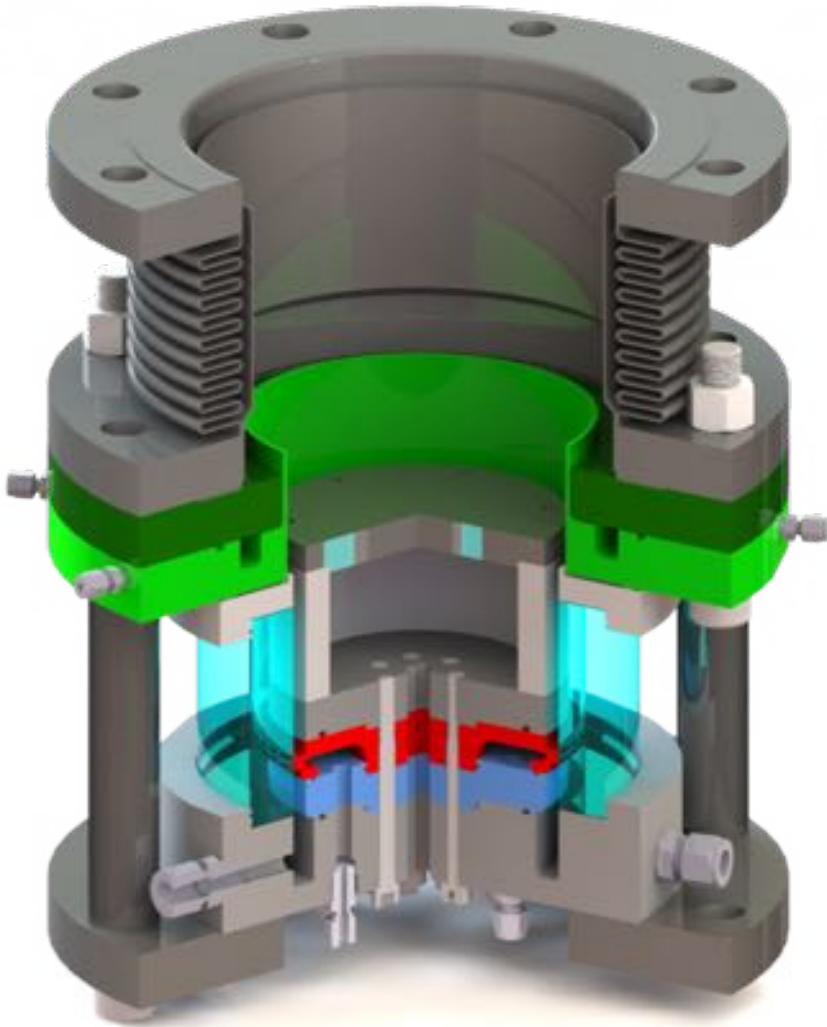
- **Sector of 6" round RDE geometry**
  - Pintle geometry is identical to RDE's, just unwrapped
  - Air plenum geometry is different than RDE's
  - Equivalent length about 1/8 of circumference of circular RDE
  - Optical access for laser diagnostics
- **Objective**
  - Conduct flow visualization and non-reacting mixing measurements under different (jet-to-air) velocity and density ratios (momentum flux ratio)
- **This system can be used in support of RDE measurements (flow visualization and mixing)**
  - Used to perform flow structure imaging and mixture fraction measurement using tracer-based laser induced fluorescence measurements
  - Variation of injection design
  - Parametric variation of velocity and density ratios
- **Can be available to project if needed**



# Schlieren imaging to identify flow structure (non-reacting mixing)



## 3-D rendering of 6" round RDE system

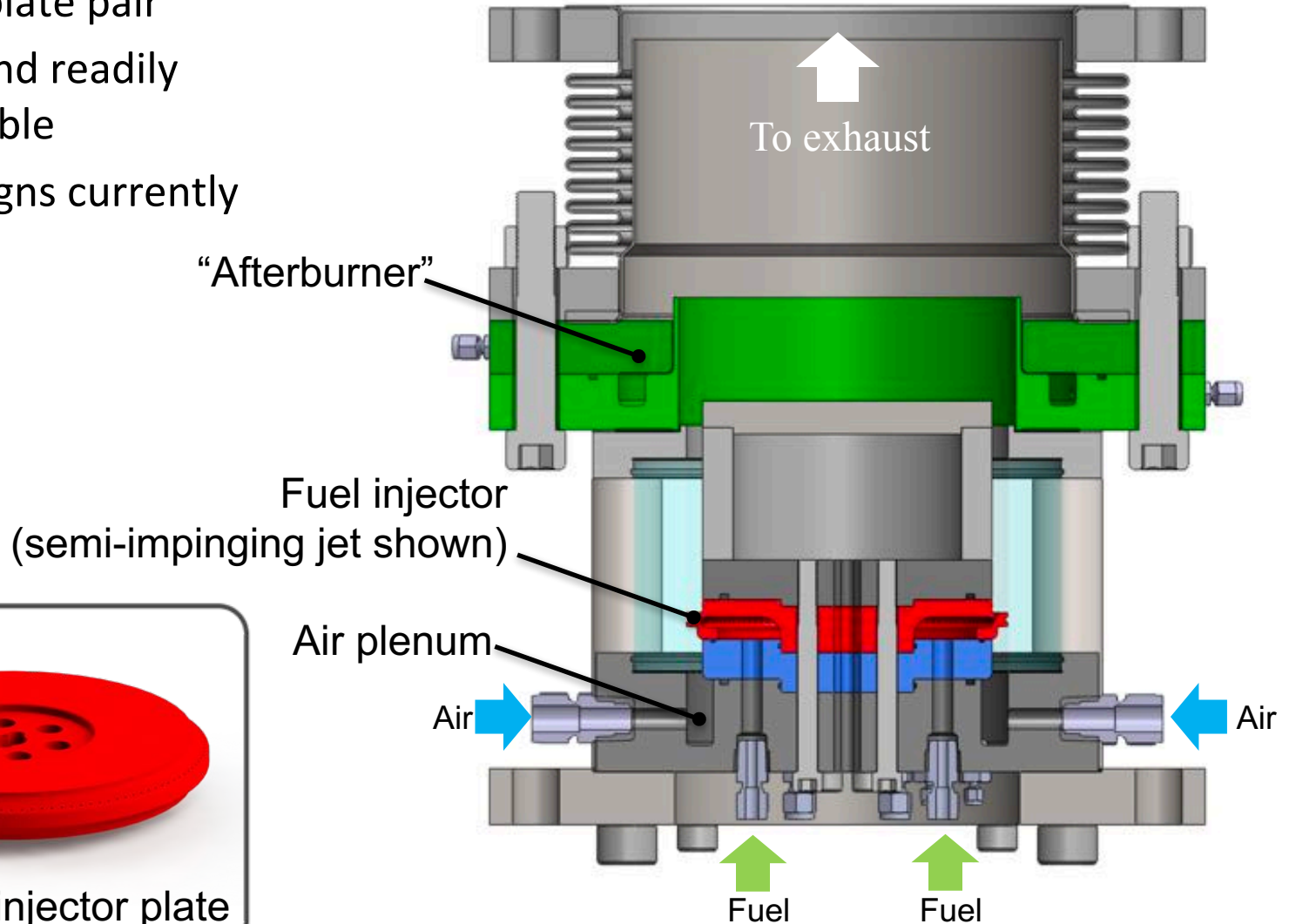
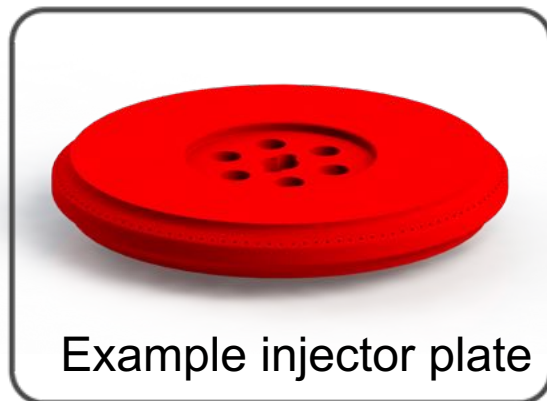




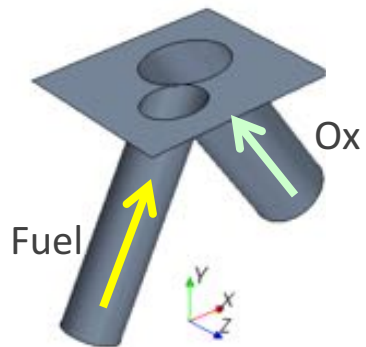
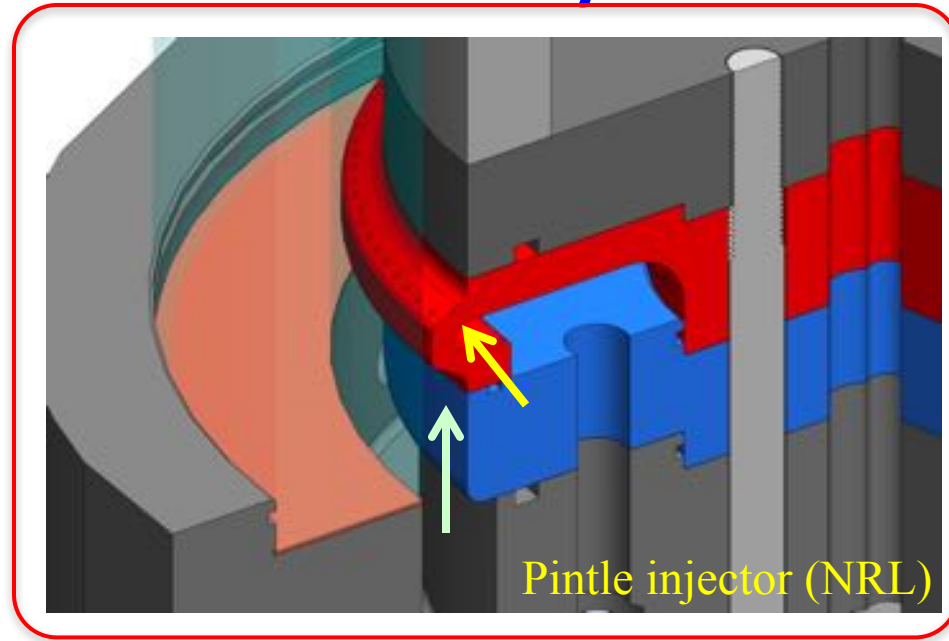
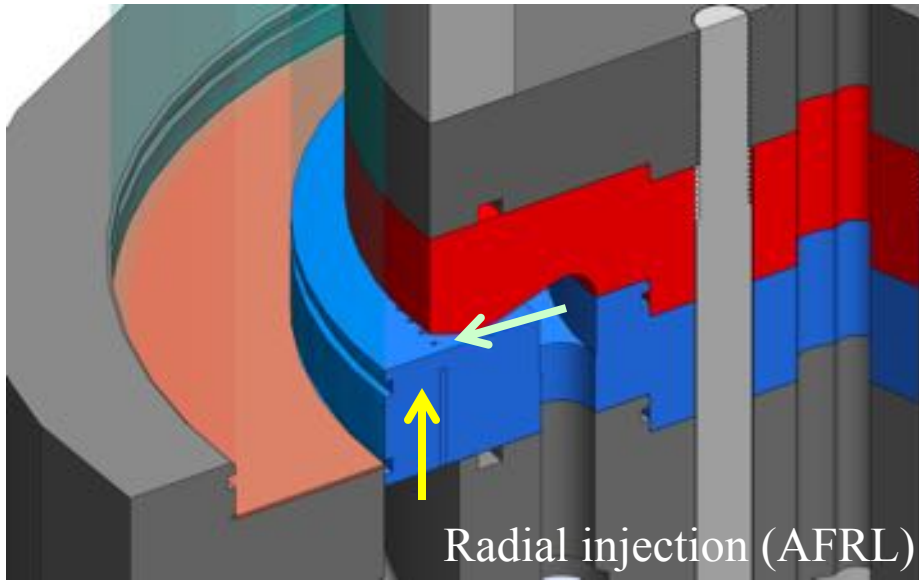
# 3-D rendering of 6" round RDE system

- **Fuel injection system**

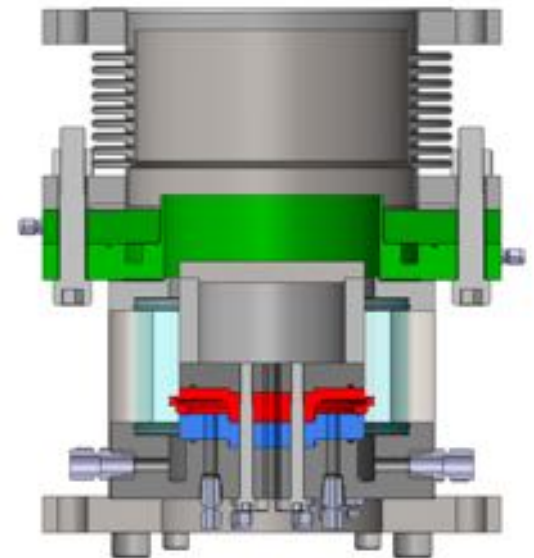
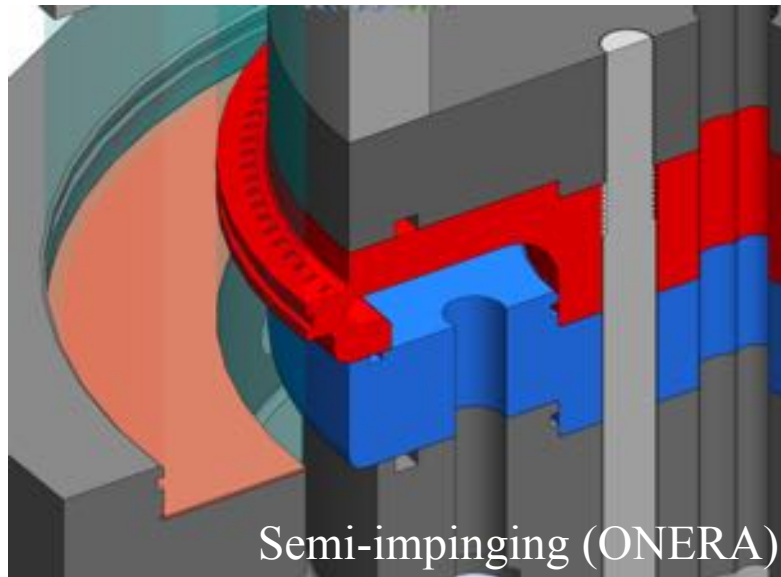
- Red/blue plate pair
- Modular and readily exchangeable
- Three designs currently available



# Three injection schemes available for 6" RDE system



From Gaillard et al., Acta Astronautica, 111:334-344 2015

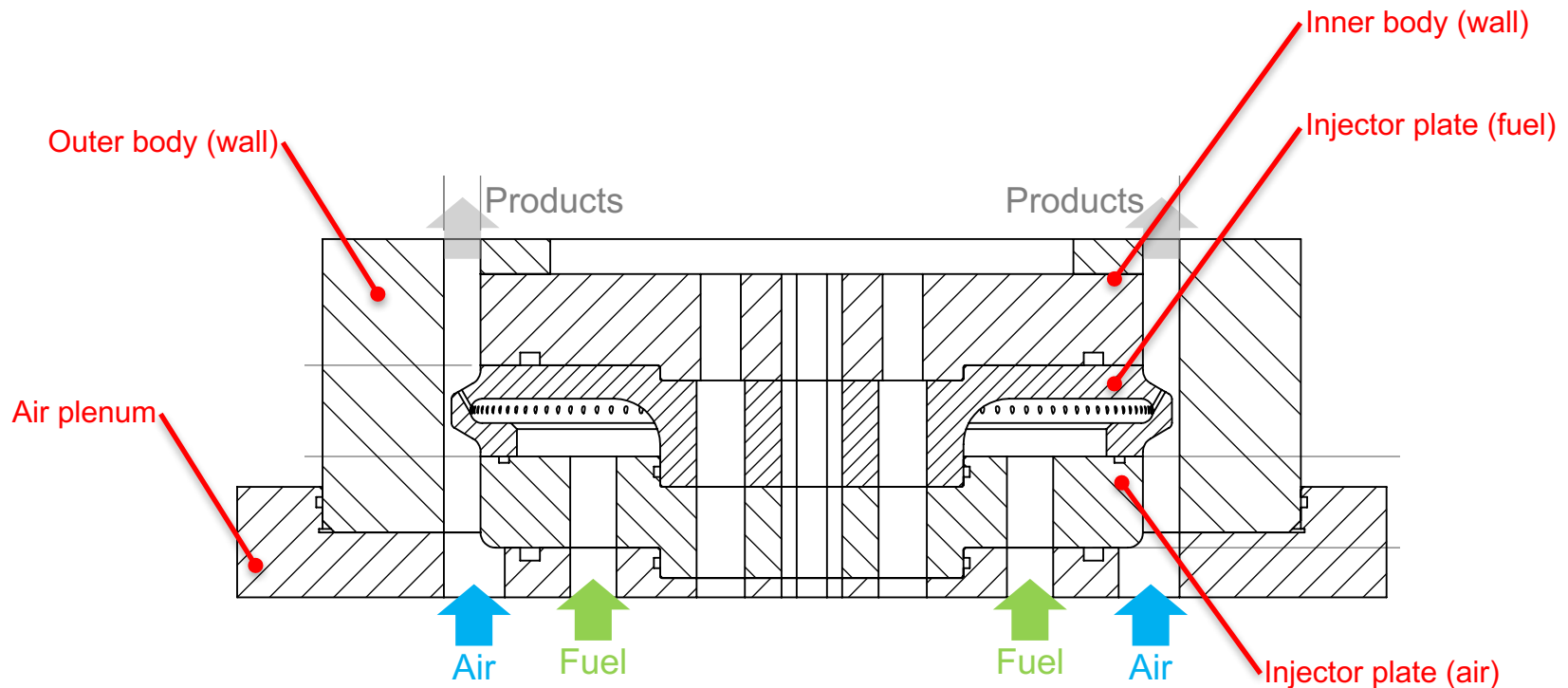
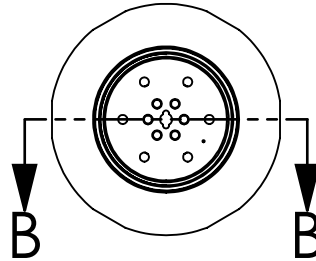




# 6" round RDE system: pintle injector details

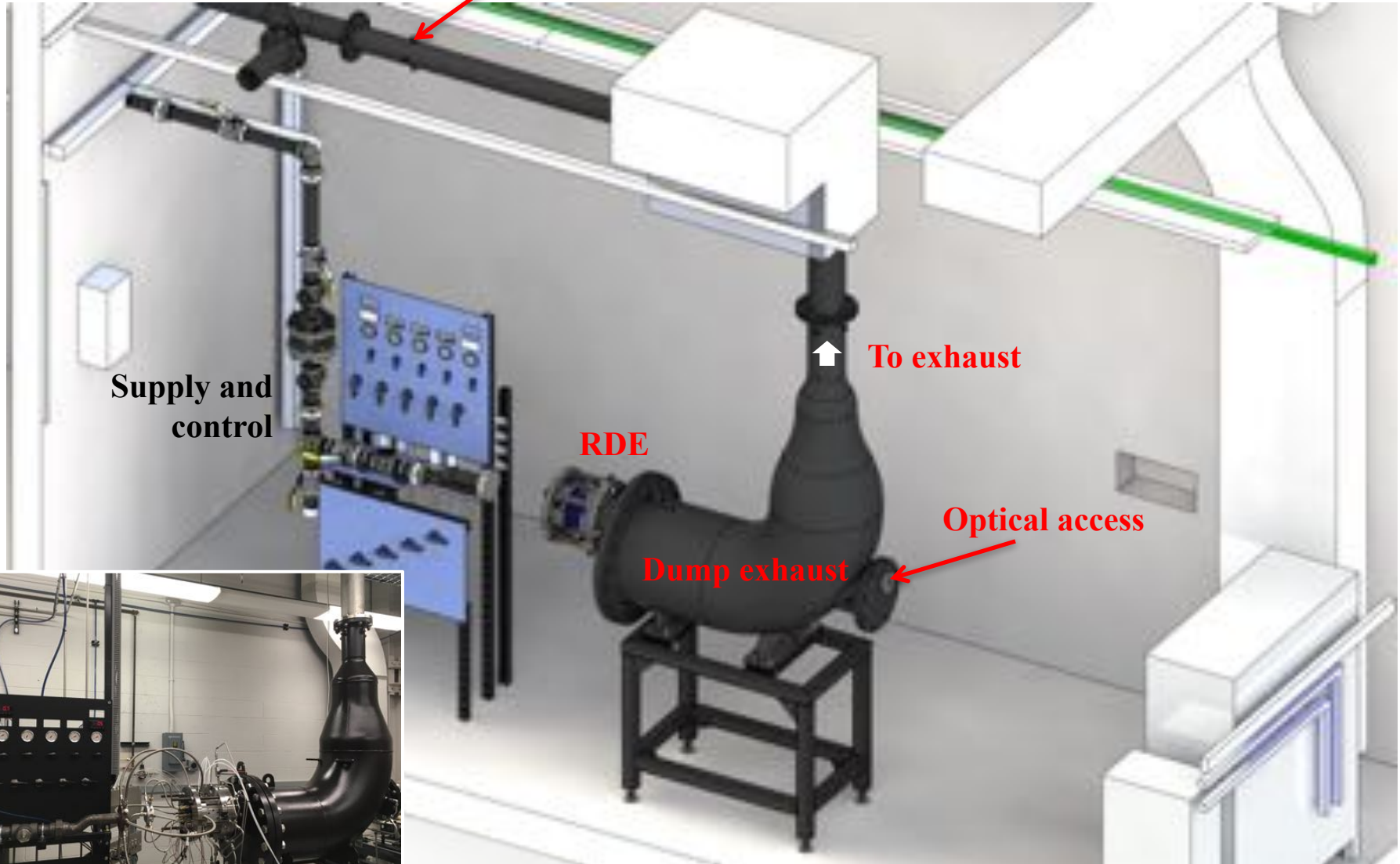


Pintle injector plate

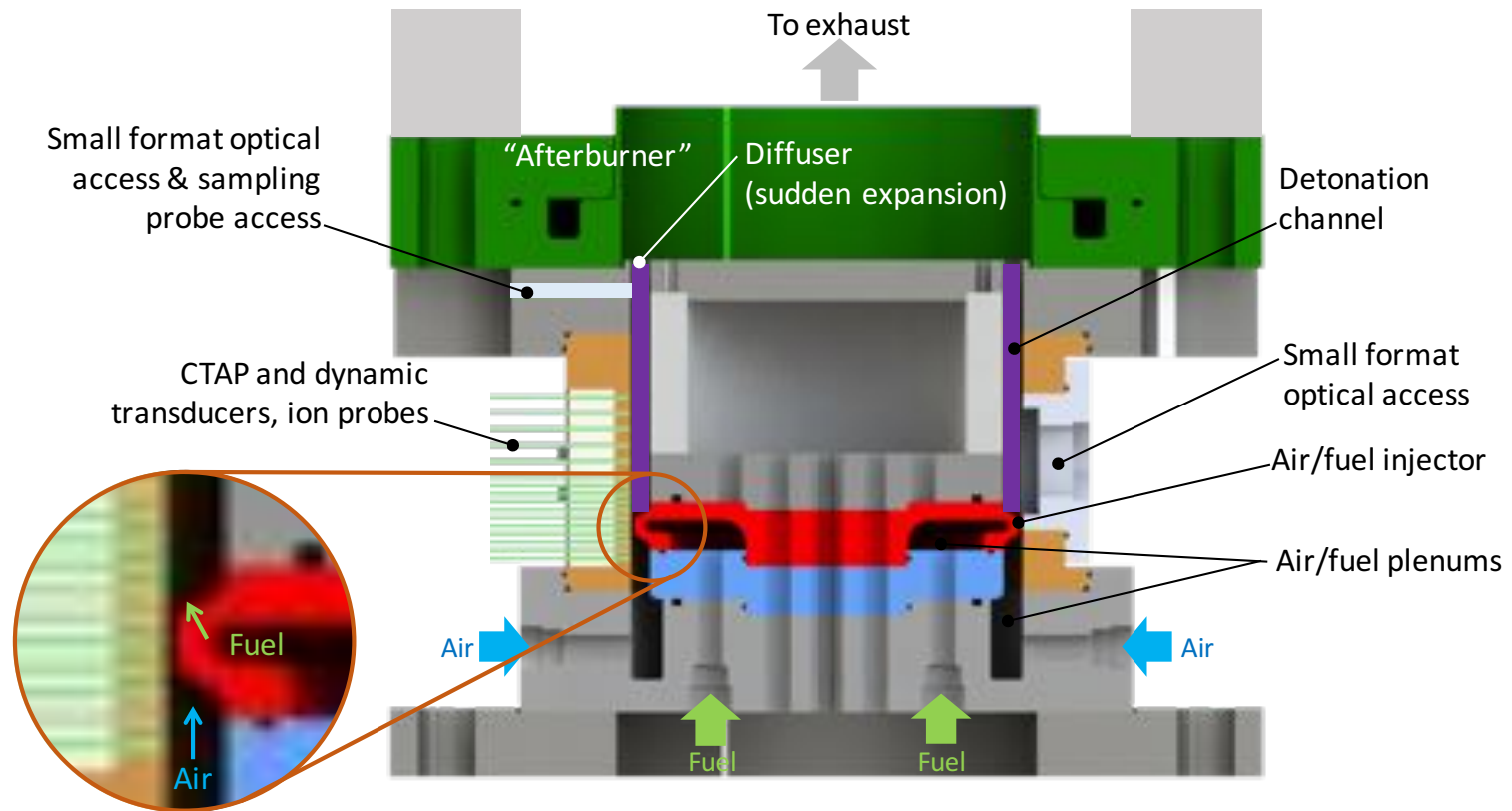


# Pictures of RDE hardware (assembled with exhaust)

Gas sampling (exhaust  
emission measurements)

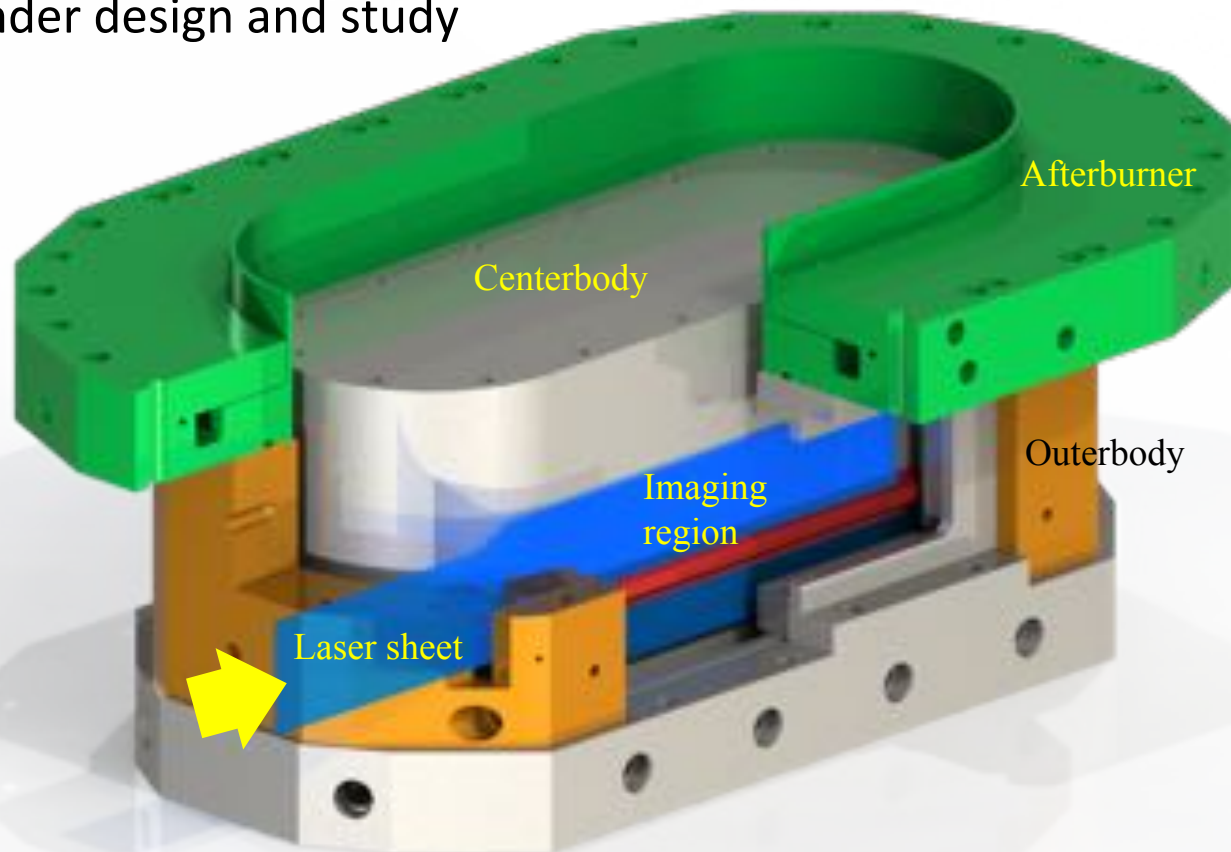


# 6" RDE system: some instrumentation



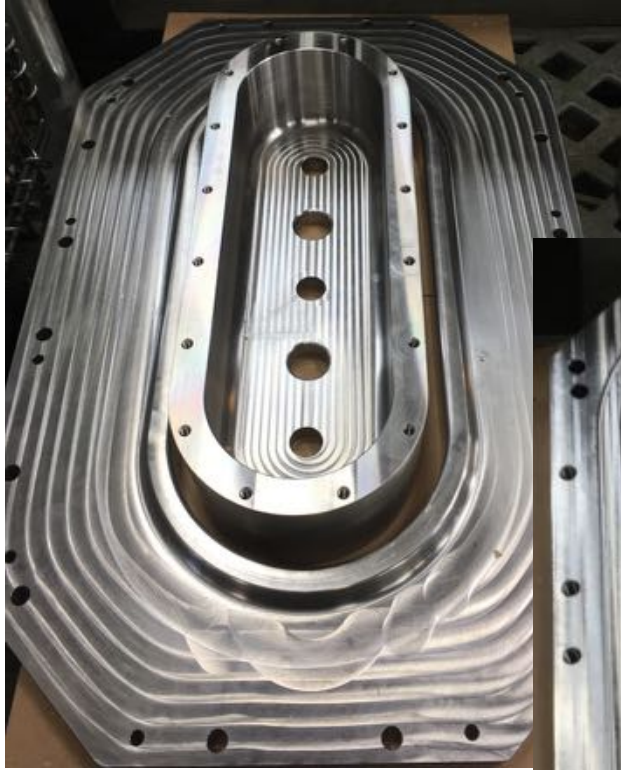
# 3-D rendering of Race-Track RDE system (12")

- **Designed with optical access in mind**
  - Allows for optical access of injection system and detonation chamber
- **Fuel injection system**
  - Follows modular design approach of round RDE
  - Red/blue pair, with similar modularity
  - Injectors under design and study

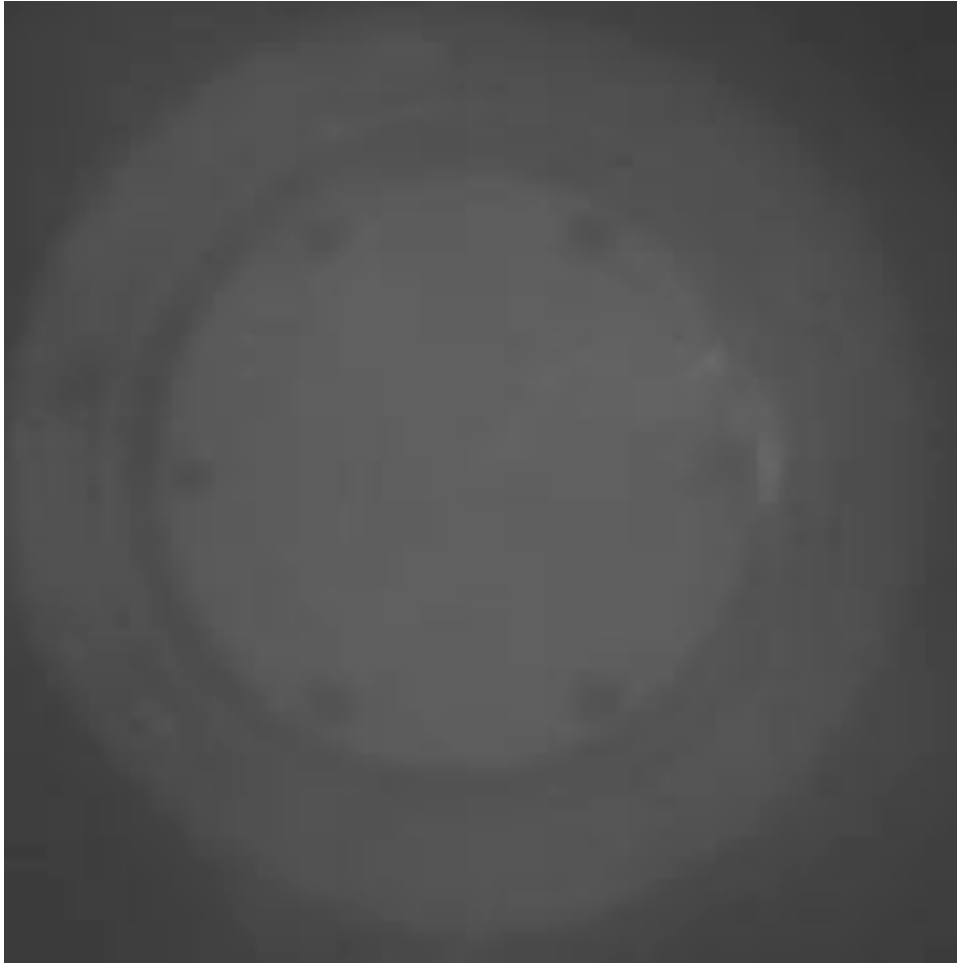




# RT-RDE Being Completed



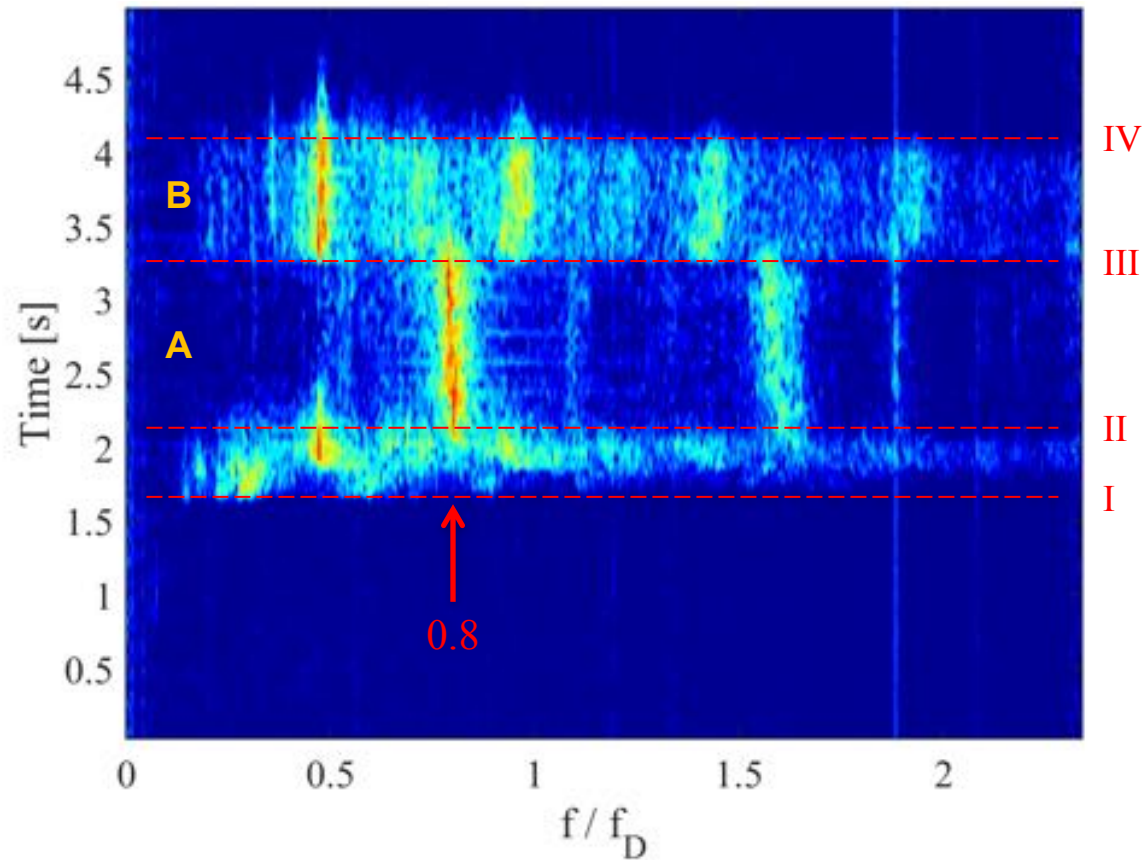
# Example of RDE operation



Mixture:  $\text{H}_2/\text{air}$   
Air flow rate: 450 g/s  
Equivalence ratio: 1

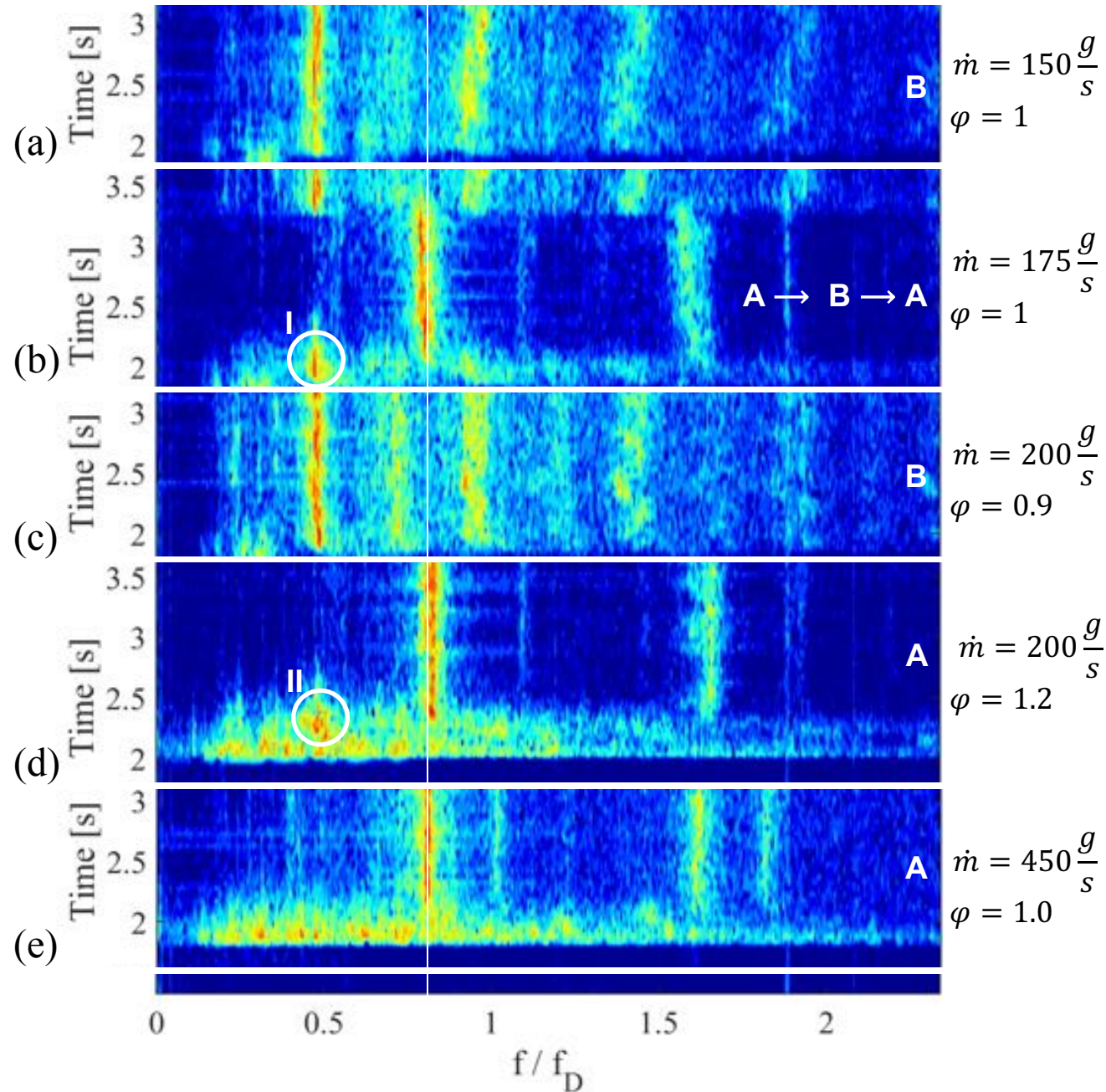
# Test sequence and ignition process (acoustic signature)

- I. Ignition
- II. Transition to detonation
- III. Detonation termination and transition to deflagration
- IV. Fuel off





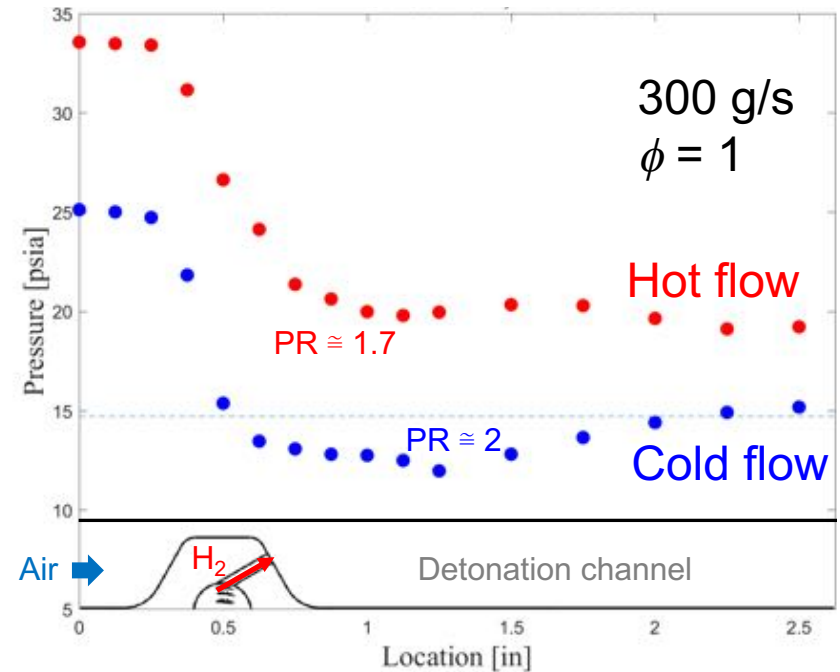
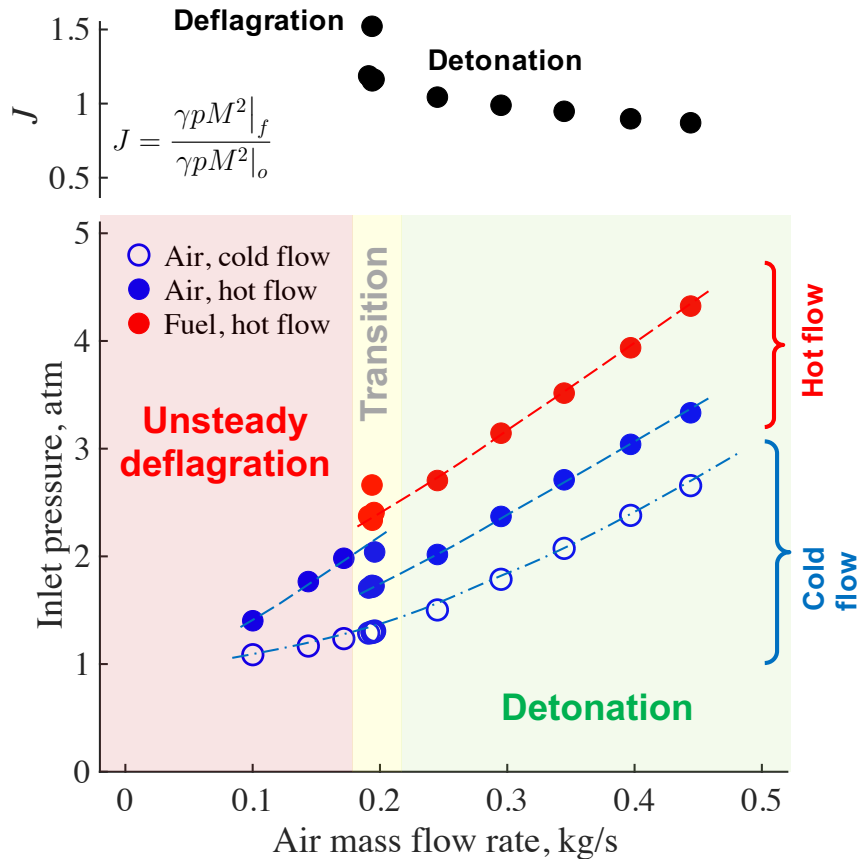
# Test sequence and ignition process (acoustic signature)





# Coupling between operation mode and plenum

- Inlet conditions depend on operation mode
  - In detonation mode, plenum pressure **lower** than in deflagration mode

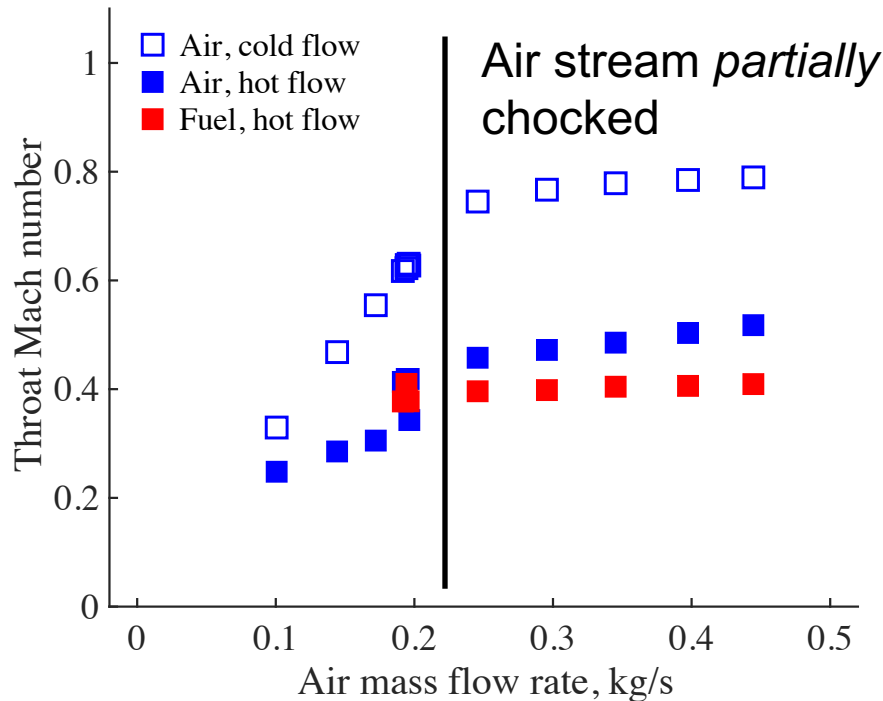


Pressure variation along detonation channel

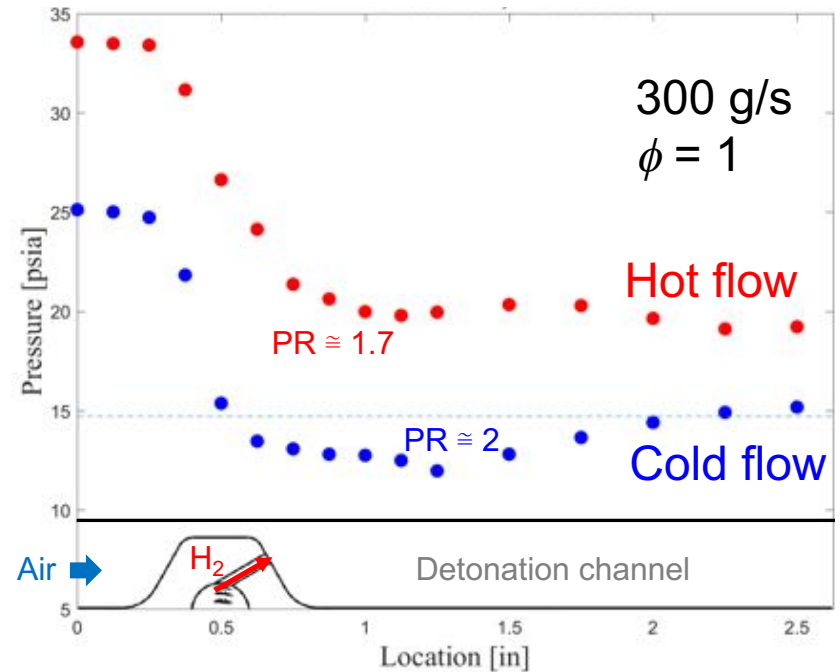
PR = plenum-to-channel pressure ratio

# Coupling between operation mode and plenum

- Inlet conditions depend on operation mode
  - In detonation mode, plenum pressure **lower** than in deflagration mode
- Air injection partially (space/time) chokes at high flow rates
  - Detonation mode is observed (correlation?)
  - Flow possibly separates at injectors (reduced cross-sectional area for flow)
  - Injector stiffness vary over detonation cycle



Estimation of Mach number at air and fuel injection throat

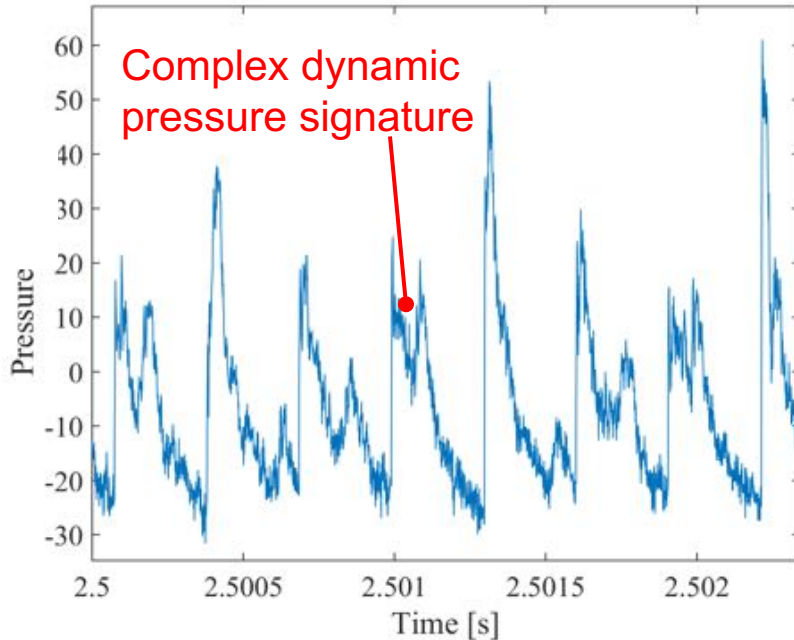


Pressure variation along detonation channel

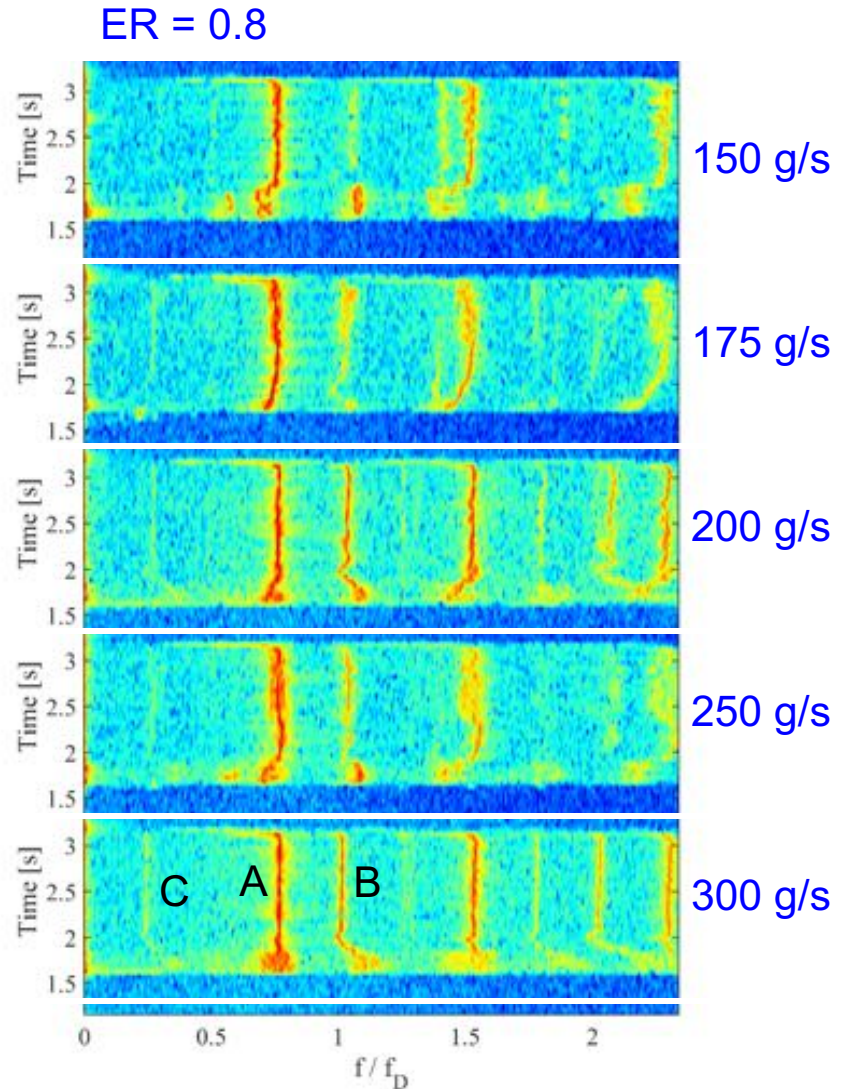
PR = plenum-to-channel pressure ratio

# Waterfall spectra of detonation chamber dynamic pressure

Detonation chamber pressure variation



- **3 main modes typically observed**
  - A: wave propagation speed at  $0.8 f_D$
  - B: Tone at  $1 f_D$
  - C: Tone at  $0.25 f_D$
- **Possibly coupling of various dynamics**
  - Plenum dynamics & detonation wave

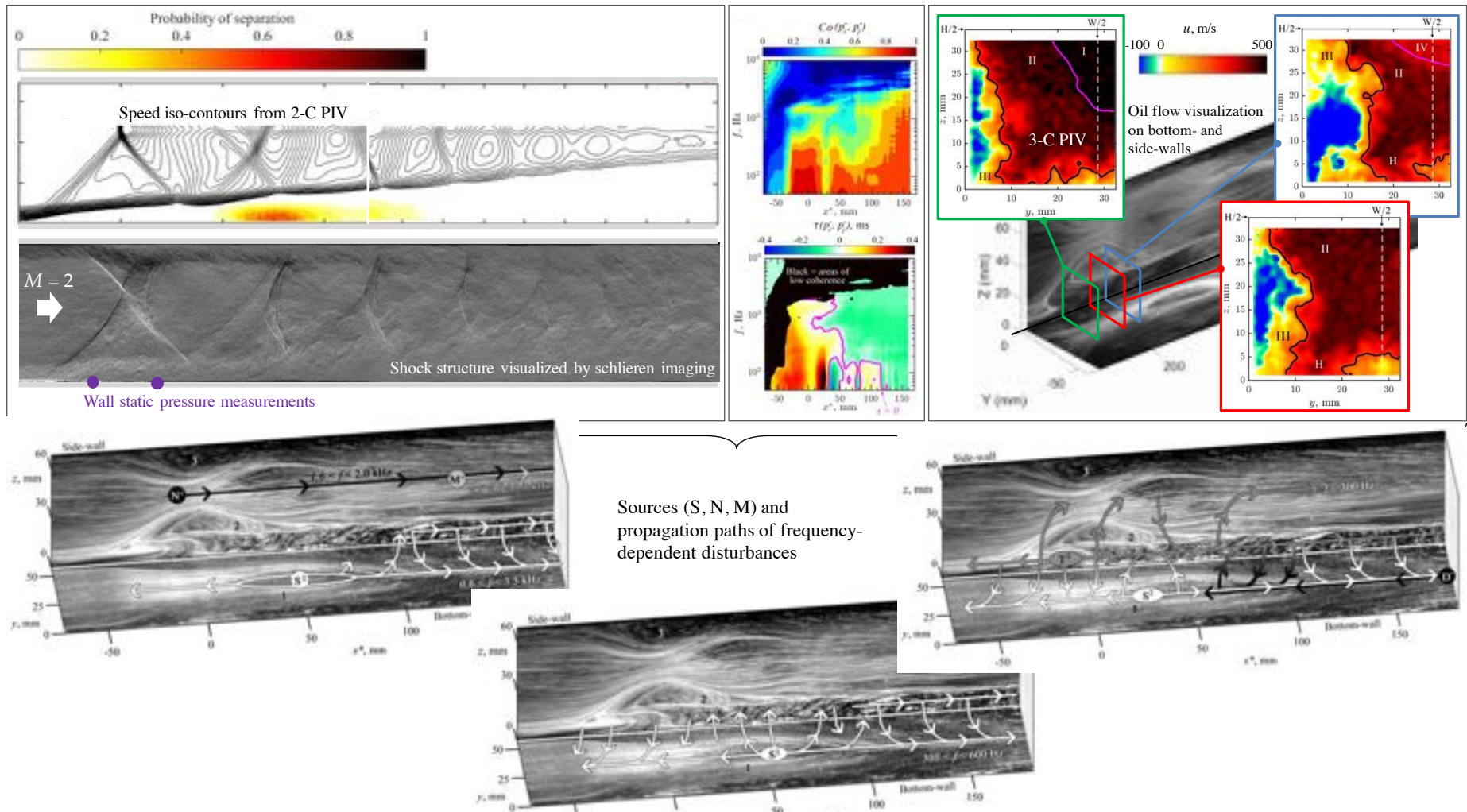


# Focus on project

- **Systems upgrade**
  - Instrumentation upgrades (dynamic pressure) to quantify dynamics
  - Addition of variable area diffuser
  - Air injector with variation in area ratio (stiffness)
  - Extend air plenum to evaluate plenum / detonation wave extent of coupling
- **Two major activities / focus**
  - Dynamics: injector/detonation/diffuser dynamics
  - Multicomponent fuels operation
- **Dynamics**
  - Identification and evaluation of RDE dynamics
  - Questions:
    - what are they?
      - Identification from macroscopic observables on round RDE
    - What do they depend on?
      - Geometric and fuel variation on round RDE
    - How do they effect detonation wave structure and overall operation
      - Detailed laser diagnostics for flowfield measurements (mixing, flow velocity and detonation structure)
      - Combined PIV and tracer PLIF (flame marker or mixture fraction) studies in RT-RDE
- **Multicomponent fuels**
  - Evaluation of use of hydrocarbons on
    - RDE operability and performance from macroscopic observables
    - Detonation structure dependence
    - Effect on detonation dynamics
  - Impact of  $\text{CH}_4$ , and CO additions
  - Impact of  $\text{C}_2\text{H}_4$  or  $\text{C}_3\text{H}_8$  additions (contaminants, fuel flexibility)
  - Impact of  $\text{CO}_2$  addition on changing heat release profile

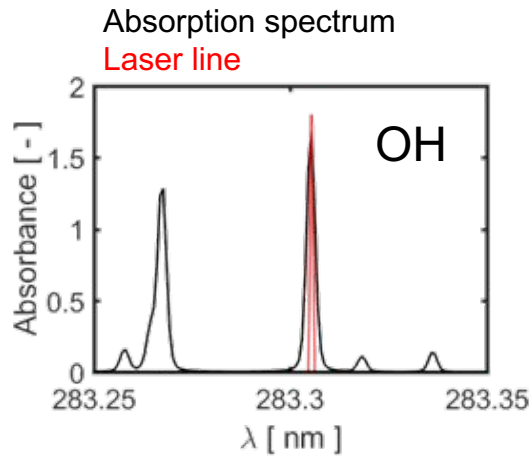
# Example of spectral and cross-spectral analysis for system's dynamics identification

Application to shock train dynamics (supersonic isolator)

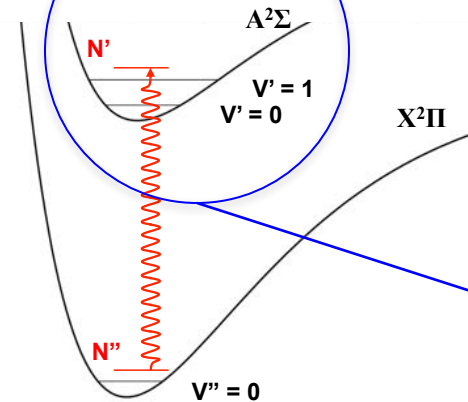




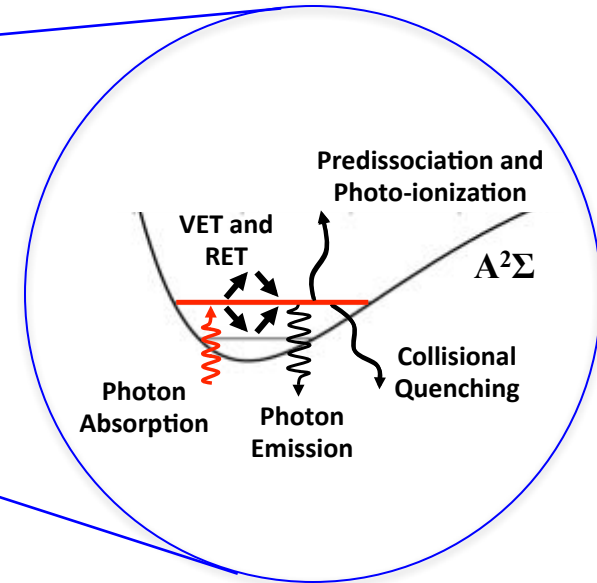
# Collaboration with AFRL / Edwards on diagnostics: augment laser diagnostics for detonating flows



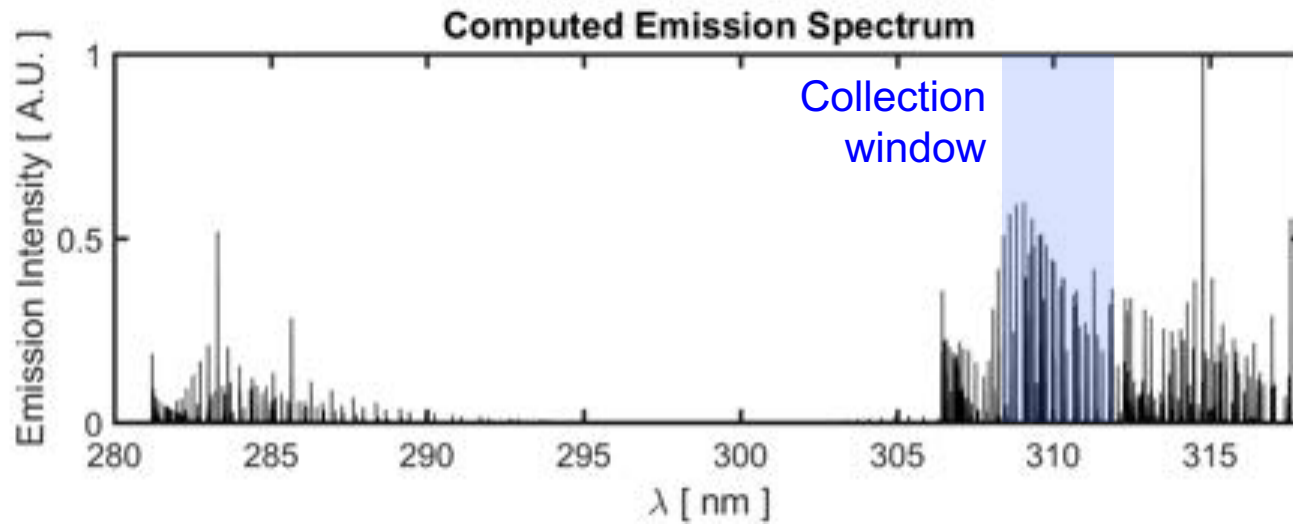
Transition/laser coupling



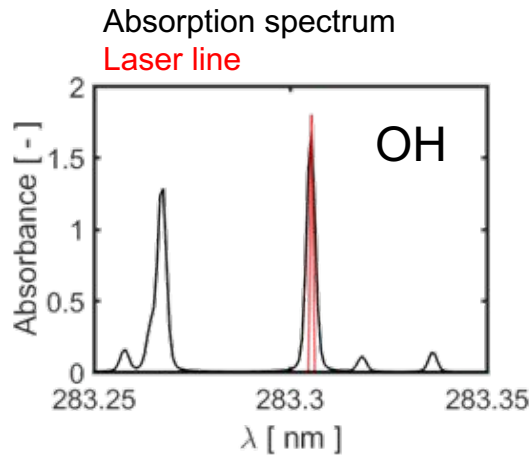
Absorption (rovibronic electronic transition)



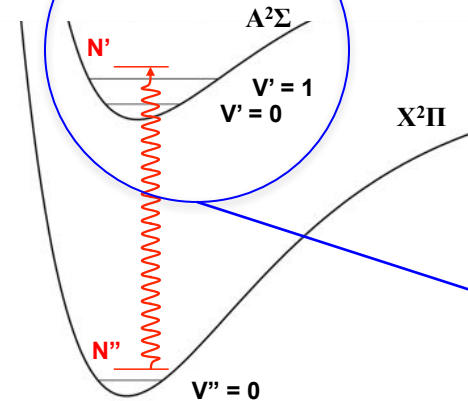
State-resolved energy transfer



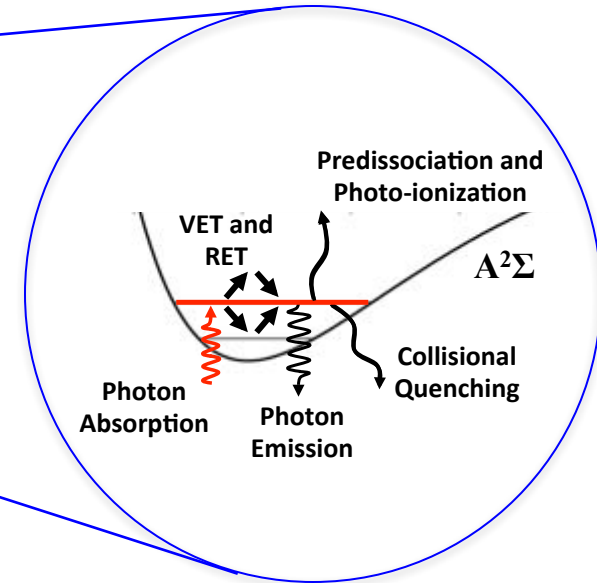
# Collaboration with AFRL / Edwards on diagnostics: augment laser diagnostics for detonating flows



Transition/laser coupling



Absorption (rovibronic electronic transition)



State-resolved energy transfer

- These methods and collaboration gives us a framework to:
  - Evaluate and optimize LIF-based imaging technique for detonating flows
  - Demonstrate methods in RDE relevant flowfields (RT-RDE)
  - Perform measurements on RT-RDE at AFRL/Edwards leveraging their instrumentation and capabilities
    - Anticipated 3 measurements campaigns

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# Goals of CFD Program

- **Develop fully-resolved adaptive mesh compressible solvers for capturing detonation processes**
  - Study structure of detonations in non-premixed systems
  - Develop reduced-order models
  - Study fuel composition effects on stability
- **Assist in the development of the experimental RDE configurations**
  - Provide detailed simulation data to complement experimental measurements
  - Conduct simulations outside of experimental parameters to extend datasets
- **Developments and studies leverage**
  - OpenFOAM suites of codes
  - U-M detonation solvers UMDetFOAM

# OpenFOAM Code Development

- **All codes and models developed using the openFOAM open source code base**
  - 10+ years experience in using this tool
    - Several NETL projects successfully completed
  - Easy transfer of code to industry/research community
    - Prior solvers transferred to Siemens Inc.
- **Highly scalable and runs on 10K+ processors**
  - Extensive code rewrites to ensure linear scalability

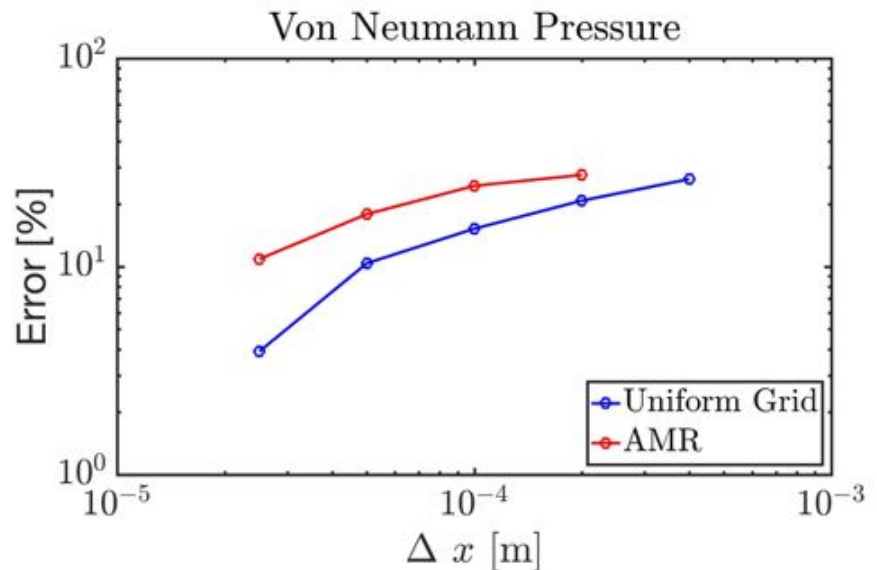
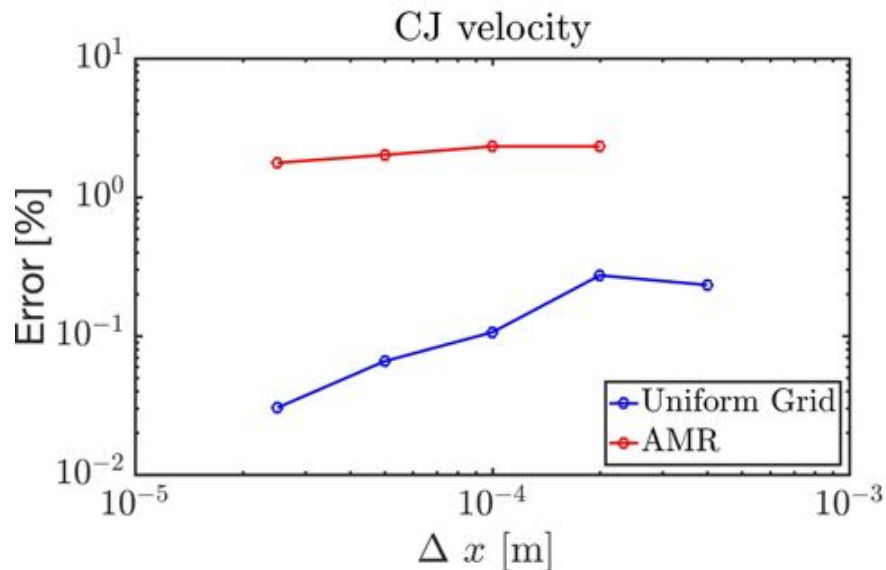
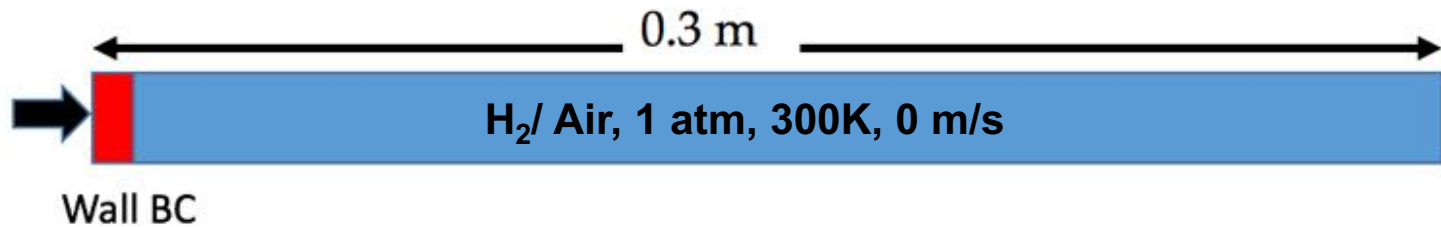
# Compressible Detonation Solved UMDetFOAM

- **Fully explicit solver**
  - Euler and N-S equations
- **Several flux schemes**
  - Locally adaptable to ensure minimal dissipation
- **CANTERA-based chemistry module**
  - Allows any detailed chemistry mechanism to be used
  - Can handle arbitrary number of species
- **Adaptive mesh refinement**
  - Locally adaptive grids to capture detonation structures

# Case Studies - 1D detonation with AMR

- Convergence test with H<sub>2</sub>/ Air mechanism
- The base grid for AMR study is  $\Delta x = 0.4$  mm
  - Shows convergence as increasing the level of refinement

**Product**  
**27.1 atm**  
**3000 K**  
**1975 m/s**

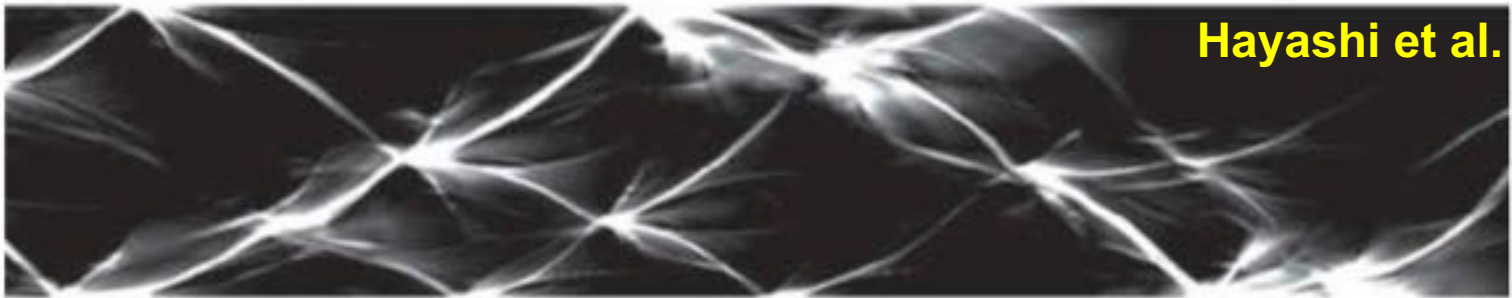
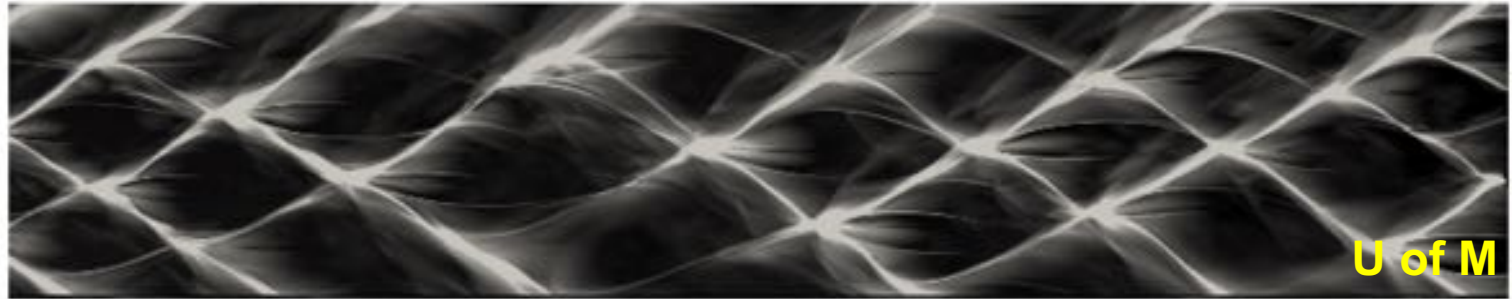


# 2-D ethylene case

- Cellular structure validation

- Longitudinal tracks from the intersection points
- 2 cell structure across the channel width

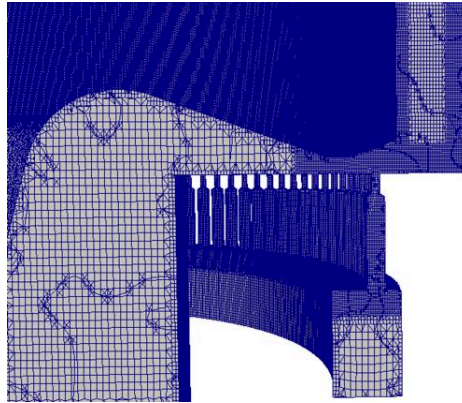
p\_max  
1.500e+06  
1.3e+6  
1.1e+6  
9e+5  
7.000e+05



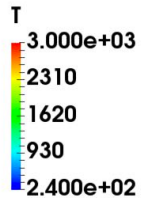
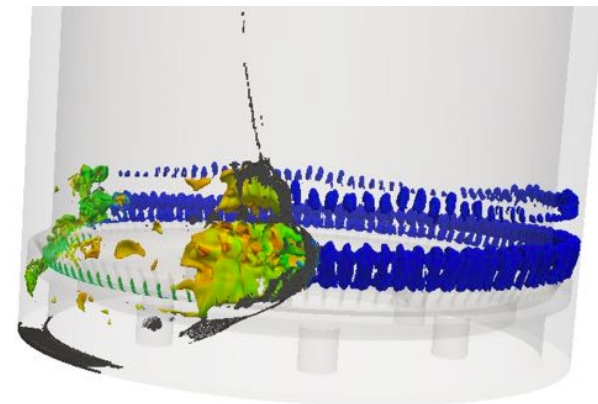
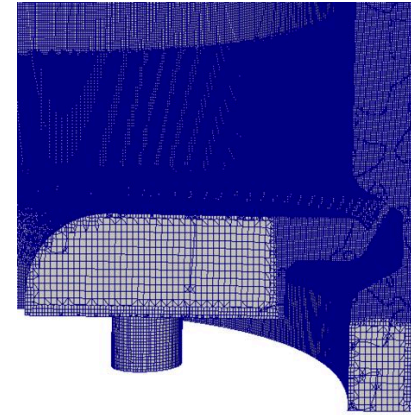
$\text{C}_2\text{H}_4 / \text{O}_2$ , 0.1 atm, 300 K  
 $\Delta = 3 \mu\text{m}$ ,  $h = 2 \text{ mm}$

# AFRL and U-M Full Geometry Modeling

AFRL geometry



Pintle geometry



Legend:

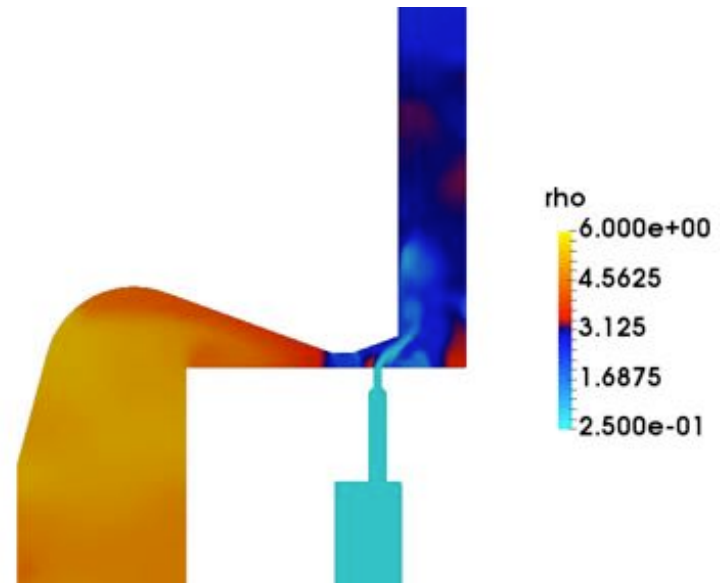
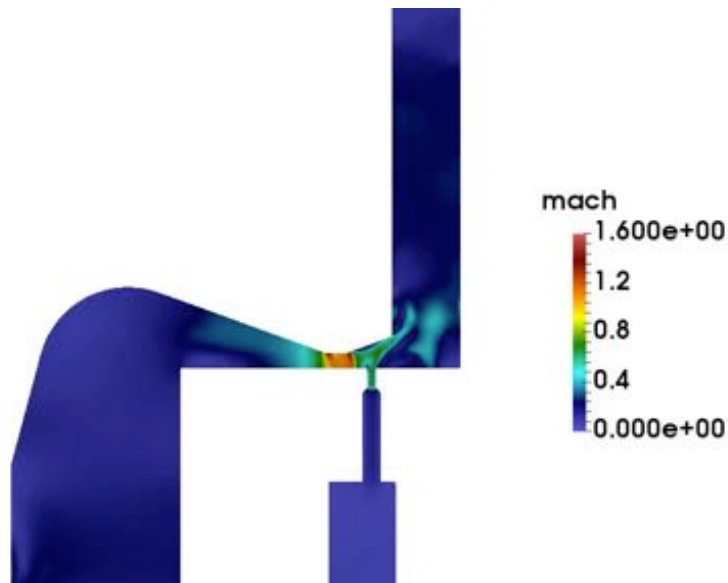
Iso-contour of density gradient (black)

Iso-contour of H2 mass fraction (0.1), colored by temperature

Iso-contour of OH mass fraction (0.0075), colored by temperature

# AFRL Injector Response

- **Flashback occurs when a detonation wave moves across injector**
  - Chocking is terminated
  - Post-combustion gases propagates back into plenum
  - Blast waves move into plenum
- **Quick injector recovery**
  - Reversed flow pushed back into channel due to high plenum pressure



# Focus on the project

- **Simulate fuel effects on cell detonation size**
  - 2D geometry
  - 2D unrolled geometry
  - Full scale geometry
- **Effect of wave structure on detonation process**
  - Coupling between inflow and detonation chamber
  - Interaction of downstream wave structures
- **Modeling detonations**
  - A tabulated modeling approach



# Outline

- Programmatic overview
- Introduction to the problem and general approach
- Experimental activities
- Computational activities
- Interactions and collaborations

# Interactions, collaborations and synergies

- **Strong coupling between experiments and computations**
  - Model development and validation
  - Experiment design and understanding
  - Strong collaborations with external partners
  - Combined investigation of the physics of detonations under MCFs (relevance to application) and impact of injector/detonation/diffuser dynamics on detonation properties and RDE performance
- **Key external collaborations**
  - **ISSI/AFRL WP** (Drs. John Hoke & Fred Schauer) on RDE operation, performance and modeling.
  - **UTRC** (Drs. Adam Holley and Peter Cocks) on detonation and RDE modeling for arbitrary fuels and geometries.
  - **GE** (Venkat Tangirala) on RDE operation, performance and modeling.
  - **AFRL/Edwards** (Dr. William Hargus) on the development and application of diagnostics applied to relevant RDE geometries.
  - **Williams International** (Kyle McDevitt) on detonation and RDE modeling for arbitrary fuels and geometries.
- **Other collaborations/interactions**
  - **NETL** (Dr. Ferguson) on modeling and RDE performance & operation
  - **University of Maryland** (Prof. Yu) on use of experimental data for validation in simple geometries
  - **NRL** (Dr. Kailasnath) on code and combustion model development

**Questions?**