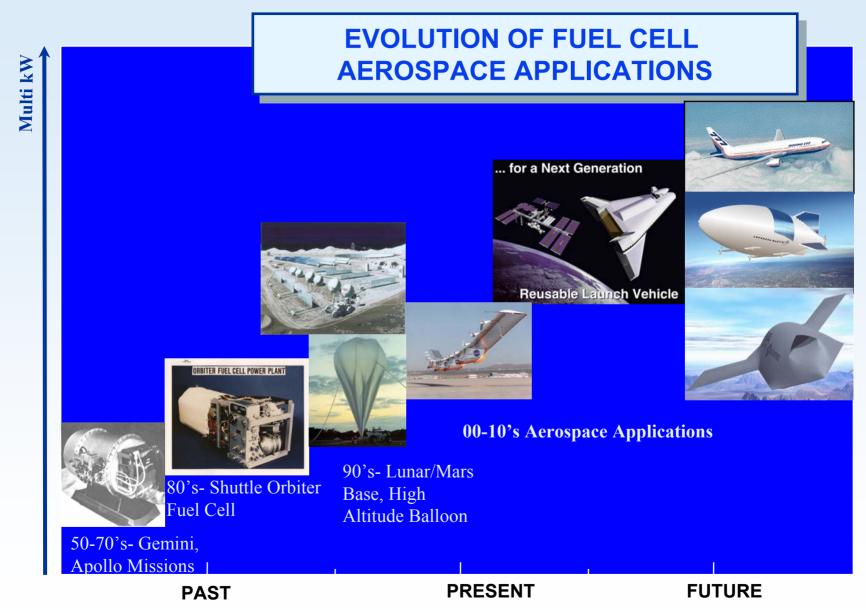
Emerging Fuel Cell Developments at NASA for Aircraft Applications

By

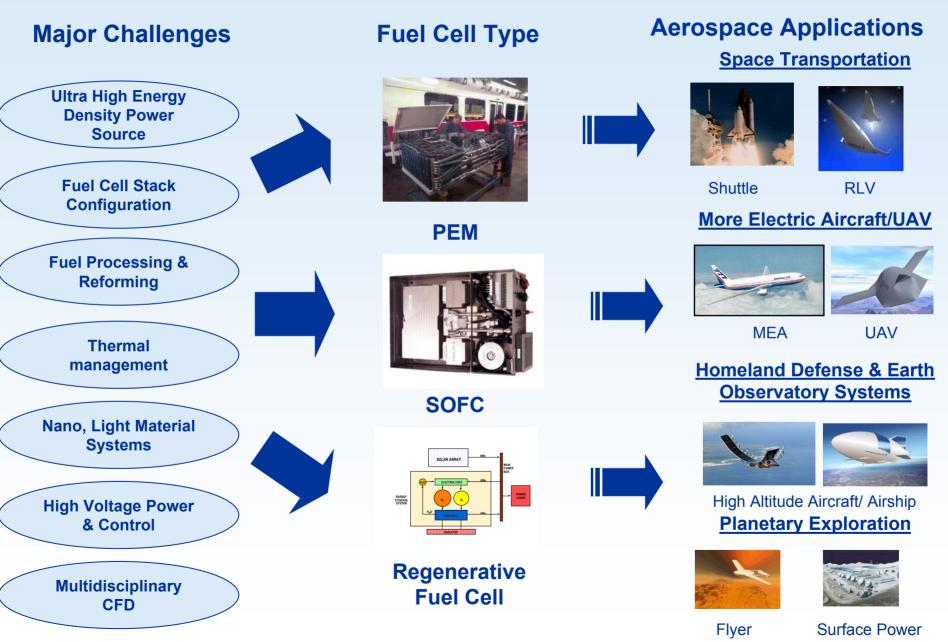
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Fuel Cells for Aerospace Missions



Early Demonstration Opportunities

For Electric Propulsion

& Power







Mars Airplane

All Electric High Altitude Airship

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Fuel Cell Based APU Demonstration

All Electric Fuel Cell Powered Light Aircraft



Solid Oxide Fuel Cell Technology for <u>Nex</u>t Generation <u>Clean Aircraft Power</u>

- NEXCAP Project -

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Formulation Objectives

- Build on Department of Energy's Successes with the Solid-State Energy Conversion Alliance (SECA) Program by Developing a Long Term Technology Plan to Advance Solid Oxide Fuel Cell Capabilities for a wide range of aircraft power and propulsion applications.
- Complement SECA's program objectives on cost reduction to address power density (kW/L) and specific power (kW/kg) challenges critical for aircraft applications.
- Establish NASA leadership and Build National Support for Aircraft Fuel Cell Powered Technology Plan.



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SOFC for Commercial Aircraft APU

Current Gas-Turbine APUs operate at ~15% load cycle efficiency, contribute up to 20% of the aircraft ground based emissions, and APU/secondary power systems account for 50% of the maintenance delays which presents 12% of the maintenance cost.

SOFC Offers:

- Solid state characteristics, making them simpler in concept and design.
- Greater fuel flexibility and simpler fuel reforming enabling the transition from petroleum to hydrogen economy.
- Higher quality heat effluent making them particularly suitable for hybrid gas turbine systems with the highest potential system efficiencies.
- Inherently environmentally friendly, producing no/negligible NOx and significantly reduced CO₂ with hydrocarbon fuels.
- Suitability to multiple markets spanning stationary, transportation, aerospace, and military applications, facilitating DOE, DOD, and NASA Collaboration.

Present GT-APU:

- On-ground and emergency In-flight electrical power
- On-ground Environmental Control System (ECS)
- Main engine start

Full time, Fuel Cell Power Unit (FCPU) Concept :

- Environmental Control System (ECS)
- Electric main engine start
- De-icing
- Onboard water generation
- electric actuated control surfaces

Aircraft Challenges

Airline Industry requires a single fuel, jet fuel, on the aircraft for operations for the foreseeable future.

- Require compact, light weight, and efficient fuel reformer and desulfurization systems
- Determine and develop most effective fuel reformer (Steam, CPOX, autothermal) from an integrated system perspective
- Determine and develop most practical method to address fuel sulfur challenge (remove at refinery, liquid or gas phase removal in ground based or flight based system, sulfur tolerant anode and catalysts, etc.)

Fuel Cell Stack/System power to weight and volume requirements for flight applications are at least an order of magnitude greater than current SOA and several times greater than SECA's 2010 goal.

- New anode/electrolyte/cathode material systems to reduce electrochemical losses
- Durable, light weight fuel cell interconnects and improved gas seals
- Advanced stack cooling concepts to enable higher power density operation
- Optimize designs for gas flow, current and temperature distribution
- Fabrication process development for large stacks without performance degradation
- Durability of stack under aircraft operating conditions (vibration, acoustics --)



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Aircraft Challenges, Con't

"Balance of Plant" represents 2/3 of the total system and it is complicated by high temperature operations.

- Highly integrated components to achieve weight and volume goals
- Effective thermal management for stack heat dissipation and system thermal balance
- Advanced controls and diagnostic systems for autonomous/long life operations
- High temperature heat exchangers
- Improved insulation materials
- Light weight electrical power management and distribution systems
- Lightweight materials and structures to reduce weight

Aircraft applications require operating life 2 to 4 times greater than DOE transportation based systems.

• Planar SOFCs still at TRL ≤ 3 for ground based applications!



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Current Capability Vs. Goal

| Attributes | Current Capability | Goal | |
|--|--|---|--|
| Total Power | 2-5 kW (Planar) –dev | 5 kW for early aviation demo | |
| | > 100 kW +(tubular) | 145 kW for 100 passenger | |
| | > 1 MW (planned) | 450 kW for 305 passenger | |
| | | 3 – 10 kW (SECA transportation) | |
| Specific Power for entire | 0.02-0.04 kW/kg – | 0.5 kW/kg (NASA/DOD) | |
| SOFC system incl. BOP | developmental | 0.1 kW/kg (DOE - SECA) | |
| Specific Power for stack | < 0.2 kW/kg for stacks | 1 kW/kg with TBD stack kW total | |
| | and 1-5 kW total power | power requirements | |
| Area Specific Power | 0.5-1 W/cm ² cell | 2 W/cm ² cell | |
| density for cell/stack (W/cm ²) | ~.4 W/cm ² stack | >1 W/cm ² stack | |
| Fuel Reformation | Mature at the industrial scale | Compact, lightweight system with high conversion efficiency | |
| Sulfur Tolerance | Limited exp. with logistic fuels, 100's of | 300 – 700 ppm current jet fuel sulfur level | |
| | hrs | Aircraft life 40,000 hrs | |



Current NASA Activities – System Study

Evaluate system-level benefits of a full-time, fuel cell based aircraft APU to determine concept feasibility.

- Contracted effort with Boeing targeting a jet fuel based fuel cell with a 2015 Entry-Into-Service (EIS) application.
 - The Boeing 777-200ER aircraft selected for the study
 - A More Electric Aircraft (MEA) architecture chosen.
 - Subcontracts with Fuel Cell Companies to provide fuel cell information
- In-house effort with a broader scope to address alternative fuels and advanced concepts for both near and far term EIS.
 - Top level assessment of fuel cell technologies to identify promising concepts for aircraft applications.
 - Establish system level and higher-fidelity modeling capability to evaluate candidate SOFC cycle concepts (models being shared with Boeing).
 - Identify critical technology areas and define a technology maturation plan.

Augmentation - Conceptual Design Studies

- Initiate contracted conceptual design studies (6 9 month efforts) with competing fuel cell and aerospace companies to
 - Identify the most promising concepts for 2015 EIS applications based on aircraft power system requirements developed under Boeing Study.
 - Develop a Technology Maturation Plan



Current NASA Activities Critical Technology Development

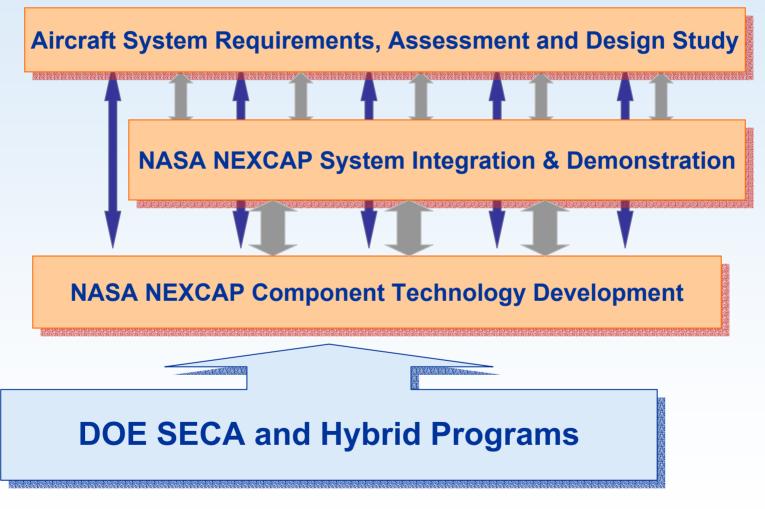
Develop compact, lightweight, and efficient jet-fuel processing technology to enable near term application of SOFCs to aircraft power systems with collaboration from DOE.

- Obtain a fundamental understanding of SOFC reforming process and to access SOFC and system integrators for technology advancement, performance improvement, and system optimization.
- Identify and characterize promising candidate hydrocarbon fuels by developing a fundamental database of chemical kinetic reaction rates and high temperature characteristics.

Improve SOFC material capabilities to meet aircraft performance, size, weight and life requirements.

- Improve power density through a combination of reducing anode thickness by a factor of 10-15 and reducing electrochemical losses by developing new and improved cathode material.
- Improved, durable high temperature seal.

NEXCAP Strawman - A Technology Maturation Project

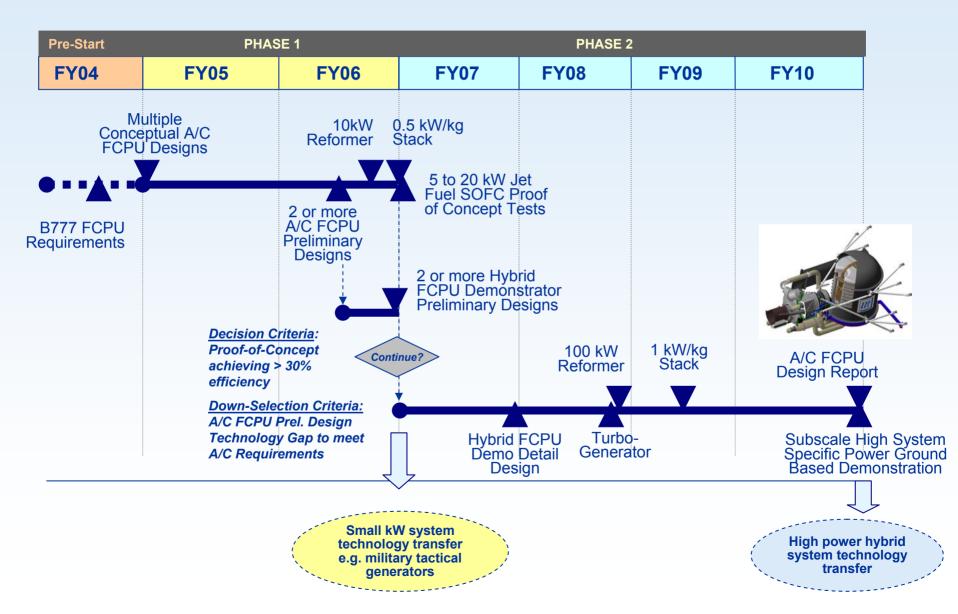




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NEXCAP Milestones and Deliverables



NEXCAP Participants

| | NASA | DOE & National Labs | DOD | Industry | Academia |
|----------------------------------|------|------------------------|-----|----------|----------|
| Follow on Flight Demo | X | | X | X | |
| Aircraft System Definition | X | x | X | X | |
| SOFC System Integration | X | x | X | X | |
| Core Technology | X | X | | X | X |





NASA's Vision for Fuel Cell Powered Aircraft

